
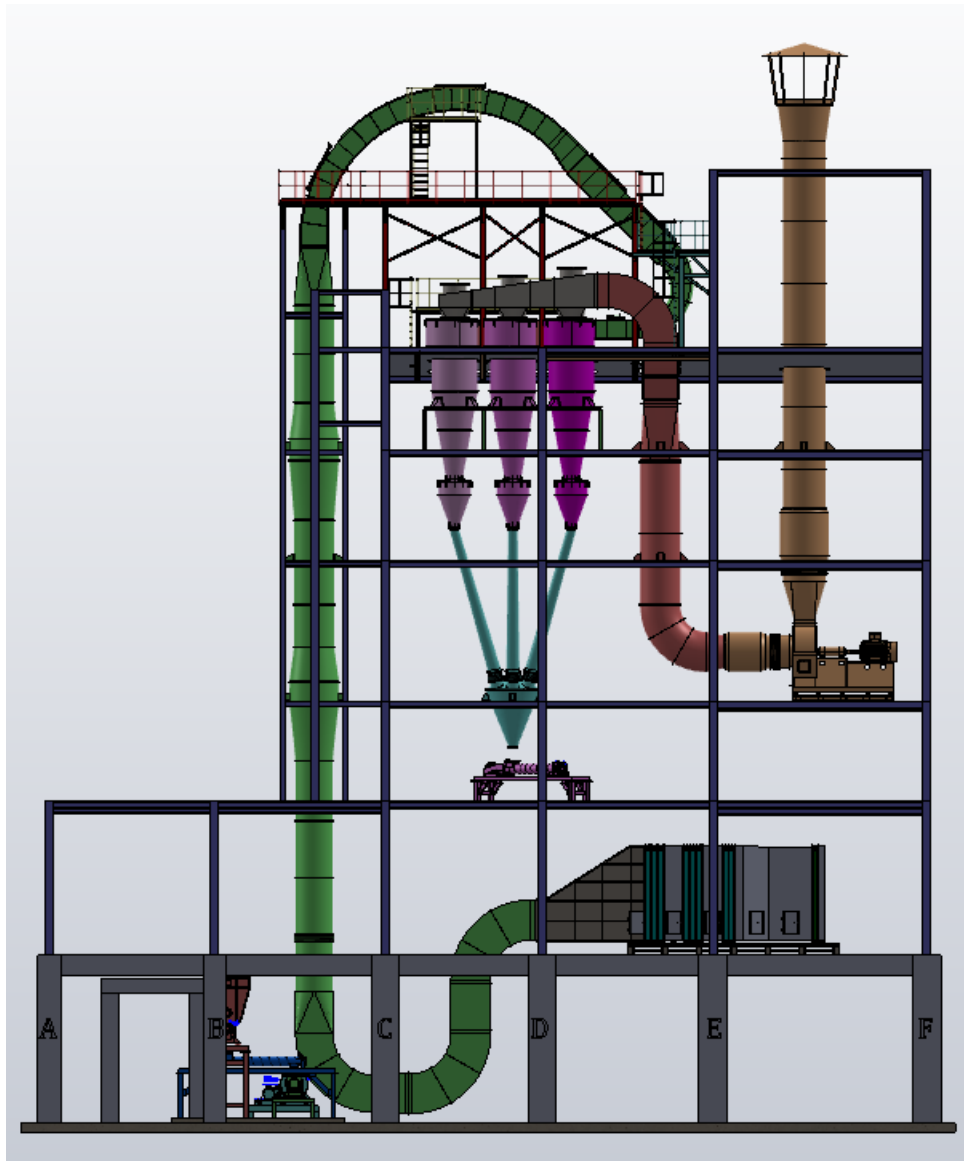
	MODIFIED STARCH FLASH DRYER PROJECT						
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.	Rev.	
	5204	ME	CL	000	02	00	

FLASH DRYER THICKNESS CALCULATION





	MODIFIED STARCH FLASH DRYER PROJECT						
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1- INTRODUCTION

Given that the duct of the flash dryer is under the negative pressure induced by the suction fan (0.065 bar), the stress analysis of the duct is performed based on the fan's maximum suction pressure, incorporating an appropriate safety factor. As is known, the maximum inlet air temperature of the fan is approximately 170°C, which decreases along the length of the dryer. To achieve a more accurate analysis and to incorporate a safety margin, the air temperature is assumed to be constant at 170°C throughout the entire length of the dryer. A maximum positive pressure of 0.5 bar (gauge) is considered in accordance with the manufacturer's recommendation.

A critical aspect of the stress analysis and minimum thickness calculation for the dryer is that the calculated minimum required thickness must be compared against the actual thickness specified for each section to ensure it does not exceed the actual value. Furthermore, based on the actual thickness, the maximum allowable working pressure for each section has been calculated. The significance of this maximum allowable working pressure lies in its application for vent sizing calculations; specifically, the Maximum over pressure generated by an explosive atmosphere (P_{req.}) used in vent sizing must not exceed the maximum allowable pressure of any section, or in other words, the weakest section within each controlled volume.

For the purpose of calculating and analyzing stresses induced by pressure, vacuum, and temperature, the dryer duct has been divided into several sections. This division falls into two main categories: sections with circular cross-sections and sections with rectangular cross-sections. Sections with circular cross-sections have been simulated and analyzed using PV Elite software, while sections with rectangular cross-sections have been analyzed using CodeCalc software. It should be noted that, according to the zoning illustrated in the figure below,

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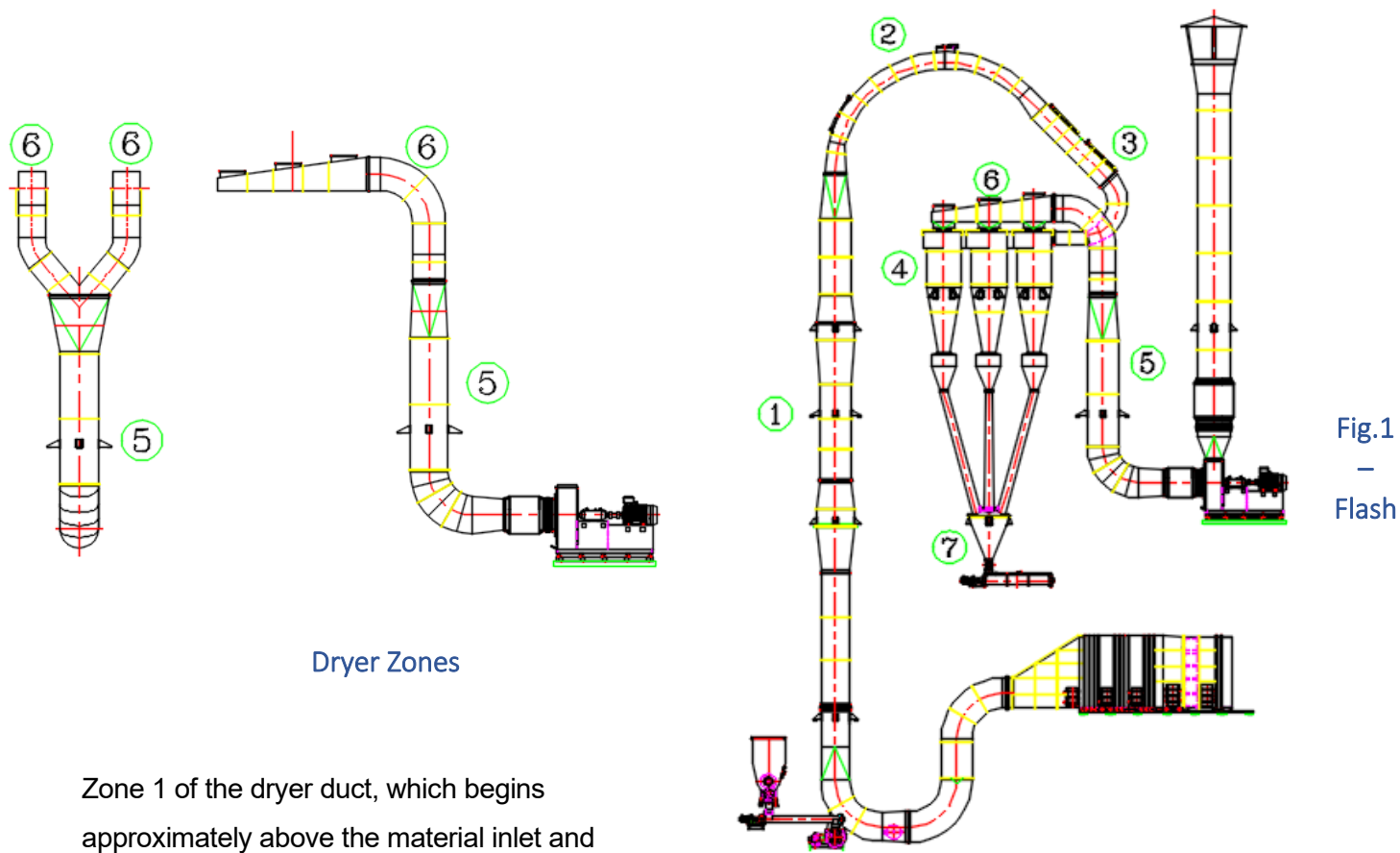




Fig.1
–
Flash

Dryer Zones

Zone 1 of the dryer duct, which begins approximately above the material inlet and extends to Zone 2. Furthermore, the cyclones (Zone 4) , Fan inlet duct (Zone5) and hopper (Zone 7) have circular cross-sections and have been analyzed using PV Elite software .

Zone 2 , Zone 3 and Zone 6 have rectangular cross-section and have been analyzed using CodeCalc software .

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2- ASSUMPTIONS

The circular section of Zone 5, with a diameter of 1400 mm, is similar to Zone 1; therefore, the analysis results for Zone 1 are also applicable to Zone 5, and a separate analysis is not required.



Zones 2, 3, and the portion of Zone 6 with a rectangular cross-section have been analyzed using CodeCalc software. Zone 3 is the most critical section, as it has both larger dimensions and the highest risk for explosion. Consequently, the analysis results for this section can be generalized to the other rectangular sections, which is a conservative approach that increases the safety factor.

However, stress analysis will be performed for Zones 2, 3, and 6 separately.

The design assumptions, based on the mechanical specifications of the equipment and the data received from the manufacturer, are as presented in the below

- Operating Internal Temperature = 55° C - 165° C
- Design Internal Temperature = 170° C
- Operating External Temperature = -20° C to 35 ° C
- Design External Temperature = 50° C
- Maximum Operating internal pressure (Opening Pressure of Explosion Vents) = 0.1 bar
- Design Internal Pressure = 0.5 bar
- Maximum External Pressure (Maximum Vacuum Pressure of fan)= 0.065 bar
- Design External Pressure (Design Vacuum Pressure) = 0.1 bar
- Circular Duct Thickness(Zone 1 , 4) = 3 mm
- Rectangular Duct Thickness (Zone 2 , 3, 6) = 3 mm
- Duct Material= AISI SS 316

Finally, following the calculations detailed in Appendices 1 through 6, a summary of the calculations and stress analysis is presented in the table below. As can be observed, the actual thicknesses considered are adequate, and the maximum allowable working pressure for each section has been duly incorporated into the calculations for the required vent size.

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3- SUMMARY OF CALCULATIONS



The stresses calculations and analysis summary for the different areas of the dryer duct are as follows :

Table 1- Assumption and Design Data



Zone	Title	Section	Material	Actual Thickness (mm)	Design Internal Pressure (Bar)	Design External Pressure (Bar)	Design Internal Temperature °C	Design External Temperature °C
1	Duct	Circular	AISI 316	3	0.5	0.1	170	50
2	Duct	Rectangular	AISI 316	3	0.5	0.1	170	50
3	Duct	Rectangular	AISI 316	3	0.5	0.1	170	50
4	Cyclone	Circular	AISI 316	3	0.5	0.1	170	50
5	Duct	Circular	AISI 316	3	0.5	0.1	170	50
6	Duct	Rectangular	AISI 316	3	0.5	0.1	170	50
7	Hopper	Circular	AISI 316	3	0.5	0.1	170	50

Table 2- Calculated Data

Zone	Title	Section	Maximum Allowable Internal Working Pressure (Bar)	Maximum Allowable External Working Pressure (Bar)	Minimum Required Thickness (mm)
1	Duct	Circular	3.5	0.137	2.6
2	Duct	Rectangular	0.83	0.1	2.8
3	Duct	Rectangular	0.81	0.1	2.8
4	Cyclone	Circular	1.6	0.27	2
5	Duct	Circular	3.5	0.137	2.6
6	Duct	Rectangular	0.81	0.1	2.8
7	Hopper	Circular	3.9	0.23	2.15

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APPENDIX 1 **CALCULATION** **CIRCULAR SECTION OF DUCT** **ZONE 1**

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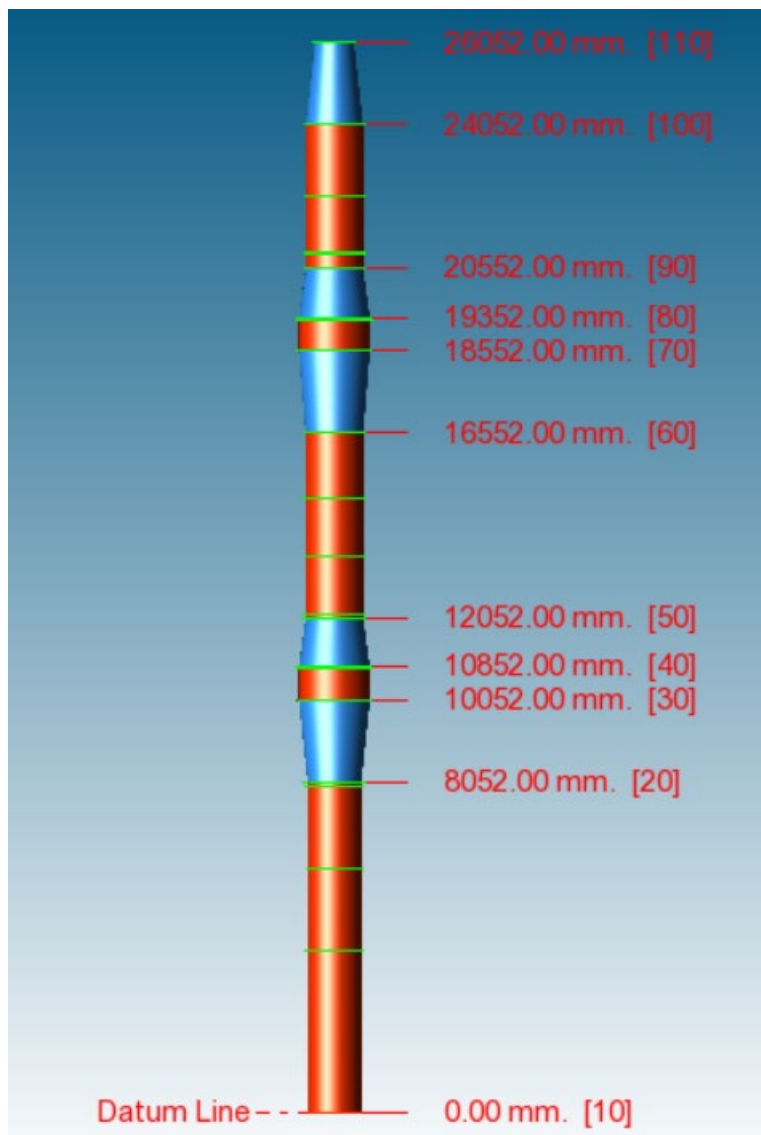


Figure A – Circular Duct - Zone 1

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DESIGN CALCULATION

In Accordance with ASME Section VIII Division 1

ASME Code Version : 2017

Analysis Performed by : SPLM Licensed User

Job File : D:\2-PAYAM\1-PROJECTS\MICROTEC PROJECT\13-MICROT

Date of Analysis : Feb 20,2026 4:14pm

PV Elite 2019 SP1, March 2019

Note:

PV Elite performs all calculations internally in Imperial Units to remain compliant with the ASME Code and any built in assumptions in the ASME Code formulas. The finalized results are reflected to show the user's set of selected units.

PV Elite 2019 SP1 Licensee: SPLM Licensed User

FileName : Circular Sections -----

Warnings and Errors: Step: 0 4:14pm Feb 20, 2026

Class From To : Basic Element Checks.

=====

Class From To: Check of Additional Element Data

=====

There were no geometry errors or warnings.

PV Elite is a trademark of Intergraph CADWorx & Analysis Solutions, Inc. 2019

FileName : Circular Sections

Input Echo:

Step: 1 4:14pm Feb 20,2026

PV Elite Vessel Analysis Program: Input Data

Design Internal Pressure (for Hydrotest)	0.5	bars
Design Internal Temperature	170.0	°C
Type of Hydrotest	not Specified	
Hydrotest Position	Horizontal	
Projection of Nozzle from Vessel Top	0	mm.
Projection of Nozzle from Vessel Bottom	0	mm.
Minimum Design Metal Temperature	-28.9	°C
Type of Construction	Welded	
Special Service	None	
Degree of Radiography	RT-1	
Use Higher Longitudinal Stresses (Flag)	Y	
Select t for Internal Pressure (Flag)	N	
Select t for External Pressure (Flag)	N	
Select t for Axial Stress (Flag)	N	
Select Location for Stiff. Rings (Flag)	N	
Consider Vortex Shedding	N	
Perform a Corroded Hydrotest	N	

Load Case 1	NP+EW+WI+FW+BW
Load Case 2	NP+EW+EE+FS+BS
Load Case 3	NP+OW+WI+FW+BW
Load Case 4	NP+OW+EQ+FS+BS
Load Case 5	NP+HW+HI
Load Case 6	NP+HW+HE
Load Case 7	IP+OW+WI+FW+BW
Load Case 8	IP+OW+EQ+FS+BS
Load Case 9	EP+OW+WI+FW+BW
Load Case 10	EP+OW+EQ+FS+BS
Load Case 11	HP+HW+HI
Load Case 12	HP+HW+HE
Load Case 13	IP+WE+EW
Load Case 14	IP+WF+CW
Load Case 15	IP+VO+OW
Load Case 16	IP+VE+EW
Load Case 17	NP+VO+OW
Load Case 18	FS+BS+IP+OW
Load Case 19	FS+BS+EP+OW

Wind Design Code	ASCE-7 93
Basic Wind Speed [V]	112.65 Km/hr
Surface Roughness Category	C: Open Terrain
Importance Factor	1.0
Type of Surface	Moderately Smooth
Base Elevation	0 mm.
Percent Wind for Hydrotest	33.0
Using User defined Wind Press. Vs Elev.	N
Damping Factor (Beta) for Wind (Ope)	0.0100
Damping Factor (Beta) for Wind (Empty)	0.0000
Damping Factor (Beta) for Wind (Filled)	0.0000

Seismic Design Code	UBC 94
UBC Seismic Zone (1=1,2=2a,3=2b,4=3,5=4)	0.000
UBC Importance Factor	1.000
UBC Soil Type	S1
UBC Horizontal Force Factor	3.000
UBC Percent Seismic for Hydrotest	0.000

Design Pressure + Static Head	Y
Consider MAP New and Cold in Noz. Design	N
Consider External Loads for Nozzle Des.	Y
Use ASME VIII-1 Appendix 1-9	N

FileName : Circular Sections

Input Echo:

Step: 1 4:14pm Feb 20,2026

Material Database Year

Current w/Addenda or Code Year

Configuration Directives:

Do not use Nozzle MDMT Interpretation VIII-1 01-37	No
Use Table G instead of exact equation for "A"	Yes
Shell Head Joints are Tapered	Yes
Compute "K" in corroded condition	Yes
Use Code Case 2286	No
Use the MAWP to compute the MDMT	Yes
For thickness ratios ≤ 0.35 , MDMT will be -155F (-104C)	Yes
For PWHT & P1 Materials the MDMT can be $< -55F (-48C)$	No
Using Metric Material Databases, ASME II D	No
Calculate B31.3 type stress for Nozzles with Loads	Yes
Reduce the MDMT due to lower membrane stress	Yes
Consider Longitudinal Stress in MDMT calcs. (Div. 1)	No

Complete Listing of Vessel Elements and Details:

Element From Node	10	
Element To Node	20	
Element Type	Cylinder	
Description		
Distance "FROM" to "TO"	8052	mm.
Inside Diameter	1300	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Allowable Stress, Ambient	137.9	N./mm ²
Allowable Stress, Operating	104.16	N./mm ²
Allowable Stress, Hydrotest	179.27	N./mm ²
Material Density	0.008027	kg./cm ³
P Number Thickness	0	mm.
Yield Stress, Operating	156.11	N./mm ²
External Pressure Chart Name	HA-2	
UNS Number	S31600	
Product Form	Plate	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Weld is pre-Heated	No	

Element From Node	10	
Detail Type	Ring	
Detail ID	Ring:[1 of 1]	
Dist. from "FROM" Node / Offset dist	3958	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1400	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	10	
Detail Type	Ring	
Detail ID	Ring:[2 of 2]	
Dist. from "FROM" Node / Offset dist	5946	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.

FileName : Circular Sections

Input Echo:

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Outside Diameter of Ring	1400	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	10	
Detail Type	Ring	
Detail ID	Ring:[3 of 3]	
Dist. from "FROM" Node / Offset dist	7939	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1400	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	10	
Detail Type	Ring	
Detail ID	Ring:[4 of 4]	
Dist. from "FROM" Node / Offset dist	7952	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1400	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	20	
Element To Node	30	
Element Type	Conical	
Description		
Distance "FROM" to "TO"	2000	mm.
Inside Diameter	1300	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Cone Diameter at "To" End	1730	mm.
Design Length of Cone	2000	mm.
Half Apex Angle of Cone	6.1357331	degrees
Toriconical (Y/N)	N	
Weld is pre-Heated	No	

Element From Node	20	
Detail Type	Ring	
Detail ID	Ring 16	
Dist. from "FROM" Node / Offset dist	0	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1500	mm.
Material Name	SA-240 304	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

FileName : Circular Sections

Input Echo: Step: 1 4:14pm Feb 20,2026

```

Element From Node          30
Element To Node            40
Element Type               Cylinder
Description
Distance "FROM" to "TO"    800 mm.
Inside Diameter            1730 mm.
Element Thickness          3 mm.
Internal Corrosion Allowance 0 mm.
Nominal Thickness          3 mm.
External Corrosion Allowance 0 mm.
Design Internal Pressure    0.5 bars
Design Temperature Internal Pressure 170 °C
Design External Pressure    0.1 bars
Design Temperature External Pressure 50 °C
Effective Diameter Multiplier 1.2
Material Name              SA-240 316
Efficiency, Longitudinal Seam 1.0
Efficiency, Circumferential Seam 1.0
Weld is pre-Heated         No

```

```

Element From Node          30
Detail Type                Ring
Detail ID                  Ring5
Dist. from "FROM" Node / Offset dist 0 mm.
Inside Diameter of Ring    1736 mm.
Thickness of Ring          10 mm.
Outside Diameter of Ring   1830 mm.
Material Name              SA-240 304
Height of Section Ring     0 mm.
Using Custom Stiffener Section No

```

```

Element From Node          40
Element To Node            50
Element Type               Conical
Description
Distance "FROM" to "TO"    1200 mm.
Inside Diameter            1730 mm.
Element Thickness          3 mm.
Internal Corrosion Allowance 0 mm.
Nominal Thickness          3 mm.
External Corrosion Allowance 0 mm.
Design Internal Pressure    0.5 bars
Design Temperature Internal Pressure 170 °C
Design External Pressure    0.1 bars
Design Temperature External Pressure 50 °C
Effective Diameter Multiplier 1.2
Material Name              SA-240 316
Efficiency, Longitudinal Seam 1.0
Efficiency, Circumferential Seam 1.0
Cone Diameter at "To" End  1400 mm.
Design Length of Cone      1200 mm.
Half Apex Angle of Cone    7.8290772 degrees
Toriconical (Y/N)          N
Weld is pre-Heated         No

```

```

Element From Node          40
Detail Type                Ring
Detail ID                  Ring:17
Dist. from "FROM" Node / Offset dist 1199 mm.
Inside Diameter of Ring    1406.3 mm.
Thickness of Ring          10 mm.
Outside Diameter of Ring   1500 mm.
Material Name              SA-240 304
Height of Section Ring     0 mm.

```


FileName : Circular Sections -----

Input Echo: Step: 1 4:14pm Feb 20,2026

Using Custom Stiffener Section	No
Element From Node	40
Detail Type	Ring
Detail ID	Ring 21
Dist. from "FROM" Node / Offset dist	0 mm.
Inside Diameter of Ring	1736 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1836 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	50
Element To Node	60
Element Type	Cylinder
Description	
Distance "FROM" to "TO"	4500 mm.
Inside Diameter	1400 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No

Element From Node	50
Detail Type	Ring
Detail ID	Ring 6
Dist. from "FROM" Node / Offset dist	90 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	50
Detail Type	Ring
Detail ID	Ring 7
Dist. from "FROM" Node / Offset dist	103 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	50
Detail Type	Ring
Detail ID	Ring 8
Dist. from "FROM" Node / Offset dist	1513 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.

FileName : Circular Sections

Input Echo:

Step: 1 4:14pm Feb 20,2026

Using Custom Stiffener Section	No
Element From Node	50
Detail Type	Ring
Detail ID	Ring 9
Dist. from "FROM" Node / Offset dist	2913 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	60
Element To Node	70
Element Type	Conical
Description	
Distance "FROM" to "TO"	2000 mm.
Inside Diameter	1400 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Cone Diameter at "To" End	1730 mm.
Design Length of Cone	2000 mm.
Half Apex Angle of Cone	4.7162204 degrees
Toriconical (Y/N)	N
Weld is pre-Heated	No

Element From Node	60
Detail Type	Ring
Detail ID	Ring 18
Dist. from "FROM" Node / Offset dist	0 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	60
Detail Type	Ring
Detail ID	Ring 23
Dist. from "FROM" Node / Offset dist	1999 mm.
Inside Diameter of Ring	1735.8 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1829.8 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	70
Element To Node	80
Element Type	Cylinder

FileName : Circular Sections

Input Echo:

Step: 1 4:14pm Feb 20,2026

Description	
Distance "FROM" to "TO"	800 mm.
Inside Diameter	1730 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No

Element From Node	70
Detail Type	Ring
Detail ID	Ring 10
Dist. from "FROM" Node / Offset dist	737 mm.
Inside Diameter of Ring	1736 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1830 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	70
Detail Type	Ring
Detail ID	Ring 11
Dist. from "FROM" Node / Offset dist	750 mm.
Inside Diameter of Ring	1736 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1830 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	80
Element To Node	90
Element Type	Conical
Description	
Distance "FROM" to "TO"	1200 mm.
Inside Diameter	1730 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Cone Diameter at "To" End	1400 mm.
Design Length of Cone	1200 mm.
Half Apex Angle of Cone	7.8290772 degrees
Toriconical (Y/N)	N
Weld is pre-Heated	No

Element From Node	80
-------------------	----

FileName : Circular Sections -----

Input Echo: Step: 1 4:14pm Feb 20,2026

Detail Type	Ring
Detail ID	Ring 24
Dist. from "FROM" Node / Offset dist	0 mm.
Inside Diameter of Ring	1736 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1829.7 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	80
Detail Type	Ring
Detail ID	Ring 26
Dist. from "FROM" Node / Offset dist	1199 mm.
Inside Diameter of Ring	1406.3 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1499.9 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	90
Element To Node	100
Element Type	Cylinder
Description	
Distance "FROM" to "TO"	3500 mm.
Inside Diameter	1400 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No

Element From Node	90
Detail Type	Ring
Detail ID	Ring 12
Dist. from "FROM" Node / Offset dist	375 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	90
Detail Type	Ring
Detail ID	Ring 13]
Dist. from "FROM" Node / Offset dist	1745 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

FileName : Circular Sections

Input Echo:

Step: 1 4:14pm Feb 20,2026

Element From Node	100
Element To Node	110
Element Type	Conical
Description	
Distance "FROM" to "TO"	2000 mm.
Inside Diameter	1400 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Cone Diameter at "To" End	950 mm.
Design Length of Cone	2000 mm.
Half Apex Angle of Cone	6.4187856 degrees
Toriconical (Y/N)	N
Weld is pre-Heated	No

Element From Node	100
Detail Type	Ring
Detail ID	Ring 22
Dist. from "FROM" Node / Offset dist	1999 mm.
Inside Diameter of Ring	956.22 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1057.1 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	100
Detail Type	Ring
Detail ID	Ring 28
Dist. from "FROM" Node / Offset dist	0 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1506.8 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

FileName : Circular Sections

XY Coordinate Calculations: Step: 2 4:14pm Feb 20,2026

XY Coordinate Calculations:

From	To	X (Horiz.) mm.	Y (Vert.) mm.	DX (Horiz.) mm.	DY (Vert.) mm.
10	20	...	8052	...	8052
20	30	...	10052	...	2000
30	40	...	10852	...	800
40	50	...	12052	...	1200
50	60	...	16552	...	4500
60	70	...	18552	...	2000
70	80	...	19352	...	800
80	90	...	20552	...	1200
90	100	...	24052	...	3500
100	110	...	26052	...	2000

FileName : Circular Sections

Internal Pressure Calculations: Step: 13 4:14pm Feb 20,2026

Element Thickness, Pressure, Diameter and Allowable Stress :

From	To	Int. Press + Liq. Hd bars	Nominal Thickness mm.	Total Corr Allowance mm.	Element Diameter mm.	Allowable Stress (SE) N./mm ²
10	20	0.5	3	...	1300	104.16
20	30	0.5	3	...	1730	104.16
30	40	0.5	3	...	1730	104.16
40	50	0.5	3	...	1730	104.16
50	60	0.5	3	...	1400	104.16
60	70	0.5	3	...	1730	104.16
70	80	0.5	3	...	1730	104.16
80	90	0.5	3	...	1730	104.16
90	100	0.5	3	...	1400	104.16
100	110	0.5	3	...	1400	104.16

Element Required Thickness and MAWP :

From	To	Design Pressure bars	M.A.W.P. Corroded bars	M.A.P. New & Cold bars	Minimum Thickness mm.	Required Thickness mm.
10	20	0.5	4.79375	6.34667	3	1.5
20	30	0.5	3.58411	4.74517	3	1.5
30	40	0.5	3.60472	4.77245	3	1.5
40	50	0.5	3.57118	4.72806	3	1.5
50	60	0.5	4.45222	5.8945	3	1.5
60	70	0.5	3.59254	4.75633	3	1.5
70	80	0.5	3.60472	4.77245	3	1.5
80	90	0.5	3.57118	4.72806	3	1.5
90	100	0.5	4.45222	5.8945	3	1.5
100	110	0.5	4.42438	5.85764	3	1.5
Minimum			3.571	4.728		

MAWP: 3.571 bars, limited by: Cone.

Internal Pressure Calculation Results :**ASME Code, Section VIII Division 1, 2017****Cylindrical Shell From 10 To 20 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 \cdot 650.0) / (104.16 \cdot 1.0 - 0.6 \cdot 0.5) \\
 &= 0.3121 + 0.0000 = 0.3121 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (104.16 \cdot 1.0 \cdot 3.0) / (650.0 + 0.6 \cdot 3.0) \\
 &= 4.794 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (650.0 + 0.6 \cdot 3.0)
 \end{aligned}$$

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FileName : Circular Sections -----

Internal Pressure Calculations: Step: 13 4:14pm Feb 20, 2026

$$= 6.347 \text{ bars}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned} &= (P * (R + 0.6 * t)) / (E * t) \\ &= (0.5 * (650.0 + 0.6 * 3.0)) / (1.0 * 3.0) \\ &= 10.864 \text{ N./mm}^2 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 * t_{nom} / R_f) * (1 - R_f / R_o)$ 0.230 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Conical Section From 20 To 30 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned} &= (P * D) / (2 * \cos(a) * (S * E - 0.6 * P)) \text{ per Appendix 1-4 (e)} \\ &= (0.5 * 1730.0) / (2 * 0.9943 * (104.16 * 1.0 - 0.6 * 0.5)) \\ &= 0.4178 + 0.0000 = 0.4178 \text{ mm.} \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned} &= (2 * S * E * t * \cos(a)) / (D + 1.2 * t * \cos(a)) \text{ per App 1-4 (e)} \\ &= (2 * 104.16 * 1.0 * 3.0 * 0.994) / (1730.0 + 1.2 * 3.0 * 0.994) \\ &= 3.584 \text{ bars} \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned} &= (2 * S * E * t * \cos(a)) / (D + 1.2 * t * \cos(a)) \text{ per App 1-4 (e)} \\ &= (2 * 137.9 * 1.0 * 3.0 * 0.9943) / (1730.0 + 1.2 * 3.0 * 0.9943) \\ &= 4.745 \text{ bars} \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned} &= (P * (D + 1.2 * t * \cos(a))) / (2 * E * t * \cos(a)) \\ &= (0.5 * (1730.0 + 1.2 * 3.0 * 0.9943)) / (2 * 1.0 * 3.0 * 0.9943) \\ &= 14.531 \text{ N./mm}^2 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 * t_{nom} / R_f) * (1 - R_f / R_o)$ 0.230 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

Note: The Pressure at the Large Diameter is used in the TR calculation.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Cylindrical Shell From 30 To 40 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned} &= (P * R) / (S * E - 0.6 * P) \text{ per UG-27 (c) (1)} \\ &= (0.5 * 865.0) / (104.16 * 1.0 - 0.6 * 0.5) \\ &= 0.4154 + 0.0000 = 0.4154 \text{ mm.} \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned} &= (S * E * t) / (R + 0.6 * t) \text{ per UG-27 (c) (1)} \\ &= (104.16 * 1.0 * 3.0) / (865.0 + 0.6 * 3.0) \\ &= 3.605 \text{ bars} \end{aligned}$$

FileName : Circular Sections -----

Internal Pressure Calculations: Step: 13 4:14pm Feb 20, 2026

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (865.0 + 0.6 \cdot 3.0) \\
 &= 4.772 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t) \\
 &= (0.5 \cdot (865.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0) \\
 &= 14.448 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.173 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Conical Section From 40 To 50 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot D) / (2 \cdot \cos(a) \cdot (S \cdot E - 0.6 \cdot P)) \text{ per Appendix 1-4 (e)} \\
 &= (0.5 \cdot 1730.0) / (2 \cdot 0.9907 \cdot (104.16 \cdot 1.0 - 0.6 \cdot 0.5)) \\
 &= 0.4193 + 0.0000 = 0.4193 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 104.16 \cdot 1.0 \cdot 3.0 \cdot 0.991) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.991) \\
 &= 3.571 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 137.9 \cdot 1.0 \cdot 3.0 \cdot 0.9907) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.9907) \\
 &= 4.728 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (D + 1.2 \cdot t \cdot \cos(a))) / (2 \cdot E \cdot t \cdot \cos(a)) \\
 &= (0.5 \cdot (1730.0 + 1.2 \cdot 3.0 \cdot 0.9907)) / (2 \cdot 1.0 \cdot 3.0 \cdot 0.9907) \\
 &= 14.583 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Cylindrical Shell From 50 To 60 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 \cdot 700.0) / (104.16 \cdot 1.0 - 0.6 \cdot 0.5) \\
 &= 0.3361 + 0.0000 = 0.3361 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (104.16 \cdot 1.0 \cdot 3.0) / (700.0 + 0.6 \cdot 3.0)
 \end{aligned}$$

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FileName : Circular Sections -----

Internal Pressure Calculations: Step: 13 4:14pm Feb 20, 2026

$$= 4.452 \text{ bars}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned} &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\ &= (137.9 \cdot 1.0 \cdot 3.0) / (700.0 + 0.6 \cdot 3.0) \\ &= 5.894 \text{ bars} \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned} &= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t) \\ &= (0.5 \cdot (700.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0) \\ &= 11.697 \text{ N./mm}^2 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Conical Section From 60 To 70 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned} &= (P \cdot D) / (2 \cdot \cos(a) \cdot (S \cdot E - 0.6 \cdot P)) \text{ per Appendix 1-4 (e)} \\ &= (0.5 \cdot 1730.0) / (2 \cdot 0.9966 \cdot (104.16 \cdot 1.0 - 0.6 \cdot 0.5)) \\ &= 0.4168 + 0.0000 = 0.4168 \text{ mm.} \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned} &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\ &= (2 \cdot 104.16 \cdot 1.0 \cdot 3.0 \cdot 0.997) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.997) \\ &= 3.593 \text{ bars} \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned} &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\ &= (2 \cdot 137.9 \cdot 1.0 \cdot 3.0 \cdot 0.9966) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.9966) \\ &= 4.756 \text{ bars} \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned} &= (P \cdot (D + 1.2 \cdot t \cdot \cos(a))) / (2 \cdot E \cdot t \cdot \cos(a)) \\ &= (0.5 \cdot (1730.0 + 1.2 \cdot 3.0 \cdot 0.9966)) / (2 \cdot 1.0 \cdot 3.0 \cdot 0.9966) \\ &= 14.496 \text{ N./mm}^2 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

Note: The Pressure at the Large Diameter is used in the TR calculation.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Cylindrical Shell From 70 To 80 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned} &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\ &= (0.5 \cdot 865.0) / (104.16 \cdot 1.0 - 0.6 \cdot 0.5) \\ &= 0.4154 + 0.0000 = 0.4154 \text{ mm.} \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

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Internal Pressure Calculations: Step: 13 4:14pm Feb 20, 2026

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)}$$

$$= (104.16 \cdot 1.0 \cdot 3.0) / (865.0 + 0.6 \cdot 3.0)$$

$$= 3.605 \text{ bars}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)}$$

$$= (137.9 \cdot 1.0 \cdot 3.0) / (865.0 + 0.6 \cdot 3.0)$$

$$= 4.772 \text{ bars}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t)$$

$$= (0.5 \cdot (865.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0)$$

$$= 14.448 \text{ N./mm}^2$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.173 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Conical Section From 80 To 90 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$= (P \cdot D) / (2 \cdot \cos(a) \cdot (S \cdot E - 0.6 \cdot P)) \text{ per Appendix 1-4 (e)}$$

$$= (0.5 \cdot 1730.0) / (2 \cdot 0.9907 \cdot (104.16 \cdot 1.0 - 0.6 \cdot 0.5))$$

$$= 0.4193 + 0.0000 = 0.4193 \text{ mm.}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)}$$

$$= (2 \cdot 104.16 \cdot 1.0 \cdot 3.0 \cdot 0.991) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.991)$$

$$= 3.571 \text{ bars}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)}$$

$$= (2 \cdot 137.9 \cdot 1.0 \cdot 3.0 \cdot 0.9907) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.9907)$$

$$= 4.728 \text{ bars}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$= (P \cdot (D + 1.2 \cdot t \cdot \cos(a))) / (2 \cdot E \cdot t \cdot \cos(a))$$

$$= (0.5 \cdot (1730.0 + 1.2 \cdot 3.0 \cdot 0.9907)) / (2 \cdot 1.0 \cdot 3.0 \cdot 0.9907)$$

$$= 14.583 \text{ N./mm}^2$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Cylindrical Shell From 90 To 100 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)}$$

$$= (0.5 \cdot 700.0) / (104.16 \cdot 1.0 - 0.6 \cdot 0.5)$$

$$= 0.3361 + 0.0000 = 0.3361 \text{ mm.}$$

Note: The thickness required was less than the Code Minimum, therefore

FileName : Circular Sections -----

Internal Pressure Calculations: Step: 13 4:14pm Feb 20, 2026

the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (104.16 \cdot 1.0 \cdot 3.0) / (700.0 + 0.6 \cdot 3.0) \\
 &= 4.452 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (700.0 + 0.6 \cdot 3.0) \\
 &= 5.894 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t) \\
 &= (0.5 \cdot (700.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0) \\
 &= 11.697 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Conical Section From 100 To 110 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot D) / (2 \cdot \cos(a) \cdot (S \cdot E - 0.6 \cdot P)) \text{ per Appendix 1-4 (e)} \\
 &= (0.5 \cdot 1400.0) / (2 \cdot 0.9937 \cdot (104.16 \cdot 1.0 - 0.6 \cdot 0.5)) \\
 &= 0.3383 + 0.0000 = 0.3383 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 104.16 \cdot 1.0 \cdot 3.0 \cdot 0.994) / (1400.0 + 1.2 \cdot 3.0 \cdot 0.994) \\
 &= 4.424 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 137.9 \cdot 1.0 \cdot 3.0 \cdot 0.9937) / (1400.0 + 1.2 \cdot 3.0 \cdot 0.9937) \\
 &= 5.858 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (D + 1.2 \cdot t \cdot \cos(a))) / (2 \cdot E \cdot t \cdot \cos(a)) \\
 &= (0.5 \cdot (1400.0 + 1.2 \cdot 3.0 \cdot 0.9937)) / (2 \cdot 1.0 \cdot 3.0 \cdot 0.9937) \\
 &= 11.771 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.315 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Elements Suitable for Internal Pressure.

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FileName : Circular Sections

External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

External Pressure Calculation Results :**External Pressure Calculations:**

From	To	Section Length mm.	Outside Diameter mm.	Corroded Thickness mm.	Factor A	Factor B N./mm ²
10	Ring	3958	1306	3	0.46719E-04	4.48502
Ring	Ring	1988	1306	3	0.95296E-04	9.14839
Ring	Ring	1993	1306	3	0.95050E-04	9.12477
Ring	Ring	12.9998	1306	3	0.0047922	91.4355
Ring	20	100	1306	3	0.0026	83.9018
20	Ring	1679.94	1306.03	3	0.00011239	10.7893
Ring	30	1752.3	1736.03	3	0.93196E-04	8.94674
30	Ring	No Calc	...	3	No Calc	No Calc
Ring	40	800	1736	3	0.00021156	20.3093
40	Ring	1752.3	1736.06	3	0.92641E-04	8.89351
Ring	Ring	1085.13	1736.06	3	0.00015138	14.5319
Ring	50	0.99994	1406.33	3	0.0041294	89.7316
50	Ring	90	1406	3	0.0029414	85.5356
Ring	Ring	13	1406	3	0.0042047	89.9431
Ring	Ring	1410	1406	3	0.000131	12.5764
Ring	Ring	1400	1406	3	0.00013196	12.6681
Ring	60	1587	1406	3	0.00011601	11.1366
60	Ring	0.99994	1406.02	3	0.004178	89.8686
Ring	Ring	1809.08	1735.86	3	0.90550E-04	8.69271
Ring	70	0.99989	1736.02	3	0.0028755	85.24
70	Ring	737	1736	3	0.00023058	22.1352
Ring	Ring	13	1736	3	0.0028938	85.3229
Ring	80	50	1736	3	0.0028938	85.3229
80	Ring	0.99989	1736.06	3	0.0028434	85.0923
Ring	Ring	1085.13	1736.06	3	0.00015138	14.5319
Ring	90	0.99994	1406.33	3	0.0041294	89.7316
90	Ring	375	1406	3	0.00052804	50.6919
Ring	Ring	1370	1406	3	0.0001349	12.9498
Ring	100	1755	1406	3	0.00010461	10.0421
100	Ring	0.99994	1406.04	3	0.0041551	89.8045
Ring	Ring	1679.26	1406.04	3	0.00010837	10.4036
Ring	110	0.99982	956.263	3	0.0083163	97.0162

External Pressure Calculations:

From	To	External Actual T. mm.	External Required T. mm.	External Design Pressure bars	External M.A.W.P. bars
10	Ring	3	2.64879	0.1	0.13736
Ring	Ring	3	1.98494	0.1	0.28018
Ring	Ring	3	1.98708	0.1	0.27946
Ring	Ring	3	0.47257	0.1	2.80031
Ring	20	3	0.57319	0.1	2.56958
20	Ring	3	1.86049	0.1	0.32853
Ring	30	3	2.24961	0.1	0.20495
30	Ring	3	No Calc	0.1	No Calc
Ring	40	3	1.62103	0.1	0.46793
40	Ring	3	2.25778	0.1	0.20299
Ring	Ring	3	1.85593	0.1	0.33169
Ring	50	3	0.51366	0.1	2.52828
50	Ring	3	0.56956	0.1	2.4333
Ring	Ring	3	0.50876	0.1	2.55869
Ring	Ring	3	1.80715	0.1	0.35777
Ring	Ring	3	1.80239	0.1	0.36038
Ring	60	3	1.89031	0.1	0.31681

FileName : Circular Sections

External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

60	Ring	3	0.51049	0.1	2.54788
Ring	Ring	3	2.27117	0.1	0.19962
Ring	70	3	0.6303	0.1	1.95727
70	Ring	3	1.56614	0.1	0.51
Ring	Ring	3	0.62816	0.1	1.96585
Ring	80	3	0.62816	0.1	1.96585
80	Ring	3	0.63409	0.1	1.9422
Ring	Ring	3	1.85593	0.1	0.33169
Ring	90	3	0.51366	0.1	2.52828
90	Ring	3	1.04348	0.1	1.44208
Ring	Ring	3	1.78809	0.1	0.3684
Ring	100	3	1.96728	0.1	0.28568
100	Ring	3	0.51198	0.1	2.53866
Ring	Ring	3	1.94504	0.1	0.2941
Ring	110	3	0.34821	0.1	4.03247

Minimum

0.137

External Pressure Calculations:

From	To	Actual Length Bet. Stiffeners mm.	Allowable Length Bet. Stiffeners mm.	Ring Inertia Required cm**4	Ring Inertia Available cm**4
10	Ring	3958	5366.66	No Calc	No Calc
Ring	Ring	1988	5367.77	4.74696	17.6358
Ring	Ring	1993	5367.78	3.17821	17.6358
Ring	Ring	12.9998	362.283	1.60148	17.6358
Ring	20	100	3105.85	0.090213	17.6358
20	Ring	1679.94	1679.94	No Calc	No Calc
Ring	30	1752.3	1752.3	2.74012	118.645
30	Ring	No Calc	No Calc	No Calc	No Calc
Ring	40	800	3576.41	1.50003	18.5486
40	Ring	1752.3	1752.3	No Calc	No Calc
Ring	Ring	1085.13	1085.13	5.32031	21.7628
Ring	50	0.99994	0.99994	1.08257	17.7314
50	Ring	90	2782.45	No Calc	No Calc
Ring	Ring	13	332.63	0.1026	17.8679
Ring	Ring	1410	4840.94	1.41706	17.8679
Ring	Ring	1400	4840.92	2.80393	17.8679
Ring	60	1587	4841.33	2.96283	17.8679
60	Ring	0.99994	0.99994	No Calc	No Calc
Ring	Ring	1809.08	1809.08	1.80308	17.8679
Ring	70	0.99989	0.99989	3.39301	18.5482
70	Ring	737	3576.31	No Calc	No Calc
Ring	Ring	13	979.513	1.41118	18.5486
Ring	80	50	1799.98	0.11813	18.5486
80	Ring	0.99989	0.99989	No Calc	No Calc
Ring	Ring	1085.13	1085.13	2.0298	18.381
Ring	90	0.99994	0.99994	1.08257	17.7021
90	Ring	375	4840.98	No Calc	No Calc
Ring	Ring	1370	4840.85	1.73585	17.8679
Ring	100	1755	4841.66	3.11292	17.8679
100	Ring	0.99994	0.99994	No Calc	No Calc
Ring	Ring	1679.26	1679.26	1.67377	21.4269
Ring	110	0.99982	0.99982	0.52653	20.0339

Elements Suitable for External Pressure.

ASME Code, Section VIII Division 1, 2017**Cylindrical Shell From 10 to Ring:[1 of 1] Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

FileName : Circular Sections

External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	3958.00	435.33	3.0306	0.0000467	4.49
EMAP = $(4*B)/(3*(D/t)) = (4*4.485)/(3*435.3333) = 0.1374$ bars						

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.649	1306.00	3958.00	493.06	3.0306	0.0000385	3.70
EMAP = $(4*B)/(3*(D/t)) = (4*3.6983)/(3*493.0556) = 0.1$ bars						

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	5366.66	435.33	4.1092	0.0000340	3.27
EMAP = $(4*B)/(3*(D/t)) = (4*3.268)/(3*435.3333) = 0.1001$ bars						

Cylindrical Shell From Ring:[1 of 1] to Ring:[2 of 2] Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	1988.00	435.33	1.5222	0.0000953	9.15
EMAP = $(4*B)/(3*(D/t)) = (4*9.1484)/(3*435.3333) = 0.2802$ bars						

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.985	1306.00	1988.00	657.96	1.5222	0.0000514	4.94
EMAP = $(4*B)/(3*(D/t)) = (4*4.9351)/(3*657.9558) = 0.1$ bars						

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	5367.77	435.33	4.1101	0.0000340	3.27
EMAP = $(4*B)/(3*(D/t)) = (4*3.2673)/(3*435.3333) = 0.1001$ bars						

Cylindrical Shell From Ring:[2 of 2] to Ring:[3 of 3] Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	1993.00	435.33	1.5260	0.0000951	9.12
EMAP = $(4*B)/(3*(D/t)) = (4*9.1248)/(3*435.3333) = 0.2795$ bars						

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.987	1306.00	1993.00	657.25	1.5260	0.0000514	4.93
EMAP = $(4*B)/(3*(D/t)) = (4*4.9298)/(3*657.2463) = 0.1$ bars						

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	5367.78	435.33	4.1101	0.0000340	3.27
EMAP = $(4*B)/(3*(D/t)) = (4*3.2673)/(3*435.3333) = 0.1001$ bars						

Cone From Ring 16 to 30 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.983	1736.03	1752.30	582.01	1.0094	0.0000932	8.95
EMAP = $(4*B)/(3*(D/t)) = (4*8.9467)/(3*582.0123) = 0.2049$ bars						

Note: The cone thickness used in the calculation has been modified
per UG-33(f), $t_e = t * \cos(\alpha)$.

FileName : Circular Sections

External Pressure Calculations: Step: 14 4:14pm Feb 20, 2026

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.237	1736.03	1752.30	776.15	1.0094	0.000606	5.82

EMAP = (4*B)/(3*(D/t)) = (4*5.8216)/(3*776.1512) = 0.1 bars

Note: The cone thickness used in the calculation has been modified per UG-33(f), $t_e = t \cdot \cos(\alpha)$.

Cylindrical Shell From Ring5 to 40 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1736.00	800.00	578.67	0.4608	0.0002116	20.31
EMAP = (4*B)/(3*(D/t)) = (4*20.3093)/(3*578.6667) = 0.4679 bars						

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.621	1736.00	800.00	1070.92	0.4608	0.0000837	8.03
EMAP = (4*B)/(3*(D/t)) = (4*8.0326)/(3*1070.9225) = 0.1 bars						

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1736.00	3576.41	578.67	2.0601	0.0000452	4.34
EMAP = (4*B)/(3*(D/t)) = (4*4.3427)/(3*578.6667) = 0.1001 bars						

Cone From Ring 21 to Ring:17 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.972	1736.06	1085.13	584.13	0.6251	0.0001514	14.53
EMAP = (4*B)/(3*(D/t)) = (4*14.5319)/(3*584.1302) = 0.3317 bars						

Note: The cone thickness used in the calculation has been modified per UG-33(f), $t_e = t \cdot \cos(\alpha)$.

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.839	1736.06	1085.13	944.21	0.6251	0.0000738	7.08
EMAP = (4*B)/(3*(D/t)) = (4*7.0822)/(3*944.2127) = 0.1 bars						

Note: The cone thickness used in the calculation has been modified per UG-33(f), $t_e = t \cdot \cos(\alpha)$.

Cone From Ring:17 to 50 Ext. Chart: HA-2 at 50 °C

Cylindrical Shell From Ring 7 to Ring 8 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1410.00	468.67	1.0028	0.0001310	12.58

EMAP = (4*B)/(3*(D/t)) = (4*12.5764)/(3*468.6667) = 0.3578 bars

Results for Required Thickness (Tca):

Tca	OD	SLen	D/t	L/D	Factor A	B
1.807	1406.00	1410.00	778.02	1.0028	0.0000608	5.84

EMAP = (4*B)/(3*(D/t)) = (4*5.8358)/(3*778.0202) = 0.1 bars

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4840.94	468.67	3.4431	0.0000366	3.52

EMAP = $(4*B)/(3*(D/t)) = (4*3.5182)/(3*468.6667) = 0.1001$ bars

Cylindrical Shell From Ring 8 to Ring 9 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1400.00	468.67	0.9957	0.0001320	12.67

EMAP = $(4*B)/(3*(D/t)) = (4*12.6681)/(3*468.6667) = 0.3604$ bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.802	1406.00	1400.00	780.07	0.9957	0.0000610	5.85

EMAP = $(4*B)/(3*(D/t)) = (4*5.8512)/(3*780.0737) = 0.1$ bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4840.92	468.67	3.4430	0.0000366	3.52

EMAP = $(4*B)/(3*(D/t)) = (4*3.5182)/(3*468.6667) = 0.1001$ bars

Cylindrical Shell From Ring 9 to 60 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1587.00	468.67	1.1287	0.0001160	11.14

EMAP = $(4*B)/(3*(D/t)) = (4*11.1366)/(3*468.6667) = 0.3168$ bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.890	1406.00	1587.00	743.79	1.1287	0.0000581	5.58

EMAP = $(4*B)/(3*(D/t)) = (4*5.5789)/(3*743.7935) = 0.1$ bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4841.33	468.67	3.4433	0.0000366	3.52

EMAP = $(4*B)/(3*(D/t)) = (4*3.5179)/(3*468.6667) = 0.1001$ bars

Cone From Ring 18 to Ring 23 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.990	1735.86	1809.08	580.58	1.0422	0.0000905	8.69

EMAP = $(4*B)/(3*(D/t)) = (4*8.6927)/(3*580.5842) = 0.1996$ bars

Note: The cone thickness used in the calculation has been modified
per UG-33(f), $t_e = t * \cos(\alpha)$.

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.263	1735.86	1809.08	766.90	1.0422	0.0000599	5.75

EMAP = $(4*B)/(3*(D/t)) = (4*5.7522)/(3*766.8951) = 0.1$ bars

Note: The cone thickness used in the calculation has been modified
per UG-33(f), $t_e = t * \cos(\alpha)$.

Cone From Ring 23 to 70 Ext. Chart: HA-2 at 50 °C

FileName : Circular Sections

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Cylindrical Shell From 70 to Ring 10 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1736.00	737.00	578.67	0.4245	0.0002306	22.14

EMAP = (4*B) / (3*(D/t)) = (4*22.1352) / (3*578.6667) = 0.51 bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.566	1736.00	737.00	1108.46	0.4245	0.0000866	8.31

EMAP = (4*B) / (3*(D/t)) = (4*8.3142) / (3*1108.459) = 0.1 bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1736.00	3576.31	578.67	2.0601	0.0000452	4.34

EMAP = (4*B) / (3*(D/t)) = (4*4.3428) / (3*578.6667) = 0.1001 bars

Cone From Ring 24 to Ring 26 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.972	1736.06	1085.13	584.13	0.6251	0.0001514	14.53

EMAP = (4*B) / (3*(D/t)) = (4*14.5319) / (3*584.1302) = 0.3317 bars

Note: The cone thickness used in the calculation has been modified
per UG-33(f), $t_e = t * \cos(\alpha)$.

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.839	1736.06	1085.13	944.21	0.6251	0.0000738	7.08

EMAP = (4*B) / (3*(D/t)) = (4*7.0822) / (3*944.2127) = 0.1 bars

Note: The cone thickness used in the calculation has been modified
per UG-33(f), $t_e = t * \cos(\alpha)$.

Cone From Ring 26 to 90 Ext. Chart: HA-2 at 50 °C**Cylindrical Shell From 90 to Ring 12 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	375.00	468.67	0.2667	0.0005280	50.69

EMAP = (4*B) / (3*(D/t)) = (4*50.6919) / (3*468.6667) = 1.4421 bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.043	1406.00	375.00	1347.41	0.2667	0.0001053	10.11

EMAP = (4*B) / (3*(D/t)) = (4*10.1068) / (3*1347.4142) = 0.1 bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4840.98	468.67	3.4431	0.0000366	3.52

EMAP = (4*B) / (3*(D/t)) = (4*3.5182) / (3*468.6667) = 0.1001 bars

Cylindrical Shell From Ring 12 to Ring 13] Ext. Chart: HA-2 at 50 °C

FileName : Circular Sections -----

External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1370.00	468.67	0.9744	0.0001349	12.95

EMAP = $(4*B)/(3*(D/t)) = (4*12.9498)/(3*468.6667) = 0.3684$ bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.788	1406.00	1370.00	786.31	0.9744	0.0000614	5.90

EMAP = $(4*B)/(3*(D/t)) = (4*5.898)/(3*786.3148) = 0.1$ bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4840.85	468.67	3.4430	0.0000366	3.52

EMAP = $(4*B)/(3*(D/t)) = (4*3.5183)/(3*468.6667) = 0.1001$ bars

Cylindrical Shell From Ring 13] to 100 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1755.00	468.67	1.2482	0.0001046	10.04

EMAP = $(4*B)/(3*(D/t)) = (4*10.0421)/(3*468.6667) = 0.2857$ bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.967	1406.00	1755.00	714.69	1.2482	0.0000558	5.36

EMAP = $(4*B)/(3*(D/t)) = (4*5.3608)/(3*714.6927) = 0.1$ bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4841.66	468.67	3.4436	0.0000366	3.52

EMAP = $(4*B)/(3*(D/t)) = (4*3.5177)/(3*468.6667) = 0.1001$ bars

Cone From Ring 28 to Ring 22 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.981	1406.04	1679.26	471.64	1.1943	0.0001084	10.40

EMAP = $(4*B)/(3*(D/t)) = (4*10.4036)/(3*471.6358) = 0.2941$ bars

Note: The cone thickness used in the calculation has been modified
per UG-33(f), $t_e = t * \cos(\alpha)$.

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.933	1406.04	1679.26	727.44	1.1943	0.0000568	5.46

EMAP = $(4*B)/(3*(D/t)) = (4*5.4564)/(3*727.4449) = 0.1$ bars

Note: The cone thickness used in the calculation has been modified
per UG-33(f), $t_e = t * \cos(\alpha)$.

Cone From Ring 22 to the end: Ext. Chart: HA-2 at 50 °C**Stiffening Ring Calcs for : Ring:[1 of 1] , SA-240 316 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 69 mm.

Area (cm ²)	Distance (mm.)	Area*Dist
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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Shell:	2.066	1.5000	3.098
Ring :	4.700	26.5000	124.550
Total:	6.766		127.648

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.015	17.3673	623.030
Ring :	8.652	-7.6327	273.815
Total:	8.667		896.845

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell	Breq	3.10 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000323

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1306.0^2 * 2973.0002(3.0 + 4.7/2973.0002)0.000032) / 10.9$$

$$= 5 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	3.03	Kgf/mm.
The Radial Shear Load	V	39.59	Kgf
The First Moment of the Area (Ring + Shell)	Q	3587.38	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.81	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.59	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	45.03	Kgf/mm.
The Combined Weld Load SRSS of VQ/I and Pext*Slen		3.14	Kgf/mm.

Stiffening Ring Calcs for : Ring:[2 of 2], SA-240 316 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 69 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.066	1.5000	3.098
Ring :	4.700	26.5000	124.550
Total:	6.766		127.648

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.015	17.3673	623.030
Ring :	8.652	-7.6327	273.815
Total:	8.667		896.845

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell	Breq	3.03 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000315

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1306.0^2 * 1990.5(3.0 + 4.7/1990.5)0.000032) / 10.9$$

$$= 3 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
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FileName : Circular Sections

External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	2.03	Kgf/mm.
The Radial Shear Load	V	26.51	Kgf
The First Moment of the Area (Ring + Shell)	Q	3587.38	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.54	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.59	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	45.03	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	2.10	Kgf/mm.

Stiffening Ring Calcs for : Ring:[3 of 3], SA-240 316 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 69 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.066	1.5000	3.098
Ring :	4.700	26.5000	124.550
Total:	6.766		127.648

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.015	17.3673	623.030
Ring :	8.652	-7.6327	273.815
Total:	8.667		896.845

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell	Breq	2.82	N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000294	

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq)/10.9$$

$$= (1306.0^2 * 1003.0(3.0 + 4.7/1003.0)0.0000294)/10.9$$

$$= 2 \text{ cm**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	1.02	Kgf/mm.
The Radial Shear Load	V	13.36	Kgf
The First Moment of the Area (Ring + Shell)	Q	3587.38	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.27	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.59	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	45.03	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	1.06	Kgf/mm.

Stiffening Ring Calcs for : Ring:[4 of 4], SA-240 316 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 69 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.066	1.5000	3.098
Ring :	4.700	26.5000	124.550
Total:	6.766		127.648

Centroid of Ring plus Shell: 19 mm.

Inertia	Distance	A*Dist ²
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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Shell:	0.015	17.3673	623.030
Ring :	8.652	-7.6327	273.815
Total:	8.667		896.845

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell	Breq	0.87 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000090

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1306.0^2 * 56.5(3.0 + 4.7/56.5)0.000009) / 10.9$$

$$= 0 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.06	Kgf/mm.
The Radial Shear Load	V	0.75	Kgf
The First Moment of the Area (Ring + Shell)	Q	3587.38	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.02	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.59	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	45.03	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.06	Kgf/mm.

Stiffening Ring Calcs for : Ring 16 , SA-240 304 , Bar Ring: 97 x 10 mm.

Effective Length of Shell: 69 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.066	1.5000	3.098
Ring :	9.700	51.5000	499.550
Total:	11.766		502.648

Centroid of Ring plus Shell: 43 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.015	41.2219	3509.954
Ring :	76.056	-8.7781	747.439
Total:	76.071		4257.393

Available Moment of Inertia, Ring plus Shell: 119 cm**4

Required Stress in Ring plus Shell	Breq	2.75 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000286

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1306.0^2 * 1716.1235(3.0 + 9.7/1716.1235)0.000029) / 10.9$$

$$= 3 \text{ cm}^4$$

Stiffening Ring Calcs for : Ring5 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 79 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	4.700	26.5000	124.550
Total:	7.081		128.122

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.018	16.5926	655.656
Ring :	8.652	-8.4075	332.221
Total:	8.670		987.878

Available Moment of Inertia, Ring plus Shell: 19 cm**4

Required Stress in Ring plus Shell Breq 3.12 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000325

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1736.0^2 * 400.0(3.0 + 4.7/400.0)0.000032) / 10.9$$

$$= 2 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style CONTINUOUS
 Location of Stiffening Ring EXTERNAL
 Radial Pressure Load Pext*Slen 0.41 Kgf/mm.
 The Radial Shear Load V 7.08 Kgf
 The First Moment of the Area (Ring + Shell) Q 3951.51 mm.³
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.15 Kgf/mm.
 The Weld Allowable Stress 0.55*S 73.09 N./mm²
 Minimum Weld Leg Size Min(6mm, t, tw) Wldmin 3.00 mm.
 The Weld Allowable Load 2* WLeg*0.55*S 44.72 Kgf/mm.
 The Combined Weld Load SRSS of VQ/I and Pext*Slen 0.43 Kgf/mm.

Stiffening Ring Calcs for : Ring 21 , SA-240 304 , Bar Ring: 50 x 10 mm.

Effective Length of Shell: 79 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	5.000	28.0000	140.000
Total:	7.381		143.572

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.018	17.9503	767.347
Ring :	10.417	-8.5497	365.487
Total:	10.434		1132.833

Available Moment of Inertia, Ring plus Shell: 22 cm**4

Required Stress in Ring plus Shell Breq 3.88 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000405

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1736.0^2 * 1418.7195(3.0 + 5.0/1418.7195)0.00004) / 10.9$$

$$= 5 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style CONTINUOUS
 Location of Stiffening Ring EXTERNAL
 Radial Pressure Load Pext*Slen 1.45 Kgf/mm.

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The Radial Shear Load	V	25.12	Kgf
The First Moment of the Area (Ring + Shell)	Q	4274.85	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.49	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	1.53	Kgf/mm.

Stiffening Ring Calcs for : Ring:17 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.686	26.4300	123.851
Total:	6.829		127.066

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	17.1057	627.176
Ring :	8.575	-7.8243	286.878
Total:	8.591		914.053

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell	Breq	2.73 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000284

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.275^2 * 543.0673 (3.0 + 4.686 / 543.0673) 0.0000284) / 10.9$$

$$= 1 \text{ cm}^{*4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.55	Kgf/mm.
The Radial Shear Load	V	7.79	Kgf
The First Moment of the Area (Ring + Shell)	Q	3666.48	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.16	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.58	Kgf/mm.

Stiffening Ring Calcs for : Ring 6 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Total: 8.668 919.992

 Available Moment of Inertia, Ring plus Shell: 18 cm**4
 Required Stress in Ring plus Shell Breq 0.87 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000091

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 51.5(3.0 + 4.7/51.5)0.000009) / 10.9$$

$$= 0 \text{ cm**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style CONTINUOUS
 Location of Stiffening Ring EXTERNAL
 Radial Pressure Load Pext*Slen 0.05 Kgf/mm.
 The Radial Shear Load V 0.74 Kgf
 The First Moment of the Area (Ring + Shell) Q 3679.97 mm.³
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.02 Kgf/mm.
 The Weld Allowable Stress 0.55*S 73.09 N./mm²
 Minimum Weld Leg Size Min(6mm, t, tw) Wldmin 3.00 mm.
 The Weld Allowable Load 2* WLeg*0.55*S 44.72 Kgf/mm.
 The Combined Weld Load SRSS of VQ/I and Pext*Slen 0.05 Kgf/mm.

Stiffening Ring Calcs for : Ring 7 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell Breq 2.88 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000300

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 711.5(3.0 + 4.7/711.5)0.00003) / 10.9$$

$$= 1 \text{ cm**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style CONTINUOUS
 Location of Stiffening Ring EXTERNAL
 Radial Pressure Load Pext*Slen 0.73 Kgf/mm.
 The Radial Shear Load V 10.20 Kgf
 The First Moment of the Area (Ring + Shell) Q 3679.97 mm.³
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.21 Kgf/mm.
 The Weld Allowable Stress 0.55*S 73.09 N./mm²
 Minimum Weld Leg Size Min(6mm, t, tw) Wldmin 3.00 mm.
 The Weld Allowable Load 2* WLeg*0.55*S 44.72 Kgf/mm.
 The Combined Weld Load SRSS of VQ/I and Pext*Slen 0.76 Kgf/mm.

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Stiffening Ring Calcs for : Ring 8 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell Breq 3.16 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000330

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 1405.0(3.0 + 4.7/1405.0)0.000033) / 10.9$$

$$= 3 \text{ cm}^{**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style CONTINUOUS
 Location of Stiffening Ring EXTERNAL
 Radial Pressure Load Pext*Slen 1.43 Kg/mm.
 The Radial Shear Load V 20.14 Kg
 The First Moment of the Area (Ring + Shell) Q 3679.97 mm.³
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.41 Kg/mm.
 The Weld Allowable Stress 0.55*S 73.09 N./mm²
 Minimum Weld Leg Size Min(6mm, t, tw) Wldmin 3.00 mm.
 The Weld Allowable Load 2* Wleg*0.55*S 44.72 Kg/mm.
 The Combined Weld Load SRSS of VQ/I and Pext*Slen 1.49 Kg/mm.

Stiffening Ring Calcs for : Ring 9 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell Breq 3.18 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000330

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 1493.5001(3.0 + 4.7/1493.5001)0.000033) / 10.9$$

$$= 3 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	1.52	Kgf/mm.
The Radial Shear Load	V	21.41	Kgf
The First Moment of the Area (Ring + Shell)	Q	3679.97	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.44	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	1.59	Kgf/mm.

Stiffening Ring Calcs for : Ring 18 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm⁴

Required Stress in Ring plus Shell Breq 3.00 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000312

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 905.0399(3.0 + 4.7/905.0399)0.000031) / 10.9$$

$$= 2 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.92	Kgf/mm.
The Radial Shear Load	V	12.98	Kgf
The First Moment of the Area (Ring + Shell)	Q	3679.97	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.27	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.96	Kgf/mm.

Stiffening Ring Calcs for : Ring 23 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 79 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Shell:	2.381	1.5000	3.572
Ring :	4.700	26.5000	124.550
Total:	7.081		128.122

Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.018	16.5928	655.646
Ring :	8.652	-8.4072	332.200
Total:	8.670		987.847

Available Moment of Inertia, Ring plus Shell: 19 cm**4

Required Stress in Ring plus Shell	Breq	3.70 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000385

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1735.8351^2 * 905.0399(3.0 + 4.7/905.0399)0.000039) / 10.9$$

$$= 3 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.92	Kgf/mm.
The Radial Shear Load	V	16.02	Kgf
The First Moment of the Area (Ring + Shell)	Q	3951.38	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.34	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load SRSS of VQ/I and Pext*Slen		0.98	Kgf/mm.

Stiffening Ring Calcs for : Ring 10 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 79 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	4.700	26.5000	124.550
Total:	7.081		128.122

Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.018	16.5926	655.656
Ring :	8.652	-8.4075	332.221
Total:	8.670		987.878

Available Moment of Inertia, Ring plus Shell: 19 cm**4

Required Stress in Ring plus Shell	Breq	3.06 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000320

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1736.0^2 * 375.0(3.0 + 4.7/375.0)0.000032) / 10.9$$

$$= 1 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
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FileName : Circular Sections

External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.38	Kgf/mm.
The Radial Shear Load	V	6.64	Kgf
The First Moment of the Area (Ring + Shell)	Q	3951.51	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.14	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.41	Kgf/mm.

Stiffening Ring Calcs for : Ring 11 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 79 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	4.700	26.5000	124.550
Total:	7.081		128.122

Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.018	16.5926	655.656
Ring :	8.652	-8.4075	332.221
Total:	8.670		987.878

Available Moment of Inertia, Ring plus Shell: 19 cm**4

Required Stress in Ring plus Shell Breq 0.73 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000076

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq)/10.9$$

$$= (1736.0^2 * 31.5(3.0 + 4.7/31.5)0.000008)/10.9$$

$$= 0 \text{ cm**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.03	Kgf/mm.
The Radial Shear Load	V	0.56	Kgf
The First Moment of the Area (Ring + Shell)	Q	3951.51	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.01	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.03	Kgf/mm.

Stiffening Ring Calcs for : Ring 24 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 79 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	4.684	26.4175	123.727
Total:	7.065		127.299

Centroid of Ring plus Shell: 18 mm.

Inertia	Distance	A*Dist ²
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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Shell:	0.018	16.5182	649.795
Ring :	8.561	-8.3993	330.411
Total:	8.579		980.207

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell	Breq	3.37 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000350

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1736.0^2 * 543.0673(3.0 + 4.6835/543.0673)0.000035) / 10.9$$

$$= 2 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.55	Kgf/mm.
The Radial Shear Load	V	9.61	Kgf
The First Moment of the Area (Ring + Shell)	Q	3933.81	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.21	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.59	Kgf/mm.

Stiffening Ring Calcs for : Ring 26 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.683	26.4150	123.701
Total:	6.826		126.916

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	17.0919	626.168
Ring :	8.558	-7.8231	286.600
Total:	8.574		912.769

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell	Breq	2.73 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000284

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.275^2 * 543.0673(3.0 + 4.683/543.0673)0.000028) / 10.9$$

$$= 1 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.55	Kgf/mm.
The Radial Shear Load	V	7.79	Kgf
The First Moment of the Area (Ring + Shell)	Q	3663.53	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.16	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Minimum Weld Leg Size Min(6mm, t, tw) Wldmin 3.00 mm.
 The Weld Allowable Load 2* WLeg*0.55*S 44.72 Kgf/mm.
 The Combined Weld Load SRSS of VQ/I and Pext*Slen 0.58 Kgf/mm.

Stiffening Ring Calcs for : Ring 12 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell Breq 2.98 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000310

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 872.5 (3.0 + 4.7/872.5) 0.000031) / 10.9$$

$$= 2 \text{ cm}^{**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style CONTINUOUS
 Location of Stiffening Ring EXTERNAL
 Radial Pressure Load Pext*Slen 0.89 Kgf/mm.
 The Radial Shear Load V 12.51 Kgf
 The First Moment of the Area (Ring + Shell) Q 3679.97 mm.³
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.26 Kgf/mm.
 The Weld Allowable Stress 0.55*S 73.09 N./mm²
 Minimum Weld Leg Size Min(6mm, t, tw) Wldmin 3.00 mm.
 The Weld Allowable Load 2* WLeg*0.55*S 44.72 Kgf/mm.
 The Combined Weld Load SRSS of VQ/I and Pext*Slen 0.93 Kgf/mm.

Stiffening Ring Calcs for : Ring 131 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm**4

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Required Stress in Ring plus Shell Breq 3.19 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000333

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 1562.5(3.0 + 4.7/1562.5)0.000033) / 10.9$$

$$= 3 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style CONTINUOUS
 Location of Stiffening Ring EXTERNAL
 Radial Pressure Load Pext*Slen 1.59 Kgf/mm.
 The Radial Shear Load V 22.40 Kgf
 The First Moment of the Area (Ring + Shell) Q 3679.97 mm.³
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.46 Kgf/mm.
 The Weld Allowable Stress 0.55*S 73.09 N./mm²
 Minimum Weld Leg Size Min(6mm, t, tw) Wldmin 3.00 mm.
 The Weld Allowable Load 2* WLeg*0.55*S 44.72 Kgf/mm.
 The Combined Weld Load SRSS of VQ/I and Pext*Slen 1.66 Kgf/mm.

Stiffening Ring Calcs for : Ring 28 , SA-240 304 , Bar Ring: 50 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	5.042	28.2100	142.235
Total:	7.185		145.450

Centroid of Ring plus Shell: 20 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.016	18.7429	752.904
Ring :	10.681	-7.9671	320.040
Total:	10.697		1072.945

Available Moment of Inertia, Ring plus Shell: 21 cm⁴

Required Stress in Ring plus Shell Breq 2.93 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000305

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 840.1315(3.0 + 5.042/840.1315)0.000031) / 10.9$$

$$= 2 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style CONTINUOUS
 Location of Stiffening Ring EXTERNAL
 Radial Pressure Load Pext*Slen 0.86 Kgf/mm.
 The Radial Shear Load V 12.05 Kgf
 The First Moment of the Area (Ring + Shell) Q 4017.02 mm.³
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.23 Kgf/mm.
 The Weld Allowable Stress 0.55*S 73.09 N./mm²
 Minimum Weld Leg Size Min(6mm, t, tw) Wldmin 3.00 mm.
 The Weld Allowable Load 2* WLeg*0.55*S 44.72 Kgf/mm.
 The Combined Weld Load SRSS of VQ/I and Pext*Slen 0.89 Kgf/mm.

Stiffening Ring Calcs for : Ring 22 , SA-240 304 , Bar Ring: 50 x 10 mm.

Effective Length of Shell: 59 mm.

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	1.767	1.5000	2.651
Ring :	5.042	28.2113	142.248
Total:	6.810		144.899
Centroid of Ring plus Shell:			21 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.013	19.7783	691.404
Ring :	10.683	-6.9330	242.361
Total:	10.696		933.765
Available Moment of Inertia, Ring plus Shell:			20 cm**4

Required Stress in Ring plus Shell	Breq	1.99 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000208

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (956.225^2 * 840.1315(3.0 + 5.0423/840.1315)0.000021) / 10.9$$

$$= 1 \text{ cm**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.86	Kgf/mm.
The Radial Shear Load	V	8.19	Kgf
The First Moment of the Area (Ring + Shell)	Q	3495.77	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.14	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.87	Kgf/mm.

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Element and Detail Weights: Step: 15 4:14pm Feb 20,2026

Element and Detail Weights:

From	To	Element Metal Wgt. kg.	Element ID Volume Cm.	Corroded Metal Wgt. kg.	Corroded ID Volume Cm.	Extra due Misc % kg.
10	20	793.769	10689519	793.769	10689519	...
20	30	231.016	3630184	231.016	3630184	...
30	40	104.889	1880831	104.889	1880831	...
40	50	143.693	2317308	143.693	2317308	...
50	60	477.657	6928454	477.657	6928454	...
60	70	238.06	3862180	238.06	3862180	...
70	80	104.889	1880831	104.889	1880831	...
80	90	143.693	2317308	143.693	2317308	...
90	100	371.509	5388796	371.509	5388796	...
100	110	179.371	2195581	179.371	2195581	...
Total		2788	41090996.00	2788	41090996.00	0

Weight of Details:

From	Type	Weight of Detail kg.	X Offset, Dtl. Cent. mm.	Y Offset, Dtl. Cent. mm.	Description
10	Ring	16.0367	...	3958	Ring:[1 of 1]
10	Ring	16.0367	...	5946	Ring:[2 of 2]
10	Ring	16.0367	...	7939	Ring:[3 of 3]
10	Ring	16.0367	...	7952	Ring:[4 of 4]
20	Ring	34.3201	Ring 16
30	Ring	21.1334	Ring5
40	Ring	17.1723	...	1199	Ring:17
40	Ring	22.5201	Ring 21
50	Ring	17.222	...	90	Ring 6
50	Ring	17.222	...	103	Ring 7
50	Ring	17.222	...	1513	Ring 8
50	Ring	17.222	...	2913	Ring 9
60	Ring	17.222	Ring 18
60	Ring	21.1315	...	1999	Ring 23
70	Ring	21.1334	...	737	Ring 10
70	Ring	21.1334	...	750	Ring 11
80	Ring	21.0573	Ring 24
80	Ring	17.161	...	1199	Ring 26
90	Ring	17.222	...	375	Ring 12
90	Ring	17.222	...	1745	Ring 13]
100	Ring	12.8003	...	1999	Ring 22
100	Ring	18.5186	Ring 28

Total Weight of Each Detail Type:

Stiffeners	412.8

Sum of the Detail Weights	412.8 kg.

Weight Summation Results: (kg.)

	Fabricated	Shop Test	Shipping	Erected	Empty	Operating
Main Elements	2788.5	2788.5	2788.5	2788.5	2788.5	2788.5
Stif. Rings	412.8	412.8	412.8	412.8	412.8	412.8
Test Liquid	...	41065.9

Totals	3201.3	44267.2	3201.3	3201.3	3201.3	3201.3

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Element and Detail Weights: Step: 15 4:14pm Feb 20,2026

Weight Summary:

Fabricated Wt.	- Bare Weight without Removable Internals	3201.3 kg.
Shop Test Wt.	- Fabricated Weight + Water (Full)	44267.2 kg.
Shipping Wt.	- Fab. Weight + removable Intls.+ Shipping App.	3201.3 kg.
Erected Wt.	- Fab. Wt + or - loose items (trays,platforms etc.)	3201.3 kg.
Ope. Wt. no Liq	- Fab. Weight + Internals. + Details + Weights	3201.3 kg.
Operating Wt.	- Empty Weight + Operating Liq. Uncorroded	3201.3 kg.
Field Test Wt.	- Empty Weight + Water (Full)	44267.2 kg.
Mass of the Vessel	1/3 of the Vertical Vessel	1107.8 kg.

Outside Surface Areas of Elements:

From	To	Surface Area cm ²
10	20	330367
20	30	96120.1
30	40	43630.4
40	50	59784.7
50	60	198769
60	70	99045.5
70	80	43630.4
80	90	59784.7
90	100	154598
100	110	74674.9
Total		1160404.250 cm ²

Element and Detail Weights:

From	To	Total Ele. Empty Wgt. kg.	Total. Ele. Oper. Wgt. kg.	Total. Ele. Hydro. Wgt. kg.	Total Dtl. Offset Mom. Kg-m.	Oper. Wgt. No Liquid kg.
10	20	857.916	857.916	857.916	...	857.916
20	30	265.336	265.336	265.336	...	265.336
30	40	126.022	126.022	126.022	...	126.022
40	50	183.386	183.386	183.386	...	183.386
50	60	546.545	546.545	546.545	...	546.545
60	70	276.413	276.413	276.413	...	276.413
70	80	147.156	147.156	147.156	...	147.156
80	90	181.912	181.912	181.912	...	181.912
90	100	405.953	405.953	405.953	...	405.953
100	110	210.689	210.689	210.689	...	210.689

Cumulative Vessel Weight

From	To	Cumulative Ope Wgt. No Liquid kg.	Cumulative Oper. Wgt. kg.	Cumulative Hydro. Wgt. kg.
10	20	3201.33	3201.33	3201.33
20	30	2343.41	2343.41	2343.41
30	40	2078.08	2078.08	2078.08
40	50	1952.05	1952.05	1952.05
50	60	1768.67	1768.67	1768.67
60	70	1222.12	1222.12	1222.12
70	80	945.71	945.71	945.71
80	90	798.554	798.554	798.554
90	100	616.643	616.643	616.643
100	110	210.689	210.689	210.689

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Element and Detail Weights: Step: 15 4:14pm Feb 20,2026

Note: The cumulative operating weights no liquid in the column above
are the cumulative operating weights minus the operating liquid
weight minus any weights absent in the empty condition.

Cumulative Vessel Moment

From	To	Cumulative Empty Mom. Kg-m.	Cumulative Oper. Mom. Kg-m.	Cumulative Hydro. Mom. Kg-m.
10	20
20	30
30	40
40	50
50	60
60	70
70	80
80	90
90	100
100	110

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Conical Section: Cone: 16 4:14pm Feb 20, 2026

Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1**Conical Section From 20 To 30 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	12.7573 mm.
Maximum Centroid Reinforcement Distance Small End	11.0651 mm.

Reinforcement Calculations for Cone / Large Cylinder:**Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	6.136 degrees

Required Area of Reinforcement for Large End Under External Pressure

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	6.136 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Large End All. Stress} * \text{Large End Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm ²

Required Area of Reinforcement, Large End, External [Arl]:

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\delta / \alpha))$$

$$= (1.0068 * 0.4426 * 868.0 * 0.107 / (134 * 1.0)) * (1 - 1/4 * ((0.1 * 868.0 - 0.443) / 0.443) * (0.187 / 6.136))$$

$$= 0.0302 \text{ cm}^2$$

FileName : Circular Sections

Conical Section:

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Force per Length, Cone Large End External Pressure [QL]:

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(868.0/2) + 0.0/(\pi(1736.0 - 3.0)) + \\
 &\quad 0.0/(\pi(868.0 - 3.0/2)(868.0 - 3.0/2)) \\
 &= 0.443 \text{ Kg/mm.}
 \end{aligned}$$

Available Area of Reinforcement, Large End, External [Ael]:

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1736.0 * 3.0)^{1/2} * (3.0 + 3.0/0.994) \\
 &= 2.3884 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement Area, Large End, External Pressure:

Area of reinforcement required per App. 1-8(1)	0.0302	cm ²
Area of reinforcement in shell per App. 1-8(2)	2.3884	cm ²
Area of reinforcement in stiffening ring	4.7000	cm ²

Intermediate Results, Large End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	46.87	cm ²
Force per Unit Length on Shell / Cone Junction	1.55	Kgf/mm.
Actual Buckling Stress associated with this Force	4.23	N./mm ²
Material Strain associated with this stress	0.000044	

Required Moment of Inertia, Large End, External Pressure [I's]:

$$\begin{aligned}
 &= A * D_l^2 * A_{t1} / 10.9 \\
 &= 0.438E-04 * 1736.0 * 1736.0 * 46.87/10.9 \\
 &= 56786.08 \text{ mm}^4
 \end{aligned}$$

Available Moment of Inertia, Large End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.191	0.0000	0.000	16.2161	0.009	313.119
Con	1.198	-2.1334	-2.555	18.3495	0.027	403.238
Sec	4.700	25.0000	117.500	-8.7840	8.652	362.643
TOT	7.088		114.945		8.688	1079.001
Centroid of Section			16.2161	Moment of Inertia		19.48

Summary of Large End Inertia Calculations

Available Moment of Inertia (Large End)	19.478	cm**4
Required Moment of Inertia (Large End)	5.679	cm**4

Reinforcement Calculations for Cone / Small Cylinder:**Required Area of Reinforcement for Small End under Internal Pressure**

Small end ratio of pressure to allowable stress	0.00048	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	6.136	degrees

Intermediate Value [k]:

$$\begin{aligned}
 &= \max(Y / (S_r * E_{rs}), 1) \\
 &= \max(0.19226E+11 / (100.3 * 0.18458E+09), 1) \\
 &= 1.0386
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)} \\
 &= 104.2 * 0.18458E+09 \\
 &= 19225886720.0 \text{ N./mm}^2
 \end{aligned}$$

Decay Length, Cone Small End:

$$\begin{aligned}
 &= 1.4 * \sqrt{R_s(t_s - c_a)} \\
 &= 1.4 * \sqrt{650.0(3.0 - 0.0)} \\
 &= 61.822 \text{ mm.}
 \end{aligned}$$

Required Area of Reinforcement, Small End, Internal [Ars]:

$$\begin{aligned}
 &= k * Q_s * R_s / (S_s * E_1) * (1 - \delta / \alpha) * \tan(\alpha) \\
 &= 1.04 * 1.6571 * 650.0 / (104 * 1.0) *
 \end{aligned}$$

FileName : Circular Sections

Conical Section:

Cone: 16 4:14pm Feb 20,2026

$$(1 - 4.0/6.14) \cdot 0.1075$$

$$= 0.0394 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= P(Rs/2) - Faxial/(\pi(Ds + Ts)) + Moment/(\pi(Rs + Ts/2)(Rs + Ts/2))$$

$$= 0.5(650.0/2) - 0.0/(\pi(1300.0 + 3.0)) +$$

$$0.0/(\pi(650.0 + 3.0/2)(650.0 + 3.0/2))$$

$$= 1.657 \text{ Kgf/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.78(Rs \cdot Ts)^{1/2} \cdot ((Ts - t) + (Tc - Tr) / \cos(\alpha))$$

$$= 0.78(650.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 0.312) + (3.0 - 0.314) / 0.99)$$

$$= 1.8563 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, Internal Pressure:

Area of reinforcement required per App. 1-5(3)	0.0394	cm ²
Area of reinforcement in shell per App. 1-5(4)	1.8563	cm ²
Area of reinforcement in stiffening ring	9.7000	cm ²

Required Area of Reinforcement for Small End Under External Pressure

Allowable Stress of Small End Material (Ext. Temp)	133.8	N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	133.8	N./mm ²

Intermediate Value [k]:

$$= \max(Y / (Srs \cdot Ers), 1)$$

$$= \max(0.25813E+11 / (132.9 \cdot 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} \cdot \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 \cdot 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Area of Reinforcement Required in Small End Shell [Ars]:

$$= k \cdot QS \cdot Rs \cdot \tan(\alpha) / (Ss \cdot E1)$$

$$= (1.0068 \cdot 0.333 \cdot 653.0 \cdot 0.1075) / (134 \cdot 1.0)$$

$$= 0.0172 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= Pext(Rs/2) + Faxial/(\pi(Ds - Ts)) + Moment/(\pi(Rs - Ts/2)(Rs - Ts/2))$$

$$= 0.1(653.0/2) + 0.0/(\pi(1306.0 - 3.0)) +$$

$$0.0/(\pi(653.0 - 3.0/2)(653.0 - 3.0/2))$$

$$= 0.333 \text{ Kgf/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.55 \cdot (Ds \cdot ts)^{1/2} \cdot ((ts - t) + (tc - tr) / \cos(\alpha))$$

$$= 0.55 \cdot (1306.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 0.573) + (3.0 - 1.893) / 0.994)$$

$$= 1.2188 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, External Pressure:

Area of reinforcement required per App. 1-8(3)	0.0172	cm ²
Area of reinforcement in shell per App. 1-8(4)	1.2188	cm ²
Area of reinforcement in stiffening ring	9.7000	cm ²

Intermediate Results, Small End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	41.37	cm ²
Force per Unit Length on Shell / Cone Junction	0.88	Kgf/mm.
Actual Buckling Stress associated with this Force	2.04	N./mm ²
Material Strain associated with this stress	0.000021	

Required Moment of Inertia, Small End, External Pressure [I's]:

$$= A \cdot Ds^2 \cdot Ats / 10.9$$

$$= 0.211E-04 \cdot 1306.0 \cdot 1306.0 \cdot 41.37 / 10.9$$

$$= 13689.88 \text{ mm}^4$$

FileName : Circular Sections

Conical Section:

Cone: 16 4:14pm Feb 20,2026

Available Moment of Inertia, Small End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.033	0.0000	0.000	41.3643	0.008	1767.127
Con	1.039	1.8504	1.922	39.5139	0.020	1621.850
Sec	9.700	50.0000	485.000	-8.6357	76.056	723.377
TOT	11.772		486.922		76.083	4112.354
Centroid of Section			41.3643	Moment of Inertia		117.21

Summary of Small End Inertia Calculations

Available Moment of Inertia (Small End)	0.1172E+03	cm**4
Required Moment of Inertia (Small End)	0.1369E+01	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

Results for Discontinuity Stresses per Bednar p. 236 2nd Edition

Stress Type	Stress	Allowable	Location
Tensile Stress	15.45	312.47	Small Cyl. Long.
Compres. Stress	-4.60	-312.47	Small Cyl. Long.
Membrane Stress	16.38	-156.24	Small End Tang.
Tensile Stress	15.49	312.47	Cone Longitudinal
Compres. Stress	-4.56	-312.47	Cone Longitudinal
Compres Stress	16.44	-156.24	Cone Tangential
Tensile Stress	22.60	312.47	Large Cyl. Long.
Compres. Stress	-8.16	-312.47	Large Cyl. Long.
Membrane Stress	5.98	-156.24	Large End Tang.
Tensile Stress	22.64	312.47	Cone Longitudinal
Compres. Stress	-8.11	-312.47	Cone Longitudinal
Compres Stress	6.06	-156.24	Cone Tangential

Note: An asterisk (*) denotes that this stress was not applicable for this combination of loads.

Maximum Allowable Pressure Calculations for Cone to Shell Junction:

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	3.584	Thickness due to internal pressure, Cone Large End
MAPnc	4.745	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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FileName : Circular Sections -----

Conical Section: Cone: 17 4:14pm Feb 20, 2026

Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1**Conical Section From 40 To 50 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	12.7573 mm.
Maximum Centroid Reinforcement Distance Small End	11.4810 mm.

Reinforcement Calculations for Cone / Large Cylinder:**Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	7.829 degrees

Required Area of Reinforcement for Large End Under External Pressure

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	7.829 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm ²

Required Area of Reinforcement, Large End, External [Arl]:

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\delta / \alpha))$$

$$= (1.0068 * 0.4426 * 868.0 * 0.138 / (134 * 1.0)) * (1 - 1/4 * ((0.1 * 868.0 - 0.443) / 0.443) * (0.187 / 7.829))$$

$$= 0.0387 \text{ cm}^2$$

FileName : Circular Sections

Conical Section:

Cone: 17 4:14pm Feb 20,2026

Force per Length, Cone Large End External Pressure [QL]:

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(868.0/2) + 0.0/(\pi(1736.0 - 3.0)) + \\
 &\quad 0.0/(\pi(868.0 - 3.0/2)(868.0 - 3.0/2)) \\
 &= 0.443 \text{ Kg/mm.}
 \end{aligned}$$

Available Area of Reinforcement, Large End, External [Ael]:

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1736.0 * 3.0)^{1/2} * (3.0 + 3.0/0.991) \\
 &= 2.3927 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement Area, Large End, External Pressure:

Area of reinforcement required per App. 1-8(1)	0.0387	cm ²
Area of reinforcement in shell per App. 1-8(2)	2.3927	cm ²
Area of reinforcement in stiffening ring	5.0000	cm ²

Intermediate Results, Large End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	35.17	cm ²
Force per Unit Length on Shell / Cone Junction	1.09	Kgf/mm.
Actual Buckling Stress associated with this Force	3.94	N./mm ²
Material Strain associated with this stress	0.000041	

Required Moment of Inertia, Large End, External Pressure [I's]:

$$\begin{aligned}
 &= A * D_l^2 * A_{t1} / 10.9 \\
 &= 0.408E-04 * 1736.0 * 1736.0 * 35.17/10.9 \\
 &= 39720.57 \text{ mm}^4
 \end{aligned}$$

Available Moment of Inertia, Large End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.191	0.0000	0.000	17.4794	0.009	363.808
Con	1.202	-2.7288	-3.280	20.2082	0.039	490.842
Sec	5.000	26.5000	132.500	-9.0206	10.417	406.853
TOT	7.393		129.220		10.465	1261.504
Centroid of Section			17.4794	Moment of Inertia		23.08

Summary of Large End Inertia Calculations

Available Moment of Inertia (Large End)	23.080	cm**4
Required Moment of Inertia (Large End)	3.972	cm**4

Reinforcement Calculations for Cone / Small Cylinder:**Required Area of Reinforcement for Small End under Internal Pressure**

Small end ratio of pressure to allowable stress	0.00048	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	7.829	degrees

Intermediate Value [k]:

$$\begin{aligned}
 &= \max(Y / (S_r * E_{rs}), 1) \\
 &= \max(0.19226E+11 / (100.3 * 0.18458E+09), 1) \\
 &= 1.0386
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)} \\
 &= 104.2 * 0.18458E+09 \\
 &= 19225886720.0 \text{ N./mm}^2
 \end{aligned}$$

Decay Length, Cone Small End:

$$\begin{aligned}
 &= 1.4 * \sqrt{R_s(t_s - c_a)} \\
 &= 1.4 * \sqrt{700.0(3.0 - 0.0)} \\
 &= 64.156 \text{ mm.}
 \end{aligned}$$

Required Area of Reinforcement, Small End, Internal [Ars]:

$$\begin{aligned}
 &= k * Q_s * R_s / (S_s * E_1) * (1 - \delta / \alpha) * \tan(\alpha) \\
 &= 1.04 * 1.7846 * 700.0 / (104 * 1.0) *
 \end{aligned}$$

FileName : Circular Sections

Conical Section:

Cone: 17 4:14pm Feb 20,2026

$$(1 - 4.0/7.83) \cdot 0.1375$$

$$= 0.0822 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= P(Rs/2) - Faxial/(\pi(Ds + Ts)) + Moment/(\pi(Rs + Ts/2)(Rs + Ts/2))$$

$$= 0.5(700.0/2) - 0.0/(\pi(1400.0 + 3.0)) +$$

$$0.0/(\pi(700.0 + 3.0/2)(700.0 + 3.0/2))$$

$$= 1.785 \text{ Kgf/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.78(Rs \cdot Ts)^{1/2} \cdot ((Ts - t) + (Tc - Tr) / \cos(\alpha))$$

$$= 0.78(700.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 0.336) + (3.0 - 0.339) / 0.99)$$

$$= 1.9122 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, Internal Pressure:

Area of reinforcement required per App. 1-5(3)	0.0822	cm ²
Area of reinforcement in shell per App. 1-5(4)	1.9122	cm ²
Area of reinforcement in stiffening ring	4.6860	cm ²

Required Area of Reinforcement for Small End Under External Pressure

Allowable Stress of Small End Material (Ext. Temp)	133.8	N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	133.8	N./mm ²

Intermediate Value [k]:

$$= \max(Y / (Srs \cdot Ers), 1)$$

$$= \max(0.25813E+11 / (132.9 \cdot 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} \cdot \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 \cdot 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Area of Reinforcement Required in Small End Shell [Ars]:

$$= k \cdot QS \cdot Rs \cdot \tan(\alpha) / (Ss \cdot E1)$$

$$= (1.0068 \cdot 0.3584 \cdot 703.0 \cdot 0.1375) / (134 \cdot 1.0)$$

$$= 0.0256 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= Pext(Rs/2) + Faxial/(\pi(Ds - Ts)) + Moment/(\pi(Rs - Ts/2)(Rs - Ts/2))$$

$$= 0.1(703.0/2) + 0.0/(\pi(1406.0 - 3.0)) +$$

$$0.0/(\pi(703.0 - 3.0/2)(703.0 - 3.0/2))$$

$$= 0.358 \text{ Kgf/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.55 \cdot (Ds \cdot ts)^{1/2} \cdot ((ts - t) + (tc - tr) / \cos(\alpha))$$

$$= 0.55 \cdot (1406.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 0.57) + (3.0 - 0.514) / 0.991)$$

$$= 1.7647 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, External Pressure:

Area of reinforcement required per App. 1-8(3)	0.0256	cm ²
Area of reinforcement in shell per App. 1-8(4)	1.7647	cm ²
Area of reinforcement in stiffening ring	4.6860	cm ²

Intermediate Results, Small End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	24.21	cm ²
Force per Unit Length on Shell / Cone Junction	0.55	Kgf/mm.
Actual Buckling Stress associated with this Force	2.35	N./mm ²
Material Strain associated with this stress	0.000024	

Required Moment of Inertia, Small End, External Pressure [I's]:

$$= A \cdot Ds^2 \cdot Ats / 10.9$$

$$= 0.244E-04 \cdot 1406.0 \cdot 1406.0 \cdot 24.21 / 10.9$$

$$= 10712.14 \text{ mm}^4$$

FileName : Circular Sections

Conical Section:

Cone: 17 4:14pm Feb 20,2026

Available Moment of Inertia, Small End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.072	0.0000	0.000	17.4694	0.008	327.033
Con	1.082	2.4558	2.656	15.0136	0.030	243.822
Sec	4.686	24.9300	116.822	-7.4606	8.575	260.827
TOT	6.839		119.478		8.613	831.682
Centroid of Section			17.4694	Moment of Inertia		16.93

Summary of Small End Inertia Calculations

Available Moment of Inertia (Small End)	0.1693E+02	cm**4
Required Moment of Inertia (Small End)	0.1071E+01	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

Results for Discontinuity Stresses per Bednar p. 236 2nd Edition

Stress Type	Stress	Allowable	Location
Tensile Stress	20.16	312.47	Small Cyl. Long.
Compres. Stress	-8.47	-312.47	Small Cyl. Long.
Membrane Stress	19.57	-156.24	Small End Tang.
Tensile Stress	20.22	312.47	Cone Longitudinal
Compres. Stress	-8.41	-312.47	Cone Longitudinal
Compres Stress	19.68	-156.24	Cone Tangential
Tensile Stress	26.87	312.47	Large Cyl. Long.
Compres. Stress	-12.43	-312.47	Large Cyl. Long.
Membrane Stress	3.63	-156.24	Large End Tang.
Tensile Stress	26.94	312.47	Cone Longitudinal
Compres. Stress	-12.36	-312.47	Cone Longitudinal
Compres Stress	3.76	-156.24	Cone Tangential

Note: An asterisk (*) denotes that this stress was not applicable for this combination of loads.

Maximum Allowable Pressure Calculations for Cone to Shell Junction:

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	3.571	Thickness due to internal pressure, Cone Large End
MAPnc	4.728	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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FileName : Circular Sections -----

Conical Section: Cone: 18 4:14pm Feb 20, 2026

Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1**Conical Section From 60 To 70 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	12.7573 mm.
Maximum Centroid Reinforcement Distance Small End	11.4810 mm.

Reinforcement Calculations for Cone / Large Cylinder:**Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	4.716 degrees

Required Area of Reinforcement for Large End Under External Pressure

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	4.716 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm ²

Required Area of Reinforcement, Large End, External [Arl]:

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\delta / \alpha))$$

$$= (1.0068 * 0.4426 * 868.0 * 0.082 / (134 * 1.0)) * (1 - 1/4 * ((0.1 * 868.0 - 0.443) / 0.443) * (0.187 / 4.716))$$

$$= 0.0232 \text{ cm}^2$$

FileName : Circular Sections

Conical Section:

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Force per Length, Cone Large End External Pressure [QL]:

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(868.0/2) + 0.0/(\pi(1736.0 - 3.0)) + \\
 &\quad 0.0/(\pi(868.0 - 3.0/2)(868.0 - 3.0/2)) \\
 &= 0.443 \text{ Kg/mm.}
 \end{aligned}$$

Available Area of Reinforcement, Large End, External [Ael]:

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1736.0 * 3.0)^{1/2} * (3.0 + 3.0/0.997) \\
 &= 2.3855 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement Area, Large End, External Pressure:

Area of reinforcement required per App. 1-8(1)	0.0232	cm ²
Area of reinforcement in shell per App. 1-8(2)	2.3855	cm ²
Area of reinforcement in stiffening ring	4.7000	cm ²

Intermediate Results, Large End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	45.86	cm ²
Force per Unit Length on Shell / Cone Junction	1.57	Kg/mm.
Actual Buckling Stress associated with this Force	4.37	N./mm ²
Material Strain associated with this stress	0.000045	

Required Moment of Inertia, Large End, External Pressure [I's]:

$$\begin{aligned}
 &= A * D_l^2 * A_{t1} / 10.9 \\
 &= 0.453E-04 * 1736.0 * 1736.0 * 45.86/10.9 \\
 &= 57448.63 \text{ mm}^4
 \end{aligned}$$

Available Moment of Inertia, Large End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.191	0.0000	0.000	16.3070	0.009	316.641
Con	1.195	-1.6373	-1.956	17.9443	0.020	384.720
Sec	4.700	25.0000	117.500	-8.6930	8.652	355.172
TOT	7.086		115.544		8.681	1056.534
Centroid of Section			16.3070	Moment of Inertia		19.25

Summary of Large End Inertia Calculations

Available Moment of Inertia (Large End)	19.246	cm**4
Required Moment of Inertia (Large End)	5.745	cm**4

Reinforcement Calculations for Cone / Small Cylinder:**Required Area of Reinforcement for Small End under Internal Pressure**

Small end ratio of pressure to allowable stress	0.00048	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	4.716	degrees

Intermediate Value [k]:

$$\begin{aligned}
 &= \max(Y / (S_r * E_{rs}), 1) \\
 &= \max(0.19226E+11 / (100.3 * 0.18458E+09), 1) \\
 &= 1.0386
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)} \\
 &= 104.2 * 0.18458E+09 \\
 &= 19225886720.0 \text{ N./mm}^2
 \end{aligned}$$

Decay Length, Cone Small End:

$$\begin{aligned}
 &= 1.4 * \sqrt{R_s(t_s - c_a)} \\
 &= 1.4 * \sqrt{700.0(3.0 - 0.0)} \\
 &= 64.156 \text{ mm.}
 \end{aligned}$$

Required Area of Reinforcement, Small End, Internal [Ars]:

$$\begin{aligned}
 &= k * Q_s * R_s / (S_s * E_1) * (1 - \delta / \alpha) * \tan(\alpha) \\
 &= 1.04 * 1.7846 * 700.0 / (104 * 1.0) *
 \end{aligned}$$

FileName : Circular Sections

Conical Section:

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$$(1 - 4.0/4.72) \cdot 0.0825$$

$$= 0.0153 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= P(Rs/2) - Faxial/(\pi(Ds + Ts)) + Moment/(\pi(Rs + Ts/2)(Rs + Ts/2))$$

$$= 0.5(700.0/2) - 0.0/(\pi(1400.0 + 3.0)) +$$

$$0.0/(\pi(700.0 + 3.0/2)(700.0 + 3.0/2))$$

$$= 1.785 \text{ Kgf/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.78(Rs \cdot Ts)^{1/2} \cdot ((Ts - t) + (Tc - Tr) / \cos(\alpha))$$

$$= 0.78(700.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 0.336) + (3.0 - 0.337) / 1.)$$

$$= 1.9072 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, Internal Pressure:

Area of reinforcement required per App. 1-5(3)	0.0153	cm ²
Area of reinforcement in shell per App. 1-5(4)	1.9072	cm ²
Area of reinforcement in stiffening ring	4.7000	cm ²

Required Area of Reinforcement for Small End Under External Pressure

Allowable Stress of Small End Material (Ext. Temp)	133.8	N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	133.8	N./mm ²

Intermediate Value [k]:

$$= \max(Y / (Srs \cdot Ers), 1)$$

$$= \max(0.25813E+11 / (132.9 \cdot 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} \cdot \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 \cdot 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Area of Reinforcement Required in Small End Shell [Ars]:

$$= k \cdot QS \cdot Rs \cdot \tan(\alpha) / (Ss \cdot E1)$$

$$= (1.0068 \cdot 0.3584 \cdot 703.0 \cdot 0.0825) / (134 \cdot 1.0)$$

$$= 0.0153 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= Pext(Rs/2) + Faxial/(\pi(Ds - Ts)) + Moment/(\pi(Rs - Ts/2)(Rs - Ts/2))$$

$$= 0.1(703.0/2) + 0.0/(\pi(1406.0 - 3.0)) +$$

$$0.0/(\pi(703.0 - 3.0/2)(703.0 - 3.0/2))$$

$$= 0.358 \text{ Kgf/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.55 \cdot (Ds \cdot ts)^{1/2} \cdot ((ts - t) + (tc - tr) / \cos(\alpha))$$

$$= 0.55 \cdot (1406.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 1.89) + (3.0 - 2.0) / 0.997)$$

$$= 0.7552 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, External Pressure:

Area of reinforcement required per App. 1-8(3)	0.0153	cm ²
Area of reinforcement in shell per App. 1-8(4)	0.7552	cm ²
Area of reinforcement in stiffening ring	4.7000	cm ²

Intermediate Results, Small End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	58.61	cm ²
Force per Unit Length on Shell / Cone Junction	1.60	Kgf/mm.
Actual Buckling Stress associated with this Force	2.82	N./mm ²
Material Strain associated with this stress	0.000029	

Required Moment of Inertia, Small End, External Pressure [I's]:

$$= A \cdot Ds^2 \cdot Ats / 10.9$$

$$= 0.292E-04 \cdot 1406.0 \cdot 1406.0 \cdot 58.61 / 10.9$$

$$= 31077.31 \text{ mm}^4$$

FileName : Circular Sections

Conical Section:

Cone: 18 4:14pm Feb 20,2026

Available Moment of Inertia, Small End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.072	0.0000	0.000	17.3925	0.008	324.163
Con	1.075	1.4735	1.584	15.9191	0.016	272.487
Sec	4.700	25.0000	117.500	-7.6075	8.652	272.005
TOT	6.847		119.084		8.676	868.655
Centroid of Section			17.3925	Moment of Inertia		17.36

Summary of Small End Inertia Calculations

Available Moment of Inertia (Small End)	0.1736E+02	cm**4
Required Moment of Inertia (Small End)	0.3108E+01	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

Results for Discontinuity Stresses per Bednar p. 236 2nd Edition

Stress Type	Stress	Allowable	Location
Tensile Stress	14.45	312.47	Small Cyl. Long.
Compres. Stress	-2.76	-312.47	Small Cyl. Long.
Membrane Stress	16.43	-156.24	Small End Tang.
Tensile Stress	14.47	312.47	Cone Longitudinal
Compres. Stress	-2.74	-312.47	Cone Longitudinal
Compres Stress	16.47	-156.24	Cone Tangential
Tensile Stress	19.03	312.47	Large Cyl. Long.
Compres. Stress	-4.59	-312.47	Large Cyl. Long.
Membrane Stress	7.94	-156.24	Large End Tang.
Tensile Stress	19.05	312.47	Cone Longitudinal
Compres. Stress	-4.56	-312.47	Cone Longitudinal
Compres Stress	7.99	-156.24	Cone Tangential

Note: An asterisk (*) denotes that this stress was not applicable for this combination of loads.

Maximum Allowable Pressure Calculations for Cone to Shell Junction:

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	3.592	Thickness due to internal pressure, Cone Large End
MAPnc	4.756	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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FileName : Circular Sections -----

Conical Section: Cone: 19 4:14pm Feb 20, 2026

Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1**Conical Section From 80 To 90 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	12.7573 mm.
Maximum Centroid Reinforcement Distance Small End	11.4810 mm.

Reinforcement Calculations for Cone / Large Cylinder:**Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	7.829 degrees

Required Area of Reinforcement for Large End Under External Pressure

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	7.829 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm ²

Required Area of Reinforcement, Large End, External [Arl]:

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\delta / \alpha))$$

$$= (1.0068 * 0.4426 * 868.0 * 0.138 / (134 * 1.0)) * (1 - 1/4 * ((0.1 * 868.0 - 0.443) / 0.443) * (0.187 / 7.829))$$

$$= 0.0387 \text{ cm}^2$$

FileName : Circular Sections

Conical Section:

Cone: 19 4:14pm Feb 20,2026

Force per Length, Cone Large End External Pressure [QL]:

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(868.0/2) + 0.0/(\pi(1736.0 - 3.0)) + \\
 &\quad 0.0/(\pi(868.0 - 3.0/2)(868.0 - 3.0/2)) \\
 &= 0.443 \text{ Kg/mm.}
 \end{aligned}$$

Available Area of Reinforcement, Large End, External [Ael]:

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1736.0 * 3.0)^{1/2} * (3.0 + 3.0/0.991) \\
 &= 2.3927 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement Area, Large End, External Pressure:

Area of reinforcement required per App. 1-8(1)	0.0387	cm ²
Area of reinforcement in shell per App. 1-8(2)	2.3927	cm ²
Area of reinforcement in stiffening ring	4.6835	cm ²

Intermediate Results, Large End, External Pressure:

Calculate Cylinder Decay Length:

$$\begin{aligned}
 &= 2 * \sqrt{R_l * t_s} \\
 &= 2 * \sqrt{868.0 * 3.0} \\
 &= 102.0588 \text{ mm.}
 \end{aligned}$$

Note: LL is zero since large cylinder length is less than the decay length

Area Available in Cone, Shell, and Reinforcement	22.85	cm ²
Force per Unit Length on Shell / Cone Junction	0.68	Kgf/mm.
Actual Buckling Stress associated with this Force	3.78	N./mm ²
Material Strain associated with this stress	0.000039	

Required Moment of Inertia, Large End, External Pressure [I's]:

$$\begin{aligned}
 &= A * D_l^2 * A_{tl} / 10.9 \\
 &= 0.392E-04 * 1736.0 * 1736.0 * 22.85/10.9 \\
 &= 24792.07 \text{ mm.}^4
 \end{aligned}$$

Available Moment of Inertia, Large End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.191	0.0000	0.000	16.0286	0.009	305.921
Con	1.202	-2.7288	-3.280	18.7574	0.039	422.893
Sec	4.684	24.9175	116.701	-8.8889	8.561	370.059
TOT	7.076		113.421		8.609	1098.873
Centroid of Section			16.0286	Moment of Inertia		19.60

Summary of Large End Inertia Calculations

Available Moment of Inertia (Large End)	19.598	cm**4
Required Moment of Inertia (Large End)	2.479	cm**4

Reinforcement Calculations for Cone / Small Cylinder:**Required Area of Reinforcement for Small End under Internal Pressure**

Small end ratio of pressure to allowable stress	0.00048	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	7.829	degrees

Intermediate Value [k]:

$$\begin{aligned}
 &= \max(Y / (S_r * E_{rs}), 1) \\
 &= \max(0.19226E+11 / (100.3 * 0.18458E+09), 1) \\
 &= 1.0386
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)} \\
 &= 104.2 * 0.18458E+09 \\
 &= 19225886720.0 \text{ N./mm}^2
 \end{aligned}$$

FileName : Circular Sections

Conical Section:

Cone: 19 4:14pm Feb 20,2026

Decay Length, Cone Small End:

$$\begin{aligned}
 &= 1.4 * \sqrt{Rs (ts - ca)} \\
 &= 1.4 * \sqrt{700.0 (3.0 - 0.0)} \\
 &= 64.156 \text{ mm.}
 \end{aligned}$$

Required Area of Reinforcement, Small End, Internal [Ars]:

$$\begin{aligned}
 &= k * QS * Rs / (Ss * E1) * (1 - \delta / \alpha) * \tan(\alpha) \\
 &= 1.04 * 1.7846 * 700.0 / (104 * 1.0) * \\
 &\quad (1 - 4.0 / 7.83) * 0.1375 \\
 &= 0.0822 \text{ cm}^2
 \end{aligned}$$

Force per Length, Cone Small End [QS]:

$$\begin{aligned}
 &= P(Rs/2) - Faxial / (\pi(Ds + Ts)) + Moment / (\pi(Rs + Ts/2)(Rs + Ts/2)) \\
 &= 0.5(700.0/2) - 0.0 / (\pi(1400.0 + 3.0)) + \\
 &\quad 0.0 / (\pi(700.0 + 3.0/2)(700.0 + 3.0/2)) \\
 &= 1.785 \text{ Kgf/mm.}
 \end{aligned}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$\begin{aligned}
 &= 0.78(Rs * Ts)^{1/2} * ((Ts - t) + (Tc - Tr) / \cos(\alpha)) \\
 &= 0.78(700.0 * 3.0)^{1/2} * ((3.0 - 0.336) + (3.0 - 0.339) / 0.99) \\
 &= 1.9122 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement Area, Small End, Internal Pressure:

Area of reinforcement required per App. 1-5(3)	0.0822	cm ²
Area of reinforcement in shell per App. 1-5(4)	1.9122	cm ²
Area of reinforcement in stiffening ring	4.6830	cm ²

Required Area of Reinforcement for Small End Under External Pressure

Allowable Stress of Small End Material (Ext. Temp)	133.8	N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	133.8	N./mm ²

Intermediate Value [k]:

$$\begin{aligned}
 &= \max(Y / (Srs * Ers), 1) \\
 &= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1) \\
 &= 1.0068
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)} \\
 &= 133.8 * 0.19291E+09 \\
 &= 25812725760.0 \text{ N./mm}^2
 \end{aligned}$$

Area of Reinforcement Required in Small End Shell [Ars]:

$$\begin{aligned}
 &= k * QS * Rs * \tan(\alpha) / (Ss * E1) \\
 &= (1.0068 * 0.3584 * 703.0 * 0.1375) / (134 * 1.0) \\
 &= 0.0256 \text{ cm}^2
 \end{aligned}$$

Force per Length, Cone Small End [QS]:

$$\begin{aligned}
 &= Pext(Rs/2) + Faxial / (\pi(Ds - Ts)) + Moment / (\pi(Rs - Ts/2)(Rs - Ts/2)) \\
 &= 0.1(703.0/2) + 0.0 / (\pi(1406.0 - 3.0)) + \\
 &\quad 0.0 / (\pi(703.0 - 3.0/2)(703.0 - 3.0/2)) \\
 &= 0.358 \text{ Kgf/mm.}
 \end{aligned}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$\begin{aligned}
 &= 0.55 * (Ds * ts)^{1/2} * [(ts - t) + (tc - tr) / \cos(\alpha)] \\
 &= 0.55 * (1406.0 * 3.0)^{1/2} * [(3.0 - 1.043) + (3.0 - 0.514) / 0.991] \\
 &= 1.5954 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement Area, Small End, External Pressure:

Area of reinforcement required per App. 1-8(3)	0.0256	cm ²
Area of reinforcement in shell per App. 1-8(4)	1.5954	cm ²
Area of reinforcement in stiffening ring	4.6830	cm ²

Intermediate Results, Small End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	28.48	cm ²
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FileName : Circular Sections

Conical Section: Cone: 19 4:14pm Feb 20,2026

Force per Unit Length on Shell / Cone Junction 0.70 Kgf/mm.
 Actual Buckling Stress associated with this Force 2.53 N./mm²
 Material Strain associated with this stress 0.000026

Required Moment of Inertia, Small End, External Pressure [I's]:

$$= A * D_s^2 * A_{ts} / 10.9$$

$$= 0.262E-04 * 1406.0 * 1406.0 * 28.48/10.9$$

$$= 13537.53 \text{ mm}^4$$

Available Moment of Inertia, Small End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.072	0.0000	0.000	17.4558	0.008	326.525
Con	1.082	2.4558	2.656	15.0000	0.030	243.382
Sec	4.683	24.9150	116.677	-7.4592	8.558	260.558
TOT	6.836		119.333		8.596	830.465
Centroid of Section			17.4558	Moment of Inertia		16.90

Summary of Small End Inertia Calculations

Available Moment of Inertia (Small End)	0.1690E+02	cm**4
Required Moment of Inertia (Small End)	0.1354E+01	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

Results for Discontinuity Stresses per Bednar p. 236 2nd Edition

Stress Type	Stress	Allowable	Location
Tensile Stress	20.16	312.47	Small Cyl. Long.
Compres. Stress	-8.47	-312.47	Small Cyl. Long.
Membrane Stress	19.57	-156.24	Small End Tang.
Tensile Stress	20.22	312.47	Cone Longitudinal
Compres. Stress	-8.41	-312.47	Cone Longitudinal
Compres Stress	19.68	-156.24	Cone Tangential
Tensile Stress	26.87	312.47	Large Cyl. Long.
Compres. Stress	-12.43	-312.47	Large Cyl. Long.
Membrane Stress	3.63	-156.24	Large End Tang.
Tensile Stress	26.94	312.47	Cone Longitudinal
Compres. Stress	-12.36	-312.47	Cone Longitudinal
Compres Stress	3.76	-156.24	Cone Tangential

Note: An asterisk (*) denotes that this stress was not applicable for this combination of loads.

Maximum Allowable Pressure Calculations for Cone to Shell Junction:

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	3.571	Thickness due to internal pressure, Cone Large End
MAPnc	4.728	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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FileName : Circular Sections -----

Conical Section: Cone: 20 4:14pm Feb 20, 2026

Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1**Conical Section From 100 To 110 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	11.4810 mm.
Maximum Centroid Reinforcement Distance Small End	9.4670 mm.

Reinforcement Calculations for Cone / Large Cylinder:**Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	6.419 degrees

Required Area of Reinforcement for Large End Under External Pressure

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	6.419 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm ²

Required Area of Reinforcement, Large End, External [Arl]:

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\delta / \alpha))$$

$$= (1.0068 * 0.3584 * 703.0 * 0.112 / (134 * 1.0)) * (1 - 1/4 * ((0.1 * 703.0 - 0.358) / 0.358) * (0.187 / 6.419))$$

$$= 0.0208 \text{ cm}^2$$

FileName : Circular Sections

Conical Section:

Cone: 20 4:14pm Feb 20,2026

Force per Length, Cone Large End External Pressure [QL]:

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(703.0/2) + 0.0/(\pi(1406.0 - 3.0)) + \\
 &\quad 0.0/(\pi(703.0 - 3.0/2)(703.0 - 3.0/2)) \\
 &= 0.358 \text{ Kg/mm.}
 \end{aligned}$$

Available Area of Reinforcement, Large End, External [Ael]:

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1406.0 * 3.0)^{1/2} * (3.0 + 3.0/0.994) \\
 &= 2.1500 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement Area, Large End, External Pressure:

Area of reinforcement required per App. 1-8(1)	0.0208	cm ²
Area of reinforcement in shell per App. 1-8(2)	2.1500	cm ²
Area of reinforcement in stiffening ring	5.0420	cm ²

Intermediate Results, Large End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	61.56	cm ²
Force per Unit Length on Shell / Cone Junction	2.00	Kgf/mm.
Actual Buckling Stress associated with this Force	3.35	N./mm ²
Material Strain associated with this stress	0.000035	

Required Moment of Inertia, Large End, External Pressure [I's]:

$$\begin{aligned}
 &= A * D_l^2 * A_{tl} / 10.9 \\
 &= 0.348E-04 * 1406.0 * 1406.0 * 61.56/10.9 \\
 &= 38820.20 \text{ mm}^4
 \end{aligned}$$

Available Moment of Inertia, Large End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.072	0.0000	0.000	18.4240	0.008	363.751
Con	1.078	-2.0093	-2.167	20.4333	0.023	450.240
Sec	5.042	26.7100	134.672	-8.2860	10.681	346.172
TOT	7.192		132.505		10.712	1160.163
Centroid of Section			18.4240	Moment of Inertia		22.31

Summary of Large End Inertia Calculations

Available Moment of Inertia (Large End)	22.314	cm ⁴
Required Moment of Inertia (Large End)	3.882	cm ⁴

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

Results for Discontinuity Stresses per Bednar p. 236 2nd Edition

Stress Type	Stress	Allowable	Location
Tensile Stress	17.57	312.47	Large Cyl. Long.
Compres. Stress	-5.87	-312.47	Large Cyl. Long.
Membrane Stress	5.24	-156.24	Large End Tang.
Tensile Stress	17.60	312.47	Cone Longitudinal
Compres. Stress	-5.84	-312.47	Cone Longitudinal
Compres Stress	5.31	-156.24	Cone Tangential

Note: An asterisk (*) denotes that this stress was not applicable for this combination of loads.

Maximum Allowable Pressure Calculations for Cone to Shell Junction:

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	4.424	Thickness due to internal pressure, Cone Large End

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FileName : Circular Sections -----

Conical Section: Cone: 20 4:14pm Feb 20,2026

MAPnc 5.857 Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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FileName : Circular Sections

Center of Gravity Calculation: Step: 21 4:14pm Feb 20,2026

Shop/Field Installation Options :

Note : The CG is computed from the first Element From Node

Center of Gravity of Stiffening Rings	14631.545 mm.
Center of Gravity of Bare Shell New and Cold	13151.735 mm.
Center of Gravity of Bare Shell Corroded	13151.735 mm.
Vessel CG in the Operating Condition	13342.544 mm.
Vessel CG in the Fabricated (Shop/Empty) Condition	13342.544 mm.

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FileName : Circular Sections

MDMT Summary:

Step: 22 4:14pm Feb 20,2026

Minimum Design Metal Temperature Results Summary :

Description	Curve	Basic MDMT °C	Reduced MDMT °C	UG-20 (f) MDMT °C	Thickness ratio	Gov Thk mm.	E*	PWHT reqd
Notes								
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
Warmest MDMT:		-196						
Required Minimum Design Metal Temperature					-28.9	°C		
Warmest Computed Minimum Design Metal Temperature					-196.0	°C		

Notes:

- [!] - This was an impact tested material.
- [1] - Governing Nozzle Weld.
- [4] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-c).
- [5] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-b).
- [6] - MDMT Calculations at the Shell/Head Joint.
- [7] - MDMT Calculations for the Straight Flange.
- [8] - Cylinder/Cone/Flange Junction MDMT.
- [9] - Calculations in the Spherical Portion of the Head.
- [10] - Calculations in the Knuckle Portion of the Head.
- [11] - Calculated (Body Flange) Flange MDMT.
- [12] - Calculated Flat Head MDMT per UCS-66.3
- [13] - Tubesheet MDMT, shell side, if applicable
- [14] - Tubesheet MDMT, tube side, if applicable
- [15] - Nozzle Material
- [16] - Shell or Head Material
- [17] - Impact Testing required
- [18] - Impact Testing not required, see UCS-66(b)(3)
- [19] - Select a valid hydrotest type to get the UG-20(f) exemption
- [20] - Cylinder/Cone Junction MDMT based on Longitudinal Stress considerations
- [21] - Bolting Material

UG-84(b)(2) was not considered.

UCS-66(g) was not considered.

UCS-66(i) was not considered.

Notes:

- Impact test temps were not entered in and not considered in the analysis.
- UCS-66(i) applies to impact tested materials not by specification and
- UCS-66(g) applies to materials impact tested per UG-84.1 General Note (c).
- The Basic MDMT includes the (30F) PWHT credit if applicable.

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FileName : Circular Sections

Vessel Design Summary:

Step: 23 4:14pm Feb 20,2026

ASME Code, Section VIII Division 1, 2017

Diameter Spec : 1300.000 x 1730.000 x 1400.000 x 1730.000 x 1400.000 mm. ID
 Vessel Design Length, Tangent to Tangent 26052.00 mm.

Distance of Bottom Tangent above Grade 8052.00 mm.
 Distance of Base above Grade 0.00 mm.
 Specified Datum Line Distance 0.00 mm.

Internal Design Temperature 170 °C
 Internal Design Pressure 0.500 bars

External Design Temperature 50 °C
 External Design Pressure 0.100 bars

Maximum Allowable Working Pressure 3.571 bars
 External Max. Allowable Working Pressure 0.137 bars
 Hydrostatic Test Pressure 0.000 bars

Required Minimum Design Metal Temperature -28.9 °C
 Warmest Computed Minimum Design Metal Temperature -196.0 °C

Wind Design Code ASCE-93
 Earthquake Design Code UBC-94

Materials of Construction:

Component Type	Material	Class	Thickness	UNS #	Normal ized	Impact Tested
Shell	SA-240 316	S31600	No	No
Cone	SA-240 316	S31600	No	No
Rings	SA-240 316	S31600	No	No
Rings	SA-240 304	S30400	No	No

Normalized is determined based on the UCS-66 material curve selection and Figure UCS-66.

Impact Tested is based on material selection and material data properties.

Element Pressures and MAWP (bars & mm.):

Element Description or Type	Design Pressure + Stat. head	Ext. Press.	Element M.A.W.P	Corrosion Allowance	Str. Flg. Gov.	In Creep Range
Cylinder	0.500	0.10	4.794	0.0000	N/A	No
Conical	0.500	0.10	3.584	0.0000	N/A	No
Cylinder	0.500	0.10	3.605	0.0000	N/A	No
Conical	0.500	0.10	3.571	0.0000	N/A	No
Cylinder	0.500	0.10	4.452	0.0000	N/A	No
Conical	0.500	0.10	3.593	0.0000	N/A	No
Cylinder	0.500	0.10	3.605	0.0000	N/A	No
Conical	0.500	0.10	3.571	0.0000	N/A	No
Cylinder	0.500	0.10	4.452	0.0000	N/A	No
Conical	0.500	0.10	4.424	0.0000	N/A	No

Stiffener Ring Specifications:

Elevation mm.	Selected Type	User Description
3958.00	Bar 47.0 x 10.	Ring:[1 of 1]
5946.00	Bar 47.0 x 10.	Ring:[2 of 2]
7939.00	Bar 47.0 x 10.	Ring:[3 of 3]
7952.00	Bar 47.0 x 10.	Ring:[4 of 4]
8052.00	Bar 97.0 x 10.	Ring 16
10052.00	Bar 47.0 x 10.	Ring5
12051.00	Bar 46.9 x 10.	Ring:17

FileName : Circular Sections

Vessel Design Summary:

Step: 23 4:14pm Feb 20,2026

10852.00	Bar	50.0 x 10.	Ring 21
12142.00	Bar	47.0 x 10.	Ring 6
12155.00	Bar	47.0 x 10.	Ring 7
13565.00	Bar	47.0 x 10.	Ring 8
14965.00	Bar	47.0 x 10.	Ring 9
16552.00	Bar	47.0 x 10.	Ring 18
18551.00	Bar	47.0 x 10.	Ring 23
19289.00	Bar	47.0 x 10.	Ring 10
19302.00	Bar	47.0 x 10.	Ring 11
19352.00	Bar	46.8 x 10.	Ring 24
20551.00	Bar	46.8 x 10.	Ring 26
20927.00	Bar	47.0 x 10.	Ring 12
22297.00	Bar	47.0 x 10.	Ring 13]
26051.00	Bar	50.4 x 10.	Ring 22
24052.00	Bar	50.4 x 10.	Ring 28

Element Types and Properties:

Element Type	"To" Elev mm.	Element Length mm.	Nominal Thickness mm.	Finished Thickness mm.	Reqd Thk Internal mm.	Reqd Thk External mm.	Long Eff	Circ Eff
Cylinder	8052.0	8052.0	3.0	3.0	1.5	...	1.00	1.00
Conical	10052.0	2000.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	10852.0	800.0	3.0	3.0	1.5	...	1.00	1.00
Conical	12052.0	1200.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	16552.0	4500.0	3.0	3.0	1.5	...	1.00	1.00
Conical	18552.0	2000.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	19352.0	800.0	3.0	3.0	1.5	...	1.00	1.00
Conical	20552.0	1200.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	24052.0	3500.0	3.0	3.0	1.5	...	1.00	1.00
Conical	26052.0	2000.0	3.0	3.0	1.5	...	1.00	1.00

External Pressure Calculations:

From	To	External Actual T. mm.	External Required T. mm.	External Design Pressure bars	External M.A.W.P. bars
10	Ring	3	2.64879	0.1	0.13736
Ring	Ring	3	1.98494	0.1	0.28018
Ring	Ring	3	1.98708	0.1	0.27946
Ring	Ring	3	0.47257	0.1	2.80031
Ring	20	3	0.57319	0.1	2.56958
20	Ring	3	1.86049	0.1	0.32853
Ring	30	3	2.24961	0.1	0.20495
30	Ring	3	No Calc	0.1	No Calc
Ring	40	3	1.62103	0.1	0.46793
40	Ring	3	2.25778	0.1	0.20299
Ring	Ring	3	1.85593	0.1	0.33169
Ring	50	3	0.51366	0.1	2.52828
50	Ring	3	0.56956	0.1	2.4333
Ring	Ring	3	0.50876	0.1	2.55869
Ring	Ring	3	1.80715	0.1	0.35777
Ring	Ring	3	1.80239	0.1	0.36038
Ring	60	3	1.89031	0.1	0.31681
60	Ring	3	0.51049	0.1	2.54788
Ring	Ring	3	2.27117	0.1	0.19962
Ring	70	3	0.6303	0.1	1.95727
70	Ring	3	1.56614	0.1	0.51
Ring	Ring	3	0.62816	0.1	1.96585
Ring	80	3	0.62816	0.1	1.96585
80	Ring	3	0.63409	0.1	1.9422
Ring	Ring	3	1.85593	0.1	0.33169
Ring	90	3	0.51366	0.1	2.52828
90	Ring	3	1.04348	0.1	1.44208

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FileName : Circular Sections

Vessel Design Summary:

Step: 23 4:14pm Feb 20,2026

Ring	Ring	3	1.78809	0.1	0.3684
Ring	100	3	1.96728	0.1	0.28568
100	Ring	3	0.51198	0.1	2.53866
Ring	Ring	3	1.94504	0.1	0.2941
Ring	110	3	0.34821	0.1	4.03247

External Pressure Calculations:

From	To	Actual Length Bet. Stiffeners mm.	Allowable Length Bet. Stiffeners mm.	Ring Inertia Required cm**4	Ring Inertia Available cm**4
10	Ring	3958	5366.66	No Calc	No Calc
Ring	Ring	1988	5367.77	4.74696	17.6358
Ring	Ring	1993	5367.78	3.17821	17.6358
Ring	Ring	12.9998	362.283	1.60148	17.6358
Ring	20	100	3105.85	0.090213	17.6358
20	Ring	1679.94	1679.94	No Calc	No Calc
Ring	30	1752.3	1752.3	2.74012	118.645
30	Ring	No Calc	No Calc	No Calc	No Calc
Ring	40	800	3576.41	1.50003	18.5486
40	Ring	1752.3	1752.3	No Calc	No Calc
Ring	Ring	1085.13	1085.13	5.32031	21.7628
Ring	50	0.99994	0.99994	1.08257	17.7314
50	Ring	90	2782.45	No Calc	No Calc
Ring	Ring	13	332.63	0.1026	17.8679
Ring	Ring	1410	4840.94	1.41706	17.8679
Ring	Ring	1400	4840.92	2.80393	17.8679
Ring	60	1587	4841.33	2.96283	17.8679
60	Ring	0.99994	0.99994	No Calc	No Calc
Ring	Ring	1809.08	1809.08	1.80308	17.8679
Ring	70	0.99989	0.99989	3.39301	18.5482
70	Ring	737	3576.31	No Calc	No Calc
Ring	Ring	13	979.513	1.41118	18.5486
Ring	80	50	1799.98	0.11813	18.5486
80	Ring	0.99989	0.99989	No Calc	No Calc
Ring	Ring	1085.13	1085.13	2.0298	18.381
Ring	90	0.99994	0.99994	1.08257	17.7021
90	Ring	375	4840.98	No Calc	No Calc
Ring	Ring	1370	4840.85	1.73585	17.8679
Ring	100	1755	4841.66	3.11292	17.8679
100	Ring	0.99994	0.99994	No Calc	No Calc
Ring	Ring	1679.26	1679.26	1.67377	21.4269
Ring	110	0.99982	0.99982	0.52653	20.0339



Factored Loads:

Un-Factored Loads:



Weights:

Fabricated - Bare W/O Removable Internals	3201.3	kg.
Shop Test - Fabricated + Water (Full)	44267.2	kg.
Shipping - Fab. + Rem. Intls.+ Shipping App.	3201.3	kg.
Erected - Fab. + Rem. Intls.+ Insul. (etc)	3201.3	kg.
Empty - Fab. + Intls. + Details + Wghts.	3201.3	kg.
Operating - Empty + Operating Liquid (No CA)	3201.3	kg.
Field Test - Empty Weight + Water (Full)	44267.2	kg.

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 Innovation and Equipment	MODIFIED STARCH FLASH DRYER PROJECT					 A.G.F.D. Tândărei	
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.		Rev.
	5204	ME	CL	000	02		00

APPENDIX 2 **CALCULATION** **CYCLONES- ZONE 4**

	MODIFIED STARCH FLASH DRYER PROJECT						
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.	Rev.	
	5204	ME	CL	000	02	00	

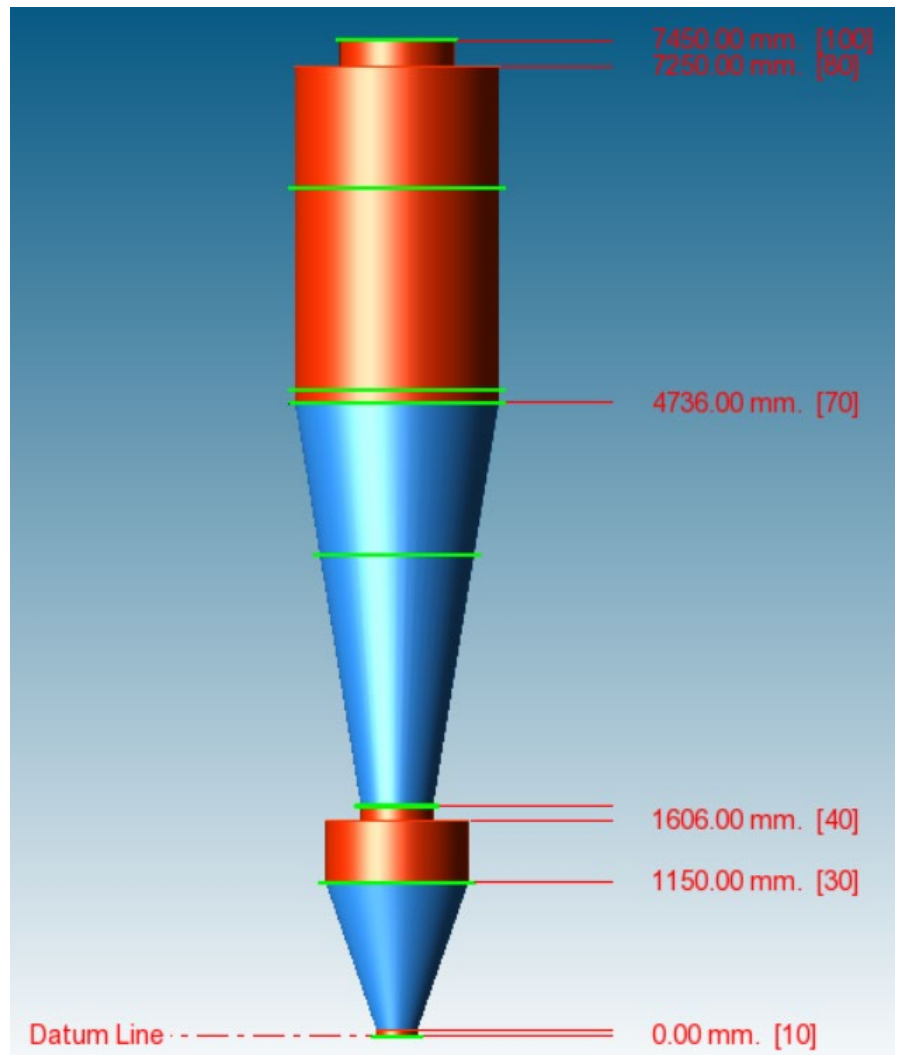
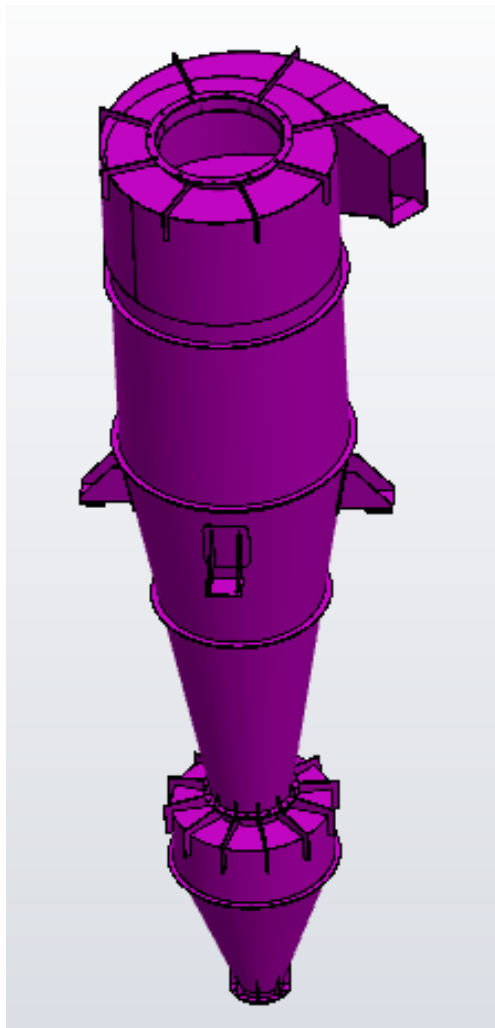


Figure B- Cyclones - Zone 4

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DESIGN CALCULATION

In Accordance with ASME Section VIII Division 1

ASME Code Version : 2017

Analysis Performed by : SPLM Licensed User

Job File : D:\2-PAYAM\1-PROJECTS\MICROTEC PROJECT\13-MICROT

Date of Analysis : Feb 20,2026 4:19pm

PV Elite 2019 SP1, March 2019

Note:

PV Elite performs all calculations internally in Imperial Units to remain compliant with the ASME Code and any built in assumptions in the ASME Code formulas. The finalized results are reflected to show the user's set of selected units.

PV Elite 2019 SP1 Licensee: SPLM Licensed User
FileName : CYCLONE-Rev 01 -----
Warnings and Errors: Step: 0 4:19pm Feb 20,2026

Class From To : Basic Element Checks.
=====

Class From To: Check of Additional Element Data
=====

Please insure the C factor entered is in accordance with Figure UG-34.
Please insure the C factor entered is in accordance with Figure UG-34.

There were no geometry errors or warnings.

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FileName : CYCLONE-Rev 01

Input Echo:

Step: 1 4:19pm Feb 20,2026

PV Elite Vessel Analysis Program: Input Data

Design Internal Pressure (for Hydrotest)	0.5	bars
Design Internal Temperature	170.0	°C
Type of Hydrotest	not Specified	
Hydrotest Position	Vertical	
Projection of Nozzle from Vessel Top	0	mm.
Projection of Nozzle from Vessel Bottom	0	mm.
Minimum Design Metal Temperature	-28.9	°C
Type of Construction	Welded	
Special Service	None	
Degree of Radiography	RT-1	
Use Higher Longitudinal Stresses (Flag)	Y	
Select t for Internal Pressure (Flag)	N	
Select t for External Pressure (Flag)	N	
Select t for Axial Stress (Flag)	N	
Select Location for Stiff. Rings (Flag)	N	
Consider Vortex Shedding	N	
Perform a Corroded Hydrotest	N	

Load Case 1	NP+EW+WI+FW+BW
Load Case 2	NP+EW+EE+FS+BS
Load Case 3	NP+OW+WI+FW+BW
Load Case 4	NP+OW+EQ+FS+BS
Load Case 5	NP+HW+HI
Load Case 6	NP+HW+HE
Load Case 7	IP+OW+WI+FW+BW
Load Case 8	IP+OW+EQ+FS+BS
Load Case 9	EP+OW+WI+FW+BW
Load Case 10	EP+OW+EQ+FS+BS
Load Case 11	HP+HW+HI
Load Case 12	HP+HW+HE
Load Case 13	IP+WE+EW
Load Case 14	IP+WF+CW
Load Case 15	IP+VO+OW
Load Case 16	IP+VE+EW
Load Case 17	NP+VO+OW
Load Case 18	FS+BS+IP+OW
Load Case 19	FS+BS+EP+OW

Wind Design Code	ASCE-7 93
Basic Wind Speed	[V] 112.65 Km/hr
Surface Roughness Category	C: Open Terrain
Importance Factor	1.0
Type of Surface	Moderately Smooth
Base Elevation	0 mm.
Percent Wind for Hydrotest	33.0
Using User defined Wind Press. Vs Elev.	N
Damping Factor (Beta) for Wind (Ope)	0.0100
Damping Factor (Beta) for Wind (Empty)	0.0000
Damping Factor (Beta) for Wind (Filled)	0.0000

Seismic Design Code	UBC 94
UBC Seismic Zone (1=1,2=2a,3=2b,4=3,5=4)	0.000
UBC Importance Factor	1.000
UBC Soil Type	S1
UBC Horizontal Force Factor	3.000
UBC Percent Seismic for Hydrotest	0.000

Design Pressure + Static Head	Y
Consider MAP New and Cold in Noz. Design	N
Consider External Loads for Nozzle Des.	Y
Use ASME VIII-1 Appendix 1-9	N

FileName : CYCLONE-Rev 01

Input Echo:

Step: 1 4:19pm Feb 20,2026

Material Database Year

Current w/Addenda or Code Year

Configuration Directives:

Do not use Nozzle MDMT Interpretation VIII-1 01-37	No
Use Table G instead of exact equation for "A"	Yes
Shell Head Joints are Tapered	Yes
Compute "K" in corroded condition	Yes
Use Code Case 2286	No
Use the MAWP to compute the MDMT	Yes
For thickness ratios <= 0.35, MDMT will be -155F (-104C)	Yes
For PWHT & P1 Materials the MDMT can be < -55F (-48C)	No
Using Metric Material Databases, ASME II D	No
Calculate B31.3 type stress for Nozzles with Loads	Yes
Reduce the MDMT due to lower membrane stress	Yes
Consider Longitudinal Stress in MDMT calcs. (Div. 1)	No

Complete Listing of Vessel Elements and Details:

Element From Node	10	
Element To Node	20	
Element Type	Cylinder	
Description		
Distance "FROM" to "TO"	50	mm.
Inside Diameter	321	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Allowable Stress, Ambient	137.9	N./mm ²
Allowable Stress, Operating	136.1	N./mm ²
Allowable Stress, Hydrotest	179.3	N./mm ²
Material Density	0.008027	kg./cm ³
P Number Thickness	0	mm.
Yield Stress, Operating	156.1	N./mm ²
External Pressure Chart Name	HA-2	
UNS Number	S31600	
Product Form	Plate	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Weld is pre-Heated	No	
Element From Node	10	
Detail Type	Ring	
Detail ID	Ring 1-1	
Dist. from "FROM" Node / Offset dist	0	mm.
Inside Diameter of Ring	327	mm.
Thickness of Ring	12	mm.
Outside Diameter of Ring	405	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	20
Element To Node	30
Element Type	Conical
Description	

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Distance "FROM" to "TO"	1100	mm.
Inside Diameter	321	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Cone Diameter at "To" End	1124	mm.
Design Length of Cone	1100	mm.
Half Apex Angle of Cone	20.052082	degrees
Toriconical (Y/N)	N	
Weld is pre-Heated	No	

Element From Node	20	
Detail Type	Ring	
Detail ID	Ring 2-1	
Dist. from "FROM" Node / Offset dist	1095	mm.
Inside Diameter of Ring	1126.3	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1227.3	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	30	
Element To Node	40	
Element Type	Cylinder	
Description		
Distance "FROM" to "TO"	456	mm.
Inside Diameter	1124	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Weld is pre-Heated	No	

Element From Node	40	
Element To Node	50	
Element Type	Flat	
Description		
Distance "FROM" to "TO"	10	mm.
Inside Diameter	1124	mm.
Element Thickness	10	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	10	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars

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Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240	316
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Flat Head Attachment Factor	0.2	
Small diameter if Non-Circular	0	mm.
Weld is pre-Heated	No	

Element From Node	50	
Element To Node	60	
Element Type	Cylinder	
Description		
Distance "FROM" to "TO"	115	mm.
Inside Diameter	566	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240	316
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Weld is pre-Heated	No	

Element From Node	50	
Detail Type	Ring	
Detail ID	Ring 5-1	
Dist. from "FROM" Node / Offset dist	108	mm.
Inside Diameter of Ring	572	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	670	mm.
Material Name	SA-240	304
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	60	
Element To Node	70	
Element Type	Conical	
Description		
Distance "FROM" to "TO"	3015	mm.
Inside Diameter	566	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240	316
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Cone Diameter at "To" End	1606	mm.
Design Length of Cone	3015	mm.

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Half Apex Angle of Cone	9.7855883	degrees
Toriconical (Y/N)	N	
Weld is pre-Heated	No	

Element From Node	60	
Detail Type	Ring	
Detail ID	Ring 6-1	
Dist. from "FROM" Node / Offset dist	5	mm.
Inside Diameter of Ring	571.73	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	669.73	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	60	
Detail Type	Ring	
Detail ID	Ring 6-2	
Dist. from "FROM" Node / Offset dist	1874	mm.
Inside Diameter of Ring	1218.4	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1314	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	60	
Detail Type	Ring	
Detail ID	Ring 6-3	
Dist. from "FROM" Node / Offset dist	3010	mm.
Inside Diameter of Ring	1610.3	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1705.9	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	70	
Element To Node	80	
Element Type	Cylinder	
Description		
Distance "FROM" to "TO"	2514	mm.
Inside Diameter	1606	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Weld is pre-Heated	No	

Element From Node	70	
Detail Type	Ring	
Detail ID	Ring 7-1	
Dist. from "FROM" Node / Offset dist	100	mm.
Inside Diameter of Ring	1612	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1706	mm.

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Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	
Element From Node	70	
Detail Type	Ring	
Detail ID	Ring 7-2	
Dist. from "FROM" Node / Offset dist	1610	mm.
Inside Diameter of Ring	1612	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1706	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	80	
Element To Node	90	
Element Type	Flat	
Description		
Distance "FROM" to "TO"	14	mm.
Inside Diameter	1606	mm.
Element Thickness	14	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	14	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Flat Head Attachment Factor	0.2	
Small diameter if Non-Circular	0	mm.
Weld is pre-Heated	No	

Element From Node	90	
Element To Node	100	
Element Type	Cylinder	
Description		
Distance "FROM" to "TO"	200	mm.
Inside Diameter	898	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Weld is pre-Heated	No	

Element From Node	90	
Detail Type	Ring	
Detail ID	Ring 8.1	
Dist. from "FROM" Node / Offset dist	195	mm.
Inside Diameter of Ring	904	mm.

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Thickness of Ring	14	mm.
Outside Diameter of Ring	956	mm.
Material Name	SA-240	316
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

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XY Coordinate Calculations: Step: 2 4:19pm Feb 20,2026

XY Coordinate Calculations:

From	To	X (Horiz.) mm.	Y (Vert.) mm.	DX (Horiz.) mm.	DY (Vert.) mm.
10	20	...	50	...	50
20	30	...	1150	...	1100
30	40	...	1606	...	456
40	50	...	1616	...	10
50	60	...	1731	...	115
60	70	...	4746	...	3015
70	80	...	7260	...	2514
80	90	...	7274	...	14
90	100	...	7474	...	200

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Internal Pressure Calculations: Step: 7 4:19pm Feb 20,2026

Element Thickness, Pressure, Diameter and Allowable Stress :

From	To	Int. Press + Liq. Hd bars	Nominal Thickness mm.	Total Corr Allowance mm.	Element Diameter mm.	Allowable Stress (SE) N./mm ²
10	20	0.5	3	...	321	136.1
20	30	0.5	3	...	1124	136.1
30	40	0.5	3	...	1124	136.07
40	50	0.5	10	...	1124	136.07
50	60	0.5	3	...	566	136.07
60	70	0.5	3	...	1606	136.07
70	80	0.5	3	...	1606	136.07
80	90	0.5	14	...	1606	136.07
90	100	0.5	3	...	898	136.07

Element Required Thickness and MAWP :

From	To	Design Pressure bars	M.A.W.P. Corroded bars	M.A.P. New & Cold bars	Minimum Thickness mm.	Required Thickness mm.
10	20	0.5	25.1557	25.4884	3	1.5
20	30	0.5	6.80386	6.89384	3	1.5
30	40	0.5	7.23977	7.33728	3	1.5
40	50	0.5	2.12356	2.15217	10	4.85235
50	60	0.5	14.3321	14.5251	3	1.5
60	70	0.5	4.99817	5.06549	3	1.5
70	80	0.5	5.07179	5.14011	3	1.5
80	90	0.5	1.65349	1.67576	14	7.69861
90	100	0.5	9.05452	9.17649	3	1.5
Minimum			1.653	1.676		

MAWP: 1.653 bars, limited by: Flat Head.

Internal Pressure Calculation Results :**ASME Code, Section VIII Division 1, 2017****Cylindrical Shell From 10 To 20 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 \cdot 160.5) / (136.1 \cdot 1.0 - 0.6 \cdot 0.5) \\
 &= 0.0590 + 0.0000 = 0.0590 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (136.1 \cdot 1.0 \cdot 3.0) / (160.5 + 0.6 \cdot 3.0) \\
 &= 25.156 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (160.5 + 0.6 \cdot 3.0) \\
 &= 25.488 \text{ bars}
 \end{aligned}$$

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Internal Pressure Calculations: Step: 7 4:19pm Feb 20, 2026

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P * (R + 0.6 * t)) / (E * t) \\
 &= (0.5 * (160.5 + 0.6 * 3.0)) / (1.0 * 3.0) \\
 &= 2.705 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 * t_{nom} / R_f) * (1 - R_f / R_o)$ 0.926 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Conical Section From 20 To 30 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P * D) / (2 * \cos(a) * (S * E - 0.6 * P)) \text{ per Appendix 1-4 (e)} \\
 &= (0.5 * 1124.0) / (2 * 0.9394 * (136.1 * 1.0 - 0.6 * 0.5)) \\
 &= 0.2199 + 0.0000 = 0.2199 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (2 * S * E * t * \cos(a)) / (D + 1.2 * t * \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 * 136.1 * 1.0 * 3.0 * 0.939) / (1124.0 + 1.2 * 3.0 * 0.939) \\
 &= 6.804 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (2 * S * E * t * \cos(a)) / (D + 1.2 * t * \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 * 137.9 * 1.0 * 3.0 * 0.9394) / (1124.0 + 1.2 * 3.0 * 0.9394) \\
 &= 6.894 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P * (D + 1.2 * t * \cos(a))) / (2 * E * t * \cos(a)) \\
 &= (0.5 * (1124.0 + 1.2 * 3.0 * 0.9394)) / (2 * 1.0 * 3.0 * 0.9394) \\
 &= 10.002 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 * t_{nom} / R_f) * (1 - R_f / R_o)$ 0.925 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

Note: The Pressure at the Large Diameter is used in the TR calculation.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Cylindrical Shell From 30 To 40 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P * R) / (S * E - 0.6 * P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 * 562.0) / (136.07 * 1.0 - 0.6 * 0.5) \\
 &= 0.2066 + 0.0000 = 0.2066 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S * E * t) / (R + 0.6 * t) \text{ per UG-27 (c) (1)} \\
 &= (136.07 * 1.0 * 3.0) / (562.0 + 0.6 * 3.0) \\
 &= 7.240 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

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Internal Pressure Calculations: Step: 7 4:19pm Feb 20, 2026

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (562.0 + 0.6 \cdot 3.0) \\
 &= 7.337 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t) \\
 &= (0.5 \cdot (562.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0) \\
 &= 9.397 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.266 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Welded Flat Head From 40 To 50 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= d \cdot \sqrt{Z \cdot C \cdot P / (S \cdot E)} \text{ per UG-34 (c) (3)} \\
 &= 566.0 \cdot \sqrt{1.0 \cdot 0.2 \cdot 0.5 / (136.07 \cdot 1.0)} \\
 &= 4.8524 + 0.0000 = 4.8524 \text{ mm.}
 \end{aligned}$$

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (t/d)^2 \cdot ((S \cdot E) / (C \cdot Z)) \text{ UG-34 (c) (3)} \\
 &= (10.0/566.0)^2 \cdot ((136.07 \cdot 1.0) / (0.2 \cdot 1.0)) \\
 &= 2.124 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (t/d)^2 \cdot ((S \cdot E) / (C \cdot Z)) \text{ per UG-34 (c) (3)} \\
 &= (10.0/566.0)^2 \cdot ((137.9 \cdot 1.0) / (0.2 \cdot 1.0)) \\
 &= 2.152 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (Z \cdot C \cdot P) / (((t/d)^2) \cdot E) \\
 &= (1.0 \cdot 0.2 \cdot 0.5) / (((10.0/566.0)^2) \cdot 1.0) \\
 &= 32.037 \text{ N./mm}^2
 \end{aligned}$$

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Cylindrical Shell From 50 To 60 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 \cdot 283.0) / (136.07 \cdot 1.0 - 0.6 \cdot 0.5) \\
 &= 0.1040 + 0.0000 = 0.1040 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (136.07 \cdot 1.0 \cdot 3.0) / (283.0 + 0.6 \cdot 3.0) \\
 &= 14.332 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (283.0 + 0.6 \cdot 3.0) \\
 &= 14.525 \text{ bars}
 \end{aligned}$$

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Internal Pressure Calculations: Step: 7 4:19pm Feb 20, 2026

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P * (R + 0.6 * t)) / (E * t) \\
 &= (0.5 * (283.0 + 0.6 * 3.0)) / (1.0 * 3.0) \\
 &= 4.747 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 * t_{nom} / R_f) * (1 - R_f / R_o)$ 0.527 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Conical Section From 60 To 70 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P * D) / (2 * \cos(a) * (S * E - 0.6 * P)) \text{ per Appendix 1-4 (e)} \\
 &= (0.5 * 1606.0) / (2 * 0.9855 * (136.07 * 1.0 - 0.6 * 0.5)) \\
 &= 0.2995 + 0.0000 = 0.2995 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (2 * S * E * t * \cos(a)) / (D + 1.2 * t * \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 * 136.07 * 1.0 * 3.0 * 0.9855) / (1606.0 + 1.2 * 3.0 * 0.9855) \\
 &= 4.998 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (2 * S * E * t * \cos(a)) / (D + 1.2 * t * \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 * 137.9 * 1.0 * 3.0 * 0.9855) / (1606.0 + 1.2 * 3.0 * 0.9855) \\
 &= 5.065 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P * (D + 1.2 * t * \cos(a))) / (2 * E * t * \cos(a)) \\
 &= (0.5 * (1606.0 + 1.2 * 3.0 * 0.9855)) / (2 * 1.0 * 3.0 * 0.9855) \\
 &= 13.612 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 * t_{nom} / R_f) * (1 - R_f / R_o)$ 0.527 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

Note: The Pressure at the Large Diameter is used in the TR calculation.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Cylindrical Shell From 70 To 80 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P * R) / (S * E - 0.6 * P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 * 803.0) / (136.07 * 1.0 - 0.6 * 0.5) \\
 &= 0.2952 + 0.0000 = 0.2952 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S * E * t) / (R + 0.6 * t) \text{ per UG-27 (c) (1)} \\
 &= (136.07 * 1.0 * 3.0) / (803.0 + 0.6 * 3.0) \\
 &= 5.072 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

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$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (803.0 + 0.6 \cdot 3.0) \\
 &= 5.140 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t) \\
 &= (0.5 \cdot (803.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0) \\
 &= 13.414 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.186 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Welded Flat Head From 80 To 90 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= d \cdot \sqrt{Z \cdot C \cdot P / (S \cdot E)} \text{ per UG-34 (c) (3)} \\
 &= 898.0 \cdot \sqrt{1.0 \cdot 0.2 \cdot 0.5 / (136.07 \cdot 1.0)} \\
 &= 7.6986 + 0.0000 = 7.6986 \text{ mm.}
 \end{aligned}$$

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (t/d)^2 \cdot ((S \cdot E) / (C \cdot Z)) \text{ UG-34 (c) (3)} \\
 &= (14.0 / 898.0)^2 \cdot ((136.07 \cdot 1.0) / (0.2 \cdot 1.0)) \\
 &= 1.653 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (t/d)^2 \cdot ((S \cdot E) / (C \cdot Z)) \text{ per UG-34 (c) (3)} \\
 &= (14.0 / 898.0)^2 \cdot ((137.9 \cdot 1.0) / (0.2 \cdot 1.0)) \\
 &= 1.676 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (Z \cdot C \cdot P) / (((t/d)^2) \cdot E) \\
 &= (1.0 \cdot 0.2 \cdot 0.5) / (((14.0 / 898.0)^2) \cdot 1.0) \\
 &= 41.145 \text{ N./mm}^2
 \end{aligned}$$

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Cylindrical Shell From 90 To 100 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 \cdot 449.0) / (136.07 \cdot 1.0 - 0.6 \cdot 0.5) \\
 &= 0.1650 + 0.0000 = 0.1650 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (136.07 \cdot 1.0 \cdot 3.0) / (449.0 + 0.6 \cdot 3.0) \\
 &= 9.055 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (449.0 + 0.6 \cdot 3.0) \\
 &= 9.176 \text{ bars}
 \end{aligned}$$

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Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P * (R + 0.6 * t)) / (E * t) \\
 &= (0.5 * (449.0 + 0.6 * 3.0)) / (1.0 * 3.0) \\
 &= 7.514 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 * t_{nom} / R_f) * (1 - R_f / R_o)$ 0.333 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Elements Suitable for Internal Pressure.

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External Pressure Calculations: Step: 8 4:19pm Feb 20,2026

External Pressure Calculation Results :**External Pressure Calculations:**

From	To	Section Length mm.	Outside Diameter mm.	Corroded Thickness mm.	Factor A	Factor B N./mm ²
10	Ring	No Calc	...	3	No Calc	No Calc
Ring	20	50	327	3	0.010394	98.9477
20	Ring	706.449	1126.74	3	0.00026759	25.6883
Ring	30	4.99191	1130.39	3	0.0054666	92.8731
30	40	456	1130	3	0.00047288	45.3961
40	50	No Calc	...	10	No Calc	No Calc
50	Ring	108	572	3	0.0031947	86.5975
Ring	60	7	572	3	0.021854	104.027
60	Ring	4.99249	573.813	3	0.021011	103.811
Ring	Ring	1374.53	1218.51	3	0.00014001	13.4413
Ring	Ring	997.779	1610.36	3	0.00017041	16.3596
Ring	70	4.99741	1612.09	3	0.003216	86.6818
70	Ring	100	1612	3	0.0024223	82.9383
Ring	Ring	1510	1612	3	0.00011383	10.9277
Ring	80	904	1612	3	0.000193	18.528
80	90	No Calc	...	14	No Calc	No Calc
90	Ring	195	904	3	0.001308	73.751
Ring	100	5	904	3	0.0091683	97.8841

External Pressure Calculations:

From	To	External Actual T. mm.	External Required T. mm.	External Design Pressure bars	External M.A.W.P. bars
10	Ring	3	No Calc	0.1	No Calc
Ring	20	3	0.19212	0.1	12.103
20	Ring	3	1.27187	0.1	0.85662
Ring	30	3	0.43542	0.1	3.08702
30	40	3	0.99796	0.1	1.60685
40	50	10	No Calc	0.1	No Calc
50	Ring	3	0.36718	0.1	6.05541
Ring	60	3	0.20698	0.1	7.27422
60	Ring	3	0.2107	0.1	7.13086
Ring	Ring	3	1.66204	0.1	0.43479
Ring	Ring	3	1.72473	0.1	0.40042
Ring	70	3	0.59194	0.1	2.11938
70	Ring	3	0.64382	0.1	2.0579
Ring	Ring	3	2.02088	0.1	0.27114
Ring	80	3	1.63721	0.1	0.45972
80	90	14	No Calc	0.1	No Calc
90	Ring	3	0.61369	0.1	3.26313
Ring	100	3	0.32711	0.1	4.3309
Minimum				0.271	

External Pressure Calculations:

From	To	Actual Length Bet. Stiffeners mm.	Allowable Length Bet. Stiffeners mm.	Ring Inertia Required cm**4	Ring Inertia Available cm**4
10	Ring	No Calc	No Calc	No Calc	No Calc
Ring	20	50	4811.25	0.00062658	9.67321
20	Ring	706.449	706.449	No Calc	No Calc

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Ring	30	4.99191	4.99191	0.36588	20.652
30	40	456	6566.1	No Calc	No Calc
40	50	No Calc	No Calc	No Calc	No Calc
50	Ring	108	4021.26	No Calc	No Calc
Ring	60	7	487.092	0.0077134	17.0396
60	Ring	4.99249	4.99249	No Calc	No Calc
Ring	Ring	1374.53	1374.53	0.093369	17.0481
Ring	Ring	997.779	997.779	1.53939	18.2173
Ring	70	4.99741	4.99741	1.5006	19.1386
70	Ring	100	2398.86	No Calc	No Calc
Ring	Ring	1510	3984.95	2.40723	18.3065
Ring	80	904	3985.66	3.60856	18.3065
80	90	No Calc	No Calc	No Calc	No Calc
90	Ring	195	5498.92	No Calc	No Calc
Ring	100	5	216.544	0.052954	4.51783

Elements Suitable for External Pressure.

ASME Code, Section VIII Division 1, 2017**Cone From 20 to Ring 2-1 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.818	1126.74	706.45	399.82	0.6270	0.0002676	25.69

EMAP = $(4*B)/(3*(D/t)) = (4*25.6883)/(3*399.8153) = 0.8566$ bars

Note: The cone thickness used in the calculation has been modified
per UG-33(f), $t_e = t * \cos(\alpha)$.

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.195	1126.74	706.45	943.05	0.6270	0.0000737	7.07

EMAP = $(4*B)/(3*(D/t)) = (4*7.0738)/(3*943.0535) = 0.1$ bars

Note: The cone thickness used in the calculation has been modified
per UG-33(f), $t_e = t * \cos(\alpha)$.

Cone From Ring 2-1 to 30 Ext. Chart: HA-2 at 50 °C**Cylindrical Shell From 30 to 40 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1130.00	456.00	376.67	0.4035	0.0004729	45.40

EMAP = $(4*B)/(3*(D/t)) = (4*45.3961)/(3*376.6667) = 1.6068$ bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
0.998	1130.00	456.00	1132.32	0.4035	0.0000885	8.49

EMAP = $(4*B)/(3*(D/t)) = (4*8.4934)/(3*1132.3154) = 0.1$ bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1130.00	6566.10	376.67	5.8107	0.0000295	2.83

EMAP = $(4*B)/(3*(D/t)) = (4*2.8274)/(3*376.6667) = 0.1001$ bars

Welded Flat Head

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Note: This element's required thickness was computed in the internal Pressure Report using the maximum of the Internal and External pressures.

Cylindrical Shell From 50 to Ring 5-1 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	572.00	108.00	190.67	0.1888	0.0031947	86.60

EMAP = $(4*B)/(3*(D/t)) = (4*86.5975)/(3*190.6667) = 6.0554$ bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
0.367	572.00	108.00	1557.80	0.1888	0.0001217	11.68

EMAP = $(4*B)/(3*(D/t)) = (4*11.6846)/(3*1557.8004) = 0.1$ bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	572.00	4021.26	190.67	7.0302	0.0000649	6.23

EMAP = $(4*B)/(3*(D/t)) = (4*6.2274)/(3*190.6667) = 0.4355$ bars

Cone From 60 to Ring 6-1 Ext. Chart: HA-2 at 50 °C**Cone From Ring 6-1 to Ring 6-2 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.956	1218.51	1374.53	412.17	1.1280	0.0001400	13.44

EMAP = $(4*B)/(3*(D/t)) = (4*13.4413)/(3*412.1667) = 0.4348$ bars

Note: The cone thickness used in the calculation has been modified per UG-33(f), $t_e = t * \cos(\alpha)$.

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.638	1218.51	1374.53	743.97	1.1280	0.0000581	5.58

EMAP = $(4*B)/(3*(D/t)) = (4*5.5803)/(3*743.9673) = 0.1$ bars

Note: The cone thickness used in the calculation has been modified per UG-33(f), $t_e = t * \cos(\alpha)$.

Cone From Ring 6-2 to Ring 6-3 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.956	1610.36	997.78	544.71	0.6196	0.0001704	16.36

EMAP = $(4*B)/(3*(D/t)) = (4*16.3596)/(3*544.7132) = 0.4004$ bars

Note: The cone thickness used in the calculation has been modified per UG-33(f), $t_e = t * \cos(\alpha)$.

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.700	1610.36	997.78	947.48	0.6196	0.0000740	7.11

EMAP = $(4*B)/(3*(D/t)) = (4*7.1067)/(3*947.4755) = 0.1$ bars

Note: The cone thickness used in the calculation has been modified per UG-33(f), $t_e = t * \cos(\alpha)$.

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Cone From Ring 6-3 to 70 Ext. Chart: HA-2 at 50 °C**Cylindrical Shell From Ring 7-1 to Ring 7-2 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1612.00	1510.00	537.33	0.9367	0.0001138	10.93

$$EMAP = (4*B) / (3*(D/t)) = (4*10.9277) / (3*537.3333) = 0.2711 \text{ bars}$$
Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.021	1612.00	1510.00	797.67	0.9367	0.0000623	5.98

$$EMAP = (4*B) / (3*(D/t)) = (4*5.983) / (3*797.673) = 0.1 \text{ bars}$$
Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1612.00	3984.95	537.33	2.4721	0.0000420	4.03

$$EMAP = (4*B) / (3*(D/t)) = (4*4.0331) / (3*537.3333) = 0.1001 \text{ bars}$$
Cylindrical Shell From Ring 7-2 to 80 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1612.00	904.00	537.33	0.5608	0.0001930	18.53

$$EMAP = (4*B) / (3*(D/t)) = (4*18.528) / (3*537.3333) = 0.4597 \text{ bars}$$
Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.637	1612.00	904.00	984.60	0.5608	0.0000769	7.39

$$EMAP = (4*B) / (3*(D/t)) = (4*7.3851) / (3*984.6019) = 0.1 \text{ bars}$$
Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1612.00	3985.66	537.33	2.4725	0.0000420	4.03

$$EMAP = (4*B) / (3*(D/t)) = (4*4.0324) / (3*537.3333) = 0.1001 \text{ bars}$$
Welded Flat Head

Note: This element's required thickness was computed in the internal Pressure Report using the maximum of the Internal and External pressures.

Cylindrical Shell From 90 to Ring 8.1 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	904.00	195.00	301.33	0.2157	0.0013080	73.75

$$EMAP = (4*B) / (3*(D/t)) = (4*73.751) / (3*301.3333) = 3.2631 \text{ bars}$$
Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
0.614	904.00	195.00	1473.05	0.2157	0.0001151	11.05

$$EMAP = (4*B) / (3*(D/t)) = (4*11.0491) / (3*1473.0518) = 0.1 \text{ bars}$$
Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	904.00	5498.92	301.33	6.0829	0.0000385	3.70

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$$\text{EMAP} = (4*B)/(3*(D/t)) = (4*3.696)/(3*301.3333) = 0.1635 \text{ bars}$$

Stiffening Ring Calcs for : Ring 1-1 , SA-240 316 , Bar Ring: 39 x 12 mm.

Effective Length of Shell: 34 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	1.034	1.5000	1.550
Ring :	4.680	22.5000	105.300
Total:	5.714		106.850

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.008	17.2011	305.816
Ring :	5.932	-3.7989	67.540
Total:	5.940		373.357

Available Moment of Inertia, Ring plus Shell: 10 cm**4

Required Stress in Ring plus Shell Breq 0.11 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000012

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq)/10.9$$

$$= (327.0^2 * 25.0(3.0 + 4.68/25.0)0.000001)/10.9$$

$$= 0 \text{ cm**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style CONTINUOUS
 Location of Stiffening Ring EXTERNAL
 Radial Pressure Load Pext*Slen 0.03 Kgf/mm.
 The Radial Shear Load V 0.08 Kgf
 The First Moment of the Area (Ring + Shell) Q 1777.89 mm.³
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.00 Kgf/mm.
 The Weld Allowable Stress 0.55*S 75.85 N./mm²
 Minimum Weld Leg Size Min(6mm, t, tw) Wldmin 3.00 mm.
 The Weld Allowable Load 2* WLeg*0.55*S 46.40 Kgf/mm.
 The Combined Weld Load SRSS of VQ/I and Pext*Slen 0.03 Kgf/mm.

Stiffening Ring Calcs for : Ring 2-1 , SA-240 316 , Bar Ring: 50 x 10 mm.

Effective Length of Shell: 64 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	1.918	1.5000	2.877
Ring :	5.046	28.2300	142.449
Total:	6.964		145.326

Centroid of Ring plus Shell: 21 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.014	19.3674	719.536
Ring :	10.707	-7.3627	273.537
Total:	10.721		993.073

Available Moment of Inertia, Ring plus Shell: 21 cm**4

Required Stress in Ring plus Shell Breq 1.91 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000200

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Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1126.35^2 * 355.7206(3.0 + 5.046/355.7206)0.00002) / 10.9$$

$$= 0 \text{ cm}^4$$

Stiffening Ring Calcs for : Ring 5-1 , SA-240 304 , Bar Ring: 49 x 10 mm.

Effective Length of Shell: 46 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	1.367	1.5000	2.051
Ring :	4.900	27.5000	134.750
Total:	6.267		136.801

Centroid of Ring plus Shell: 22 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.010	20.3287	564.925
Ring :	9.804	-5.6713	157.604
Total:	9.814		722.529

Available Moment of Inertia, Ring plus Shell: 17 cm⁴

Required Stress in Ring plus Shell Breq 0.37 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000039

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (572.0^2 * 57.5(3.0 + 4.9/57.5)0.000004) / 10.9$$

$$= 0 \text{ cm}^4$$

Stiffening Ring Calcs for : Ring 6-1 , SA-240 316 , Bar Ring: 49 x 10 mm.

Effective Length of Shell: 46 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	1.369	1.5000	2.054
Ring :	4.900	27.5000	134.750
Total:	6.269		136.804

Centroid of Ring plus Shell: 22 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.010	20.3220	565.404
Ring :	9.804	-5.6780	157.975
Total:	9.814		723.379

Available Moment of Inertia, Ring plus Shell: 17 cm⁴

Required Stress in Ring plus Shell Breq 1.16 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000121

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (573.7247^2 * 689.7628(3.0 + 4.9/689.7628)0.000012) / 10.9$$

$$= 0 \text{ cm}^4$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.
 Stiffening Ring Attachment Style INTERMITTENT

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External Pressure Calculations: Step: 8 4:19pm Feb 20,2026

Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.70	Kgf/mm.
The Radial Shear Load	V	4.04	Kgf
The First Moment of the Area (Ring + Shell)	Q	2782.23	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.07	Kgf/mm.
The Weld Allowable Stress	0.55*S	75.85	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
Maximum Space between Welds 8*TCA		50.80	mm.
The Weld Allowable Load	WLeg*0.55*S	23.20	Kgf/mm.
The Combined Weld Load SRSS of VQ/I and Pext*Slen		0.71	Kgf/mm.

Stiffening Ring Calcs for : Ring 6-2 , SA-240 316 , Bar Ring: 48 x 10 mm.

Effective Length of Shell: 67 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	1.995	1.5000	2.993
Ring :	4.782	26.9100	128.684
Total:	6.777		131.676

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.015	17.9295	641.370
Ring :	9.113	-7.4805	267.592
Total:	9.128		908.962

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell Breq 2.69 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000280

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1218.4214^2 * 1186.1562(3.0 + 4.782/1186.1562)0.000028) / 10.9$$

$$= 2 \text{ cm}^{**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	INTERMITTENT		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	1.21	Kgf/mm.
The Radial Shear Load	V	14.74	Kgf
The First Moment of the Area (Ring + Shell)	Q	3577.18	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.29	Kgf/mm.
The Weld Allowable Stress	0.55*S	75.85	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
Maximum Space between Welds 8*TCA		50.80	mm.
The Weld Allowable Load	WLeg*0.55*S	23.20	Kgf/mm.
The Combined Weld Load SRSS of VQ/I and Pext*Slen		1.24	Kgf/mm.

Stiffening Ring Calcs for : Ring 6-3 , SA-240 316 , Bar Ring: 48 x 10 mm.

Effective Length of Shell: 76 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.294	1.5000	3.440
Ring :	4.782	26.9100	128.684
Total:	7.076		132.124

Centroid of Ring plus Shell: 19 mm.

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External Pressure Calculations: Step: 8 4:19pm Feb 20,2026

	Inertia	Distance	A*Dist ²
Shell:	0.017	17.1731	676.429
Ring :	9.113	-8.2369	324.442
Total:	9.130		1000.870

Available Moment of Inertia, Ring plus Shell:	19 cm**4		
Required Stress in Ring plus Shell	Breq	3.05 N./mm ²	
Required Strain in Ring plus Shell	Areq	0.0000318	

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1610.2754^2 * 501.3885(3.0 + 4.782/501.3885)0.000032) / 10.9$$

$$= 2 \text{ cm}^{*4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	INTERMITTENT		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.51	Kgf/mm.
The Radial Shear Load	V	8.23	Kgf
The First Moment of the Area (Ring + Shell)	Q	3938.88	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.17	Kgf/mm.
The Weld Allowable Stress	0.55*S	75.85	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
Maximum Space between Welds 8*TCA		50.80	mm.
The Weld Allowable Load	WLeg*0.55*S	23.20	Kgf/mm.
The Combined Weld Load SRSS of VQ/I and Pext*Slen		0.54	Kgf/mm.

Stiffening Ring Calcs for : Ring 7-1 , SA-240 316 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 76 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.295	1.5000	3.442
Ring :	4.700	26.5000	124.550
Total:	6.995		127.993

Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.017	16.7981	647.554
Ring :	8.652	-8.2020	316.180
Total:	8.669		963.734

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell	Breq	3.37 N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000350

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1612.0^2 * 805.0(3.0 + 4.7/805.0)0.000035) / 10.9$$

$$= 2 \text{ cm}^{*4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.82	Kgf/mm.
The Radial Shear Load	V	13.23	Kgf
The First Moment of the Area (Ring + Shell)	Q	3854.93	mm. ³

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External Pressure Calculations: Step: 8 4:19pm Feb 20,2026

Weld Shear Flow due to Rad. Shear Load	VQ/I	0.28	Kgf/mm.
The Weld Allowable Stress	0.55*S	75.85	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	46.40	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.87	Kgf/mm.

Stiffening Ring Calcs for : Ring 7-2 , SA-240 316 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 76 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.295	1.5000	3.442
Ring :	4.700	26.5000	124.550
Total:	6.995		127.993

Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.017	16.7981	647.554
Ring :	8.652	-8.2020	316.180
Total:	8.669		963.734

Available Moment of Inertia, Ring plus Shell: 18 cm**4

Required Stress in Ring plus Shell	Breq	3.57	N./mm ²
Required Strain in Ring plus Shell	Areq	0.0000370	

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1612.0^2 * 1207.0 (3.0 + 4.7/1207.0) 0.000037) / 10.9$$

$$= 4 \text{ cm}^{**4}$$

Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	1.23	Kgf/mm.
The Radial Shear Load	V	19.84	Kgf
The First Moment of the Area (Ring + Shell)	Q	3854.93	mm. ³
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.42	Kgf/mm.
The Weld Allowable Stress	0.55*S	75.85	N./mm ²
Minimum Weld Leg Size Min(6mm, t, tw)	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	46.40	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	1.30	Kgf/mm.

Stiffening Ring Calcs for : Ring 8.1 , SA-240 316 , Bar Ring: 26 x 14 mm.

Effective Length of Shell: 57 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	1.719	1.5000	2.578
Ring :	3.640	16.0000	58.240
Total:	5.359		60.818

Centroid of Ring plus Shell: 11 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.013	9.8497	166.726
Ring :	2.051	-4.6503	78.716
Total:	2.063		245.442

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External Pressure Calculations: Step: 8 4:19pm Feb 20,2026

Available Moment of Inertia, Ring plus Shell: 5 cm**4

Required Stress in Ring plus Shell Breq 1.02 N./mm²

Required Strain in Ring plus Shell Areq 0.0000106

Required Moment of Inertia, Ring plus Shell:

$$\begin{aligned}
 &= (OD^2 * Slen(Tca + Aring/Slen)Areq)/10.9 \\
 &= (904.0^2*100.0(3.0+3.64/100.0)0.000011)/10.9 \\
 &= 0 \text{ cm**4}
 \end{aligned}$$

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Element and Detail Weights: Step: 9 4:19pm Feb 20,2026

Element and Detail Weights:

From	To	Element Metal Wgt. kg.	Element ID Volume Cm.	Corroded Metal Wgt. kg.	Corroded ID Volume Cm.	Extra due Misc % kg.
10	20	1.22562	4047.14	1.22562	4047.14	...
20	30	64.2908	497493	64.2908	497493	...
30	40	38.8802	452548	38.8802	452548	...
40	50	79.6512	...	79.6512
50	60	4.95053	28940	4.95053	28940	...
60	70	252.081	3006753	252.081	3006753	...
70	80	306.025	5093591	306.025	5093591	...
80	90	227.656	...	227.656
90	100	13.6332	126692	13.6332	126692	...
Total		988	9210065.00	988	9210065.00	0

Weight of Details:

From	Type	Weight of Detail kg.	X Offset, Dtl. Cent. mm.	Y Offset, Dtl. Cent. mm.	Description
10	Ring	4.31963	Ring 1-1
20	Ring	14.9752	...	1095	Ring 2-1
50	Ring	7.67375	...	108	Ring 5-1
60	Ring	7.67041	...	5	Ring 6-1
60	Ring	15.2695	...	1874	Ring 6-2
60	Ring	19.9959	...	3010	Ring 6-3
70	Ring	19.6637	...	100	Ring 7-1
70	Ring	19.6637	...	1610	Ring 7-2
90	Ring	8.53696	...	195	Ring 8.1

Total Weight of Each Detail Type:

Stiffeners	117.8
Sum of the Detail Weights	117.8 kg.

Weight Summation Results: (kg.)

	Fabricated	Shop Test	Shipping	Erected	Empty	Operating
Main Elements	988.4	988.4	988.4	988.4	988.4	988.4
Stif. Rings	117.8	117.8	117.8	117.8	117.8	117.8
Test Liquid	...	9204.4
Totals	1106.2	10310.6	1106.2	1106.2	1106.2	1106.2

Weight Summary:

Fabricated Wt.	- Bare Weight without Removable Internals	1106.2 kg.
Shop Test Wt.	- Fabricated Weight + Water (Full)	10310.6 kg.
Shipping Wt.	- Fab. Weight + removable Intls.+ Shipping App.	1106.2 kg.
Erected Wt.	- Fab. Wt + or - loose items (trays,platforms etc.)	1106.2 kg.
Ope. Wt. no Liq	- Fab. Weight + Internals. + Details + Weights	1106.2 kg.
Operating Wt.	- Empty Weight + Operating Liq. Uncorroded	1106.2 kg.
Field Test Wt.	- Empty Weight + Water (Full)	10310.6 kg.
Mass of the Upper 1/3 of the Vertical Vessel		546.7 kg.

Outside Surface Areas of Elements:

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Element and Detail Weights: Step: 9 4:19pm Feb 20,2026

From	To	Surface Area cm ²
10	20	513.65
20	30	26814
30	40	16188
40	50	...
50	60	2066.54
60	70	104969
70	80	127315
80	90	...
90	100	5680
Total		283546.125 cm ²

Element and Detail Weights:

From	To	Total Ele. Empty Wgt. kg.	Total. Ele. Oper. Wgt. kg.	Total. Ele. Hydro. Wgt. kg.	Total Dtl. Offset Mom. Kg-m.	Oper. Wgt. No Liquid kg.
10	20	5.54525	5.54525	5.54525	...	5.54525
20	30	79.2661	79.2661	79.2661	...	79.2661
30	40	38.8802	38.8802	38.8802	...	38.8802
40	50	79.6512	79.6512	79.6512	...	79.6512
50	60	12.6243	12.6243	12.6243	...	12.6243
60	70	295.017	295.017	295.017	...	295.017
70	80	345.352	345.352	345.352	...	345.352
80	90	227.656	227.656	227.656	...	227.656
90	100	22.1701	22.1701	22.1701	...	22.1701

Cumulative Vessel Weight

From	To	Cumulative Ope Wgt. No Liquid kg.	Cumulative Oper. Wgt. kg.	Cumulative Hydro. Wgt. kg.
10	20	1106.16	1106.16	1106.16
20	30	1100.62	1100.62	1100.62
30	40	1021.35	1021.35	1021.35
40	50	982.471	982.471	982.471
50	60	902.82	902.82	902.82
60	70	890.195	890.195	890.195
70	80	595.178	595.178	595.178
80	90	249.826	249.826	249.826
90	100	22.1701	22.1701	22.1701

Note: The cumulative operating weights no liquid in the column above
are the cumulative operating weights minus the operating liquid
weight minus any weights absent in the empty condition.

Cumulative Vessel Moment

From	To	Cumulative Empty Mom. Kg-m.	Cumulative Oper. Mom. Kg-m.	Cumulative Hydro. Mom. Kg-m.
10	20
20	30
30	40
40	50
50	60
60	70

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70	80	
80	90	
90	100	

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Conical Section:

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Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1**Conical Section From 20 To 30 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	10.2926 mm.
Maximum Centroid Reinforcement Distance Small End	5.5368 mm.

Note: No ring was found close enough to the small end to be considered.

Reinforcement Calculations for Cone / Large Cylinder:**Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00037
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	20.052 degrees

Intermediate Value [k]:

$$= \max(Y / (S_r * E_{rl}), 1)$$

$$= \max(0.25122E+11 / (136.1 * 0.18458E+09), 1)$$

$$= 1.0002$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)}$$

$$= 136.1 * 0.18458E+09$$

$$= 25121794048.0 \text{ N./mm}^2$$

Decay Length, Cone Large End:

$$= 2 * \sqrt{RL(ts - ca)}$$

$$= 2 * \sqrt{562.0(3.0 - 0.0)}$$

$$= 82.122 \text{ mm.}$$
Required Area of Reinforcement, Large End, Internal [Arl]:

$$= k * Q_L * R_L / (S_s * E_L) * (1 - \delta / \alpha) * \tan(\alpha)$$

$$= 1.0 * 1.4328 * 562.0 / (136 * 1.0) *$$

$$(1 - 11.0 / 20.05) * 0.365$$

$$= 0.0956 \text{ cm}^2$$

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Force per Length, Cone Large End [QL]:

$$\begin{aligned}
 &= P(Rl/2) - Faxial/(pi(Dl + Tl)) + Moment/(pi(Rl + Tl/2)(Rl + Tl/2)) \\
 &= 0.5(562.0/2) - 0.0/(pi(1124.0 + 3.0)) + \\
 &\quad 0.0/(pi(562.0 + 3.0/2)(562.0 + 3.0/2)) \\
 &= 1.433 \text{ Kg/mm.}
 \end{aligned}$$

Area of Reinforcement Available in Large end Shell [Ael]:

$$\begin{aligned}
 &= (Ts-t) * \text{sqrt}(Rl*Ts) + (Tc-Tr) * \text{sqrt}(Rl*Tc/\cos(\alpha)) \\
 &= (3.0 - 0.2066) * \text{SQRT}(562.0 * 3.0) + \\
 &\quad (3.0 - 0.2199) * \text{SQRT}(562.0 * 3.0/0.9394) \\
 &= 2.3248 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement area, Large end, Internal Pressure:

Area of reinforcement required per App. 1-5(1)	0.0956	cm ²
Area of reinforcement in shell per App. 1-5(2)	2.3248	cm ²
Area of reinforcement in stiffening ring	5.0460	cm ²

Required Area of Reinforcement for Large End Under External Pressure

Large end ratio of pressure to allowable stress	0.00007	
Large end max. half apex angle w/o reinforcement	0.181	degrees
Large end actual half apex angle	20.052	degrees

Intermediate Value [k]:

$$\begin{aligned}
 &= \max(Y / (Srl * Erl), 1) \\
 &= \max(0.26602E+11 / (137.9 * 0.19291E+09), 1) \\
 &= 1.0000
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)} \\
 &= 137.9 * 0.19291E+09 \\
 &= 26602469376.0 \text{ N./mm}^2
 \end{aligned}$$

Allowable Stress of Large End Material (Ext. Temp)	137.9	N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	137.9	N./mm ²

Required Area of Reinforcement, Large End, External [Arl]:

$$\begin{aligned}
 &= (k*QL*Rl*\tan(\text{angle})/(Ss*E1)) * (1-1/4((P*Rl-QL)/QL)) * (\text{delta}/\alpha) \\
 &= (1.0*0.2881*565.0*0.365/(138*1.0)) * \\
 &\quad (1-1/4((0.1*565.0-0.288)/0.288)) * (0.181/20.052) \\
 &= 0.0422 \text{ cm}^2
 \end{aligned}$$

Force per Length, Cone Large End External Pressure [QL]:

$$\begin{aligned}
 &= Pext(Rl/2) + Faxial/(pi(Dl - Tl)) + Moment/(pi(Rl - Tl/2)(Rl - Tl/2)) \\
 &= 0.1(565.0/2) + 0.0/(pi(1130.0 - 3.0)) + \\
 &\quad 0.0/(pi(565.0 - 3.0/2)(565.0 - 3.0/2)) \\
 &= 0.288 \text{ Kg/mm.}
 \end{aligned}$$

Available Area of Reinforcement, Large End, External [Ael]:

$$\begin{aligned}
 &= 0.55 * (Dl*ts)^{1/2} * (ts + tc/\cos(\alpha)) \\
 &= 0.55 * (1130.0 * 3.0)^{1/2} * (3.0 + 3.0/0.939) \\
 &= 1.9834 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement Area, Large End, External Pressure:

Area of reinforcement required per App. 1-8(1)	0.0422	cm ²
Area of reinforcement in shell per App. 1-8(2)	1.9834	cm ²
Area of reinforcement in stiffening ring	5.0460	cm ²

Intermediate Results, Large End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	29.45	cm ²
Force per Unit Length on Shell / Cone Junction	0.61	Kgf/mm.
Actual Buckling Stress associated with this Force	1.72	N./mm ²
Material Strain associated with this stress	0.000018	

Required Moment of Inertia, Large End, External Pressure [I's]:

$$= A * Dl^2 * Atl / 10.9$$

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$$= 0.178\text{E-}04 * 1130.0 * 1130.0 * 29.45/10.9$$

$$= 6151.72 \text{ mm}^4$$

Available Moment of Inertia, Large End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	0.961	0.0000	0.000	18.3378	0.007	323.055
Con	1.023	-5.8442	-5.977	24.1820	0.125	598.033
Sec	5.046	26.7300	134.880	-8.3923	10.707	355.391
TOT	7.029		128.903		10.839	1276.479
Centroid of Section			18.3378	Moment of Inertia		23.60

Summary of Large End Inertia Calculations

Available Moment of Inertia (Large End)	23.604	cm**4
Required Moment of Inertia (Large End)	0.615	cm**4

Reinforcement Calculations for Cone / Small Cylinder:

Required Area of Reinforcement for Small End under Internal Pressure

Small end ratio of pressure to allowable stress	0.00037	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	20.052	degrees

Intermediate Value [k]:

$$= \max(Y / (Sr * Ers), 1)$$

$$= \max(0.25122\text{E}+11 / (136.1 * 0.18458\text{E}+09), 1)$$

$$= 1.0002$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)}$$

$$= 136.1 * 0.18458\text{E}+09$$

$$= 25121794048.0 \text{ N./mm}^2$$

Decay Length, Cone Small End:

$$= 1.4 * \sqrt{Rs(ts - ca)}$$

$$= 1.4 * \sqrt{160.5(3.0 - 0.0)}$$

$$= 30.720 \text{ mm.}$$

Required Area of Reinforcement, Small End, Internal [Ars]:

$$= k * QS * Rs / (Ss * E1) * (1 - \delta / \alpha) * \tan(\alpha)$$

$$= 1.0 * 0.4092 * 160.5 / (136 * 1.0) *$$

$$(1 - 4.0 / 20.05) 0.365$$

$$= 0.0138 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= P(Rs/2) - \text{Faxial} / (\pi (Ds + Ts)) + \text{Moment} / (\pi (Rs + Ts/2) (Rs + Ts/2))$$

$$= 0.5(160.5/2) - 0.0 / (\pi (321.0 + 3.0)) +$$

$$0.0 / (\pi (160.5 + 3.0/2) (160.5 + 3.0/2))$$

$$= 0.409 \text{ Kg/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.78 (Rs * Ts)^{1/2} * ((Ts - t) + (Tc - Tr) / \cos(\alpha))$$

$$= 0.78 (160.5 * 3.0)^{1/2} * ((3.0 - 0.059) + (3.0 - 0.063) / 0.94)$$

$$= 1.0385 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, Internal Pressure:

Area of reinforcement required per App. 1-5(3)	0.0138	cm ²
Area of reinforcement in shell per App. 1-5(4)	1.0385	cm ²
Area of reinforcement in stiffening ring	0.0000	cm ²

Required Area of Reinforcement for Small End Under External Pressure

Allowable Stress of Small End Material (Ext. Temp)	137.9	N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	137.9	N./mm ²

Intermediate Value [k]:

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$$= \max(Y / (Srs * Ers), 1)$$

$$= \max(0.26602E+11 / (137.9 * 0.19291E+09), 1)$$

$$= 1.0000$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 137.9 * 0.19291E+09$$

$$= 26602469376.0 \text{ N./mm}^2$$

Area of Reinforcement Required in Small End Shell [Ars]:

$$= k * QS * Rs * \tan(\alpha) / (Ss * E1)$$

$$= (1.0 * 0.0834 * 163.5 * 0.365 / (138 * 1.0))$$

$$= 0.00354 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= P_{ext}(Rs/2) + F_{axial} / (\pi(Ds - Ts)) + \text{Moment} / (\pi(Rs - Ts/2)(Rs - Ts/2))$$

$$= 0.1(163.5/2) + 0.0 / (\pi(327.0 - 3.0)) +$$

$$0.0 / (\pi(163.5 - 3.0/2)(163.5 - 3.0/2))$$

$$= 0.083 \text{ Kg/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.55 * (Ds * ts)^{1/2} * [(ts - t) + (tc - tr) / \cos(\alpha)]$$

$$= 0.55 * (327.0 * 3.0)^{1/2} * [(3.0 - 1.272) + (3.0 - 0.205) / 0.939]$$

$$= 0.8103 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, External Pressure:

Area of reinforcement required per App. 1-8(3)	0.0035	cm ²
Area of reinforcement in shell per App. 1-8(4)	0.8103	cm ²
Area of reinforcement in stiffening ring	0.0000	cm ²

Intermediate Results, Small End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	18.31	cm ²
Force per Unit Length on Shell / Cone Junction	0.89	Kgf/mm.
Actual Buckling Stress associated with this Force	1.17	N./mm ²
Material Strain associated with this stress	0.000012	

Required Moment of Inertia, Small End, External Pressure [I's]:

$$= A * Ds^2 * A_{ts} / 10.9$$

$$= 0.121E-04 * 327.0 * 327.0 * 18.31 / 10.9$$

$$= 217.43 \text{ mm}^4$$

Available Moment of Inertia, Small End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	0.517	0.0000	0.000	1.6211	0.004	1.358
Con	0.550	3.1438	1.730	-1.5228	0.023	1.276
Sec	0.000	1.5000	0.000	0.1211	0.000	0.000
TOT	1.067		1.730		0.027	2.634
Centroid of Section			1.6211	Moment of Inertia		0.05

Summary of Small End Inertia Calculations

Available Moment of Inertia (Small End)	0.5301E-01	cm**4
Required Moment of Inertia (Small End)	0.2174E-01	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

Results for Discontinuity Stresses per Bednar p. 236 2nd Edition

Stress Type	Stress	Allowable	Location
Tensile Stress	5.51	408.30	Small Cyl. Long.
Compres. Stress	-2.81	-408.30	Small Cyl. Long.
Membrane Stress	4.99	-204.15	Small End Tang.
Tensile Stress	5.60	408.30	Cone Longitudinal

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Compres. Stress	-2.72	-408.30	Cone Longitudinal
Compres Stress	5.16	-204.15	Cone Tangential
Tensile Stress	31.69	408.30	Large Cyl. Long.
Compres. Stress	-22.30	-408.30	Large Cyl. Long.
Membrane Stress	-5.47	-204.15	Large End Tang.
Tensile Stress	31.99	408.30	Cone Longitudinal
Compres. Stress	-21.99	-408.30	Cone Longitudinal
Compres Stress	-4.86	-204.15	Cone Tangential

Note: An asterisk (*) denotes that this stress was not applicable for this combination of loads.

Maximum Allowable Pressure Calculations for Cone to Shell Junction:

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	6.804	Thickness due to internal pressure, Cone Large End
MAPnc	6.893	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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Conical Section:

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Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1**Conical Section From 60 To 70 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	12.2933 mm.
Maximum Centroid Reinforcement Distance Small End	7.3229 mm.

Reinforcement Calculations for Cone / Large Cylinder:**Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00037
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	9.786 degrees

Required Area of Reinforcement for Large End Under External Pressure

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.181 degrees
Large end actual half apex angle	9.786 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.26602E+11 / (137.9 * 0.19291E+09), 1)$$

$$= 1.0000$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 137.9 * 0.19291E+09$$

$$= 26602469376.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	137.9 N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	137.9 N./mm ²

Required Area of Reinforcement, Large End, External [Arl]:

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\Delta / \alpha))$$

$$= (1.0 * 0.411 * 806.0 * 0.172 / (138 * 1.0)) * (1 - 1/4 * ((0.1 * 806.0 - 0.411) / 0.411) * (0.181 / 9.786))$$

$$= 0.0404 \text{ cm}^2$$

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Force per Length, Cone Large End External Pressure [QL]:

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(806.0/2) + 0.0/(\pi(1612.0 - 3.0)) + \\
 &\quad 0.0/(\pi(806.0 - 3.0/2)(806.0 - 3.0/2)) \\
 &= 0.411 \text{ Kg/mm.}
 \end{aligned}$$

Available Area of Reinforcement, Large End, External [Ael]:

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1612.0 * 3.0)^{1/2} * (3.0 + 3.0/0.985) \\
 &= 2.3118 \text{ cm}^2
 \end{aligned}$$

Summary of Reinforcement Area, Large End, External Pressure:

Area of reinforcement required per App. 1-8(1)	0.0404	cm ²
Area of reinforcement in shell per App. 1-8(2)	2.3118	cm ²
Area of reinforcement in stiffening ring	4.7820	cm ²

Intermediate Results, Large End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	52.17	cm ²
Force per Unit Length on Shell / Cone Junction	1.37	Kgf/mm.
Actual Buckling Stress associated with this Force	3.11	N./mm ²
Material Strain associated with this stress	0.000032	

Required Moment of Inertia, Large End, External Pressure [I's]:

$$\begin{aligned}
 &= A * D_l^2 * A_{t1} / 10.9 \\
 &= 0.322E-04 * 1612.0 * 1612.0 * 52.17/10.9 \\
 &= 40104.17 \text{ mm}^4
 \end{aligned}$$

Available Moment of Inertia, Large End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	1.147	0.0000	0.000	16.5877	0.009	315.719
Con	1.164	-3.2983	-3.840	19.8860	0.051	460.457
Sec	4.782	25.4100	121.511	-8.8223	9.113	372.194
TOT	7.094		117.670		9.173	1148.371
Centroid of Section			16.5877	Moment of Inertia		20.66

Summary of Large End Inertia Calculations

Available Moment of Inertia (Large End)	20.656	cm**4
Required Moment of Inertia (Large End)	4.010	cm**4

Reinforcement Calculations for Cone / Small Cylinder:**Required Area of Reinforcement for Small End under Internal Pressure**

Small end ratio of pressure to allowable stress	0.00037	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	9.786	degrees

Intermediate Value [k]:

$$\begin{aligned}
 &= \max(Y / (S_r * E_{rs}), 1) \\
 &= \max(0.25116E+11 / (136.1 * 0.18458E+09), 1) \\
 &= 1.0000
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)} \\
 &= 136.1 * 0.18458E+09 \\
 &= 25115742208.0 \text{ N./mm}^2
 \end{aligned}$$

Decay Length, Cone Small End:

$$\begin{aligned}
 &= 1.4 * \sqrt{R_s(t_s - c_a)} \\
 &= 1.4 * \sqrt{283.0(3.0 - 0.0)} \\
 &= 40.793 \text{ mm.}
 \end{aligned}$$

Required Area of Reinforcement, Small End, Internal [Ars]:

$$\begin{aligned}
 &= k * Q_S * R_s / (S_s * E_1) * (1 - \delta / \alpha) * \tan(\alpha) \\
 &= 1.0 * 0.7215 * 283.0 / (136 * 1.0) *
 \end{aligned}$$

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$$(1 - 4.0/9.79) \cdot 0.1725 \\ = 0.015 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= P(Rs/2) - Faxial/(\pi(Ds + Ts)) + Moment/(\pi(Rs + Ts/2)(Rs + Ts/2)) \\ = 0.5(283.0/2) - 0.0/(\pi(566.0 + 3.0)) + \\ 0.0/(\pi(283.0 + 3.0/2)(283.0 + 3.0/2)) \\ = 0.721 \text{ Kgf/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.78(Rs \cdot Ts)^{1/2} \cdot ((Ts - t) + (Tc - Tr) / \cos(\alpha)) \\ = 0.78(283.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 0.104) + (3.0 - 0.106) / 0.99) \\ = 1.3257 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, Internal Pressure:

Area of reinforcement required per App. 1-5(3)	0.0150	cm ²
Area of reinforcement in shell per App. 1-5(4)	1.3257	cm ²
Area of reinforcement in stiffening ring	4.9000	cm ²

Required Area of Reinforcement for Small End Under External Pressure

Allowable Stress of Small End Material (Ext. Temp)	137.9	N./mm ²
Allowable Stress of Cone Material (Ext. Temp)	137.9	N./mm ²

Intermediate Value [k]:

$$= \max(Y / (Srs \cdot Ers), 1) \\ = \max(0.26602E+11 / (137.9 \cdot 0.19291E+09), 1) \\ = 1.0000$$

where [Y] is:

$$= \text{Cone Allowable Stress} \cdot \text{Cone Elastic Modulus (Ext. temp.)} \\ = 137.9 \cdot 0.19291E+09 \\ = 26602469376.0 \text{ N./mm}^2$$

Area of Reinforcement Required in Small End Shell [Ars]:

$$= k \cdot QS \cdot Rs \cdot \tan(\alpha) / (Ss \cdot E1) \\ = (1.0 \cdot 0.1458 \cdot 286.0 \cdot 0.1725) / (138 \cdot 1.0) \\ = 0.00512 \text{ cm}^2$$

Force per Length, Cone Small End [QS]:

$$= Pext(Rs/2) + Faxial/(\pi(Ds - Ts)) + Moment/(\pi(Rs - Ts/2)(Rs - Ts/2)) \\ = 0.1(286.0/2) + 0.0/(\pi(572.0 - 3.0)) + \\ 0.0/(\pi(286.0 - 3.0/2)(286.0 - 3.0/2)) \\ = 0.146 \text{ Kgf/mm.}$$

Area of Reinforcement Available in Small End Shell [Aes]:

$$= 0.55 \cdot (Ds \cdot ts)^{1/2} \cdot ((ts - t) + (tc - tr) / \cos(\alpha)) \\ = 0.55 \cdot (572.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 0.211) + (3.0 - 0.21) / 0.985) \\ = 1.2805 \text{ cm}^2$$

Summary of Reinforcement Area, Small End, External Pressure:

Area of reinforcement required per App. 1-8(3)	0.0051	cm ²
Area of reinforcement in shell per App. 1-8(4)	1.2805	cm ²
Area of reinforcement in stiffening ring	4.9000	cm ²

Intermediate Results, Small End, External Pressure:

Area Available in Cone, Shell, and Reinforcement	52.41	cm ²
Force per Unit Length on Shell / Cone Junction	2.04	Kgf/mm.
Actual Buckling Stress associated with this Force	1.63	N./mm ²
Material Strain associated with this stress	0.000017	

Required Moment of Inertia, Small End, External Pressure [I's]:

$$= A \cdot Ds^2 \cdot Ats / 10.9 \\ = 0.169E-04 \cdot 572.0 \cdot 572.0 \cdot 52.41 / 10.9 \\ = 2666.48 \text{ mm}^4$$

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Available Moment of Inertia, Small End, External Pressure:

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di ²
Shl	0.684	0.0000	0.000	20.5131	0.005	287.610
Con	0.694	1.9648	1.363	18.5483	0.014	238.626
Sec	4.900	26.0000	127.400	-5.4869	9.804	147.521
TOT	6.277		128.763		9.823	673.757
Centroid of Section			20.5131	Moment of Inertia		16.56

Summary of Small End Inertia Calculations

Available Moment of Inertia (Small End)	0.1656E+02	cm**4
Required Moment of Inertia (Small End)	0.2666E+00	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

Results for Discontinuity Stresses per Bednar p. 236 2nd Edition

Stress Type	Stress	Allowable	Location
Tensile Stress	7.00	408.20	Small Cyl. Long.
Compres. Stress	-2.26	-408.20	Small Cyl. Long.
Membrane Stress	7.29	-204.10	Small End Tang.
Tensile Stress	7.04	408.20	Cone Longitudinal
Compres. Stress	-2.23	-408.20	Cone Longitudinal
Compres Stress	7.36	-204.10	Cone Tangential
Tensile Stress	28.73	408.20	Large Cyl. Long.
Compres. Stress	-15.32	-408.20	Large Cyl. Long.
Membrane Stress	1.29	-204.10	Large End Tang.
Tensile Stress	28.83	408.20	Cone Longitudinal
Compres. Stress	-15.22	-408.20	Cone Longitudinal
Compres Stress	1.49	-204.10	Cone Tangential

Note: An asterisk (*) denotes that this stress was not applicable for this combination of loads.

Maximum Allowable Pressure Calculations for Cone to Shell Junction:

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	4.998	Thickness due to internal pressure, Cone Large End
MAPnc	5.065	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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Center of Gravity Calculation: Step: 12 4:19pm Feb 20,2026

Shop/Field Installation Options :

Note : The CG is computed from the first Element From Node

Center of Gravity of Stiffening Rings	4055.186 mm.
Center of Gravity of Bare Shell New and Cold	4759.567 mm.
Center of Gravity of Bare Shell Corroded	4759.567 mm.
Vessel CG in the Operating Condition	4684.575 mm.
Vessel CG in the Fabricated (Shop/Empty) Condition	4684.575 mm.

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MDMT Summary:

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Minimum Design Metal Temperature Results Summary :

Description	Curve	Basic MDMT °C	Reduced MDMT °C	UG-20 (f) MDMT °C	Thickness ratio	Gov Thk mm.	E*	PWHT reqd
	Notes							
	[16]	-196						
	[16]	-196						
	[16]	-196						
	[16]	-196						
	[16]	-196						
	[16]	-196						
	[16]	-196						
	[16]	-196						
	[16]	-196						
	[16]	-196						
Warmest MDMT:		-196						

Required Minimum Design Metal Temperature -28.9 °C

Warmest Computed Minimum Design Metal Temperature -196.0 °C

Notes:

- [!] - This was an impact tested material.
- [1] - Governing Nozzle Weld.
- [4] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-c).
- [5] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-b).
- [6] - MDMT Calculations at the Shell/Head Joint.
- [7] - MDMT Calculations for the Straight Flange.
- [8] - Cylinder/Cone/Flange Junction MDMT.
- [9] - Calculations in the Spherical Portion of the Head.
- [10] - Calculations in the Knuckle Portion of the Head.
- [11] - Calculated (Body Flange) Flange MDMT.
- [12] - Calculated Flat Head MDMT per UCS-66.3
- [13] - Tubesheet MDMT, shell side, if applicable
- [14] - Tubesheet MDMT, tube side, if applicable
- [15] - Nozzle Material
- [16] - Shell or Head Material
- [17] - Impact Testing required
- [18] - Impact Testing not required, see UCS-66(b)(3)
- [19] - Select a valid hydrotest type to get the UG-20(f) exemption
- [20] - Cylinder/Cone Junction MDMT based on Longitudinal Stress considerations
- [21] - Bolting Material

UG-84(b)(2) was not considered.

UCS-66(g) was not considered.

UCS-66(i) was not considered.

Notes:

Impact test temps were not entered in and not considered in the analysis.
 UCS-66(i) applies to impact tested materials not by specification and
 UCS-66(g) applies to materials impact tested per UG-84.1 General Note (c).
 The Basic MDMT includes the (30F) PWHT credit if applicable.

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Vessel Design Summary:

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ASME Code, Section VIII Division 1, 2017

Diameter Spec : 321.000 x 1124.000 x 566.000 x 1606.000 x 898.000 mm. ID
 Vessel Design Length, Tangent to Tangent 7474.00 mm.

Distance of Bottom Tangent above Grade 50.00 mm.
 Distance of Base above Grade 0.00 mm.
 Specified Datum Line Distance 0.00 mm.

Internal Design Temperature 170 °C
 Internal Design Pressure 0.500 bars

External Design Temperature 50 °C
 External Design Pressure 0.100 bars

Maximum Allowable Working Pressure 1.653 bars
 External Max. Allowable Working Pressure 0.271 bars
 Hydrostatic Test Pressure 0.000 bars

Required Minimum Design Metal Temperature -28.9 °C
 Warmest Computed Minimum Design Metal Temperature -196.0 °C

Wind Design Code ASCE-93
 Earthquake Design Code UBC-94

Materials of Construction:

Component Type	Material	Class	Thickness	UNS #	Normal ized	Impact Tested
Shell	SA-240 316	S31600	No	No
Cone	SA-240 316	S31600	No	No
Cover	SA-240 316	S31600	No	No
Rings	SA-240 316	S31600	No	No
Rings	SA-240 304	S30400	No	No

Normalized is determined based on the UCS-66 material curve selection and Figure UCS-66.

Impact Tested is based on material selection and material data properties.

Element Pressures and MAWP (bars & mm.):

Element Description or Type	Design Pressure + Stat. head	Ext. Press.	Element M.A.W.P	Corrosion Allowance	Str. Flg. Gov.	In Creep Range
Cylinder	0.500	0.10	25.156	0.0000	N/A	No
Conical	0.500	0.10	6.804	0.0000	N/A	No
Cylinder	0.500	0.10	7.240	0.0000	N/A	No
Wld Flat	0.500	0.10	2.124	0.0000	N/A	No
Cylinder	0.500	0.10	14.332	0.0000	N/A	No
Conical	0.500	0.10	4.998	0.0000	N/A	No
Cylinder	0.500	0.10	5.072	0.0000	N/A	No
Wld Flat	0.500	0.10	1.653	0.0000	N/A	No
Cylinder	0.500	0.10	9.055	0.0000	N/A	No

Stiffener Ring Specifications:

Elevation mm.	Selected Type	User Description
0.00	Bar 39.0 x 12.	Ring 1-1
1145.00	Bar 50.5 x 10.	Ring 2-1
1724.00	Bar 49.0 x 10.	Ring 5-1
1736.00	Bar 49.0 x 10.	Ring 6-1
3605.00	Bar 47.8 x 10.	Ring 6-2
4741.00	Bar 47.8 x 10.	Ring 6-3
4846.00	Bar 47.0 x 10.	Ring 7-1

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Vessel Design Summary:

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6356.00	Bar	47.0 x 10.	Ring 7-2
7469.00	Bar	26.0 x 14.	Ring 8.1

Element Types and Properties:

Element Type	"To" Elev mm.	Element Length mm.	Nominal Thickness mm.	Finished Thickness mm.	Reqd Thk Internal mm.	Reqd Thk External mm.	Long Eff	Circ Eff
Cylinder	50.0	50.0	3.0	3.0	1.5	...	1.00	1.00
Conical	1150.0	1100.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	1606.0	456.0	3.0	3.0	1.5	...	1.00	1.00
Wld Flat	1616.0	10.0	10.0	10.0	4.9	...	1.00	1.00
Cylinder	1731.0	115.0	3.0	3.0	1.5	...	1.00	1.00
Conical	4746.0	3015.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	7260.0	2514.0	3.0	3.0	1.5	...	1.00	1.00
Wld Flat	7274.0	14.0	14.0	14.0	7.7	...	1.00	1.00
Cylinder	7474.0	200.0	3.0	3.0	1.5	...	1.00	1.00

External Pressure Calculations:

From	To	External Actual T. mm.	External Required T. mm.	External Design Pressure bars	External M.A.W.P. bars
10	Ring	3	No Calc	0.1	No Calc
Ring	20	3	0.19212	0.1	12.103
20	Ring	3	1.27187	0.1	0.85662
Ring	30	3	0.43542	0.1	3.08702
30	40	3	0.99796	0.1	1.60685
40	50	10	No Calc	0.1	No Calc
50	Ring	3	0.36718	0.1	6.05541
Ring	60	3	0.20698	0.1	7.27422
60	Ring	3	0.2107	0.1	7.13086
Ring	Ring	3	1.66204	0.1	0.43479
Ring	Ring	3	1.72473	0.1	0.40042
Ring	70	3	0.59194	0.1	2.11938
70	Ring	3	0.64382	0.1	2.0579
Ring	Ring	3	2.02088	0.1	0.27114
Ring	80	3	1.63721	0.1	0.45972
80	90	14	No Calc	0.1	No Calc
90	Ring	3	0.61369	0.1	3.26313
Ring	100	3	0.32711	0.1	4.3309

External Pressure Calculations:

From	To	Actual Length Bet. Stiffeners mm.	Allowable Length Bet. Stiffeners mm.	Ring Inertia Required cm**4	Ring Inertia Available cm**4
10	Ring	No Calc	No Calc	No Calc	No Calc
Ring	20	50	4811.25	0.00062658	9.67321
20	Ring	706.449	706.449	No Calc	No Calc
Ring	30	4.99191	4.99191	0.36588	20.652
30	40	456	6566.1	No Calc	No Calc
40	50	No Calc	No Calc	No Calc	No Calc
50	Ring	108	4021.26	No Calc	No Calc
Ring	60	7	487.092	0.0077134	17.0396
60	Ring	4.99249	4.99249	No Calc	No Calc
Ring	Ring	1374.53	1374.53	0.093369	17.0481
Ring	Ring	997.779	997.779	1.53939	18.2173
Ring	70	4.99741	4.99741	1.5006	19.1386
70	Ring	100	2398.86	No Calc	No Calc
Ring	Ring	1510	3984.95	2.40723	18.3065
Ring	80	904	3985.66	3.60856	18.3065

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FileName : CYCLONE-Rev 01

Vessel Design Summary:

Step: 14 4:19pm Feb 20,2026

80	90	No Calc	No Calc	No Calc	No Calc
90	Ring	195	5498.92	No Calc	No Calc
Ring	100	5	216.544	0.052954	4.51783



Factored Loads:

Un-Factored Loads:



Weights:

Fabricated - Bare W/O Removable Internals	1106.2	kg.
Shop Test - Fabricated + Water (Full)	10310.6	kg.
Shipping - Fab. + Rem. Intls.+ Shipping App.	1106.2	kg.
Erected - Fab. + Rem. Intls.+ Insul. (etc)	1106.2	kg.
Empty - Fab. + Intls. + Details + Wghts.	1106.2	kg.
Operating - Empty + Operating Liquid (No CA)	1106.2	kg.
Field Test - Empty Weight + Water (Full)	10310.6	kg.

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 Innovation and Equipment	MODIFIED STARCH FLASH DRYER PROJECT					 A.G.F.D. Tândărei	
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.		Rev.
	5204	ME	CL	000	02		00

APPENDIX 3 **CALCULATION** **RECTANGULAR SECTION** **ZONE 2**

	MODIFIED STARCH FLASH DRYER PROJECT						
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.	Rev.	
	5204	ME	CL	000	02	00	

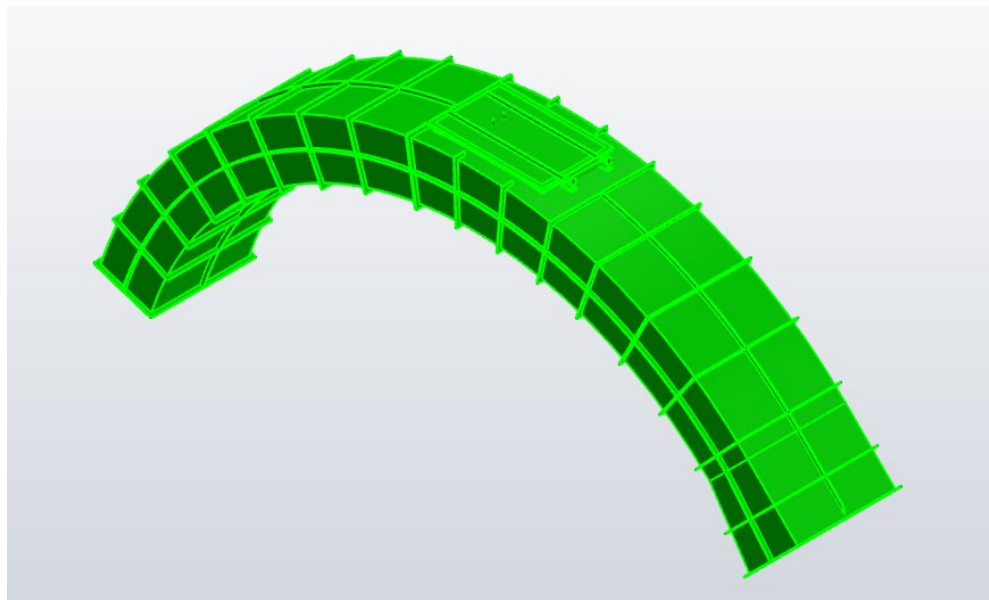
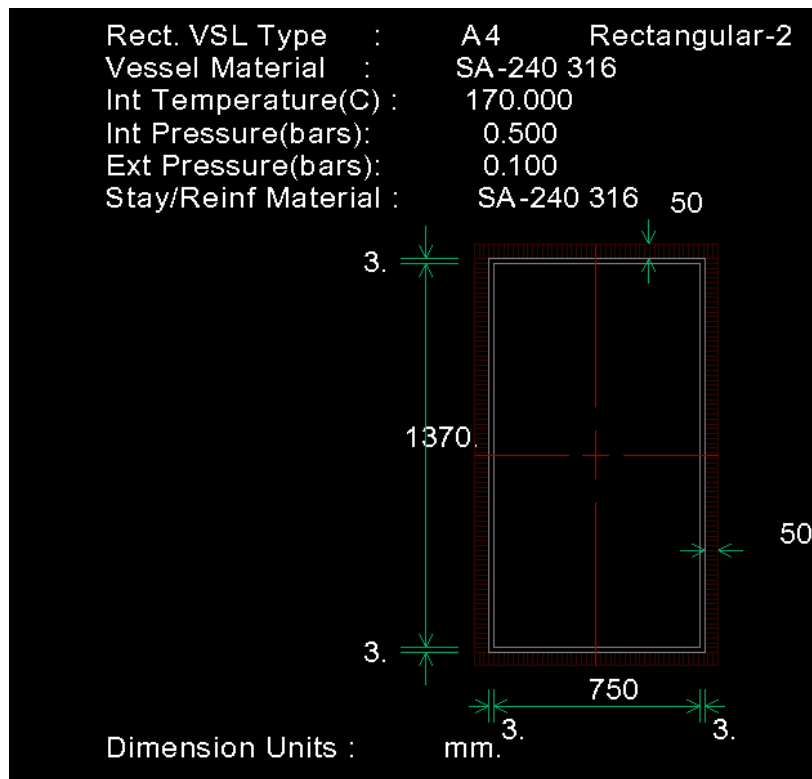


Figure C- Rectangular Duct - Zone 2

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MICROTEC
Rectangular Duct for Flash Dryer - Section 1

DESIGN CALCULATION

ASME Code Version : 2017

Analysis Performed by : SL Client

Job File : D:\2-PAYAM\1-Projects\Microtec Project\13-Microtec -Flash Dryer-5204\5204-Eng

Date of Analysis : Feb 20,2026

PV Elite 2019 , January 2019

Note:

PV Elite performs all calculations internally in U.S. Customary Units to remain compliant with the ASME Code and any built in assumptions in the ASME Code formulae. Finalized results are converted to the user set of selected units using conversion constants with adequate significant digits.

MICROTEC

Rectangular Duct for Flash Dryer - Section 1

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 2

Rectves Analysis : Rectangular-2 Item: 1 5:19p Feb 20,2026

Input Echo, COMPONENT 1, Description: Rectangular-2

Figure Number Analyzed A4

Design Internal Pressure P 0.5000 bars
 Design External Pressure Pext 0.1000 bars
 Design Temperature Temp 170.0000 C

VESSEL MATERIAL DATA:
 Material Specification SA-240 316
 Shell Allowable Stress at Design Temp S 136.0672 N./mm²
 Shell Allowable Stress at Ambient SA 137.9000 N./mm²
 Shell Yield Stress at Design Temperature Sy 156.1065 N./mm²

SHORT-SIDE VESSEL DATA:
 Short-side Length Dimension H 750.0000 mm.
 Minimum Thickness of Short-side Plates t1 3.0000 mm.
 Mid-side Joint Efficiency on Short-side E 1.0000
 Corner Joint Efficiency on Short-side EC 1.0000

LONG-SIDE VESSEL DATA:
 Long-side Length Dimension h 1370.0000 mm.
 Minimum Thickness of Long-side Plates t2 3.0000 mm.
 Mid-side Joint Efficiency on Long-side E 1.0000

REINFORCEMENT MATERIAL DATA:
 Reinforcement Material Specification SA-240 316
 Reinf Allowable Stress at Design Temp Sr 136.0672 N./mm²
 Reinf Allowable Stress at Ambient SA 137.9000 N./mm²
 Reinf Yield Stress at Design Temp Sy 156.1065 N./mm²

C-Factor for Reinforcement (from UG-47) 2.1000
 DELTA (Reinforcement Material Parameter) 485.0000 N./mm²^{0.5}

SHORT-SIDE RECTANGULAR BEAM DATA:
 Outside Distance from Outside of Vessel 50.0000 mm.
 Width of Reinforcing Member 10.0000 mm.

LONG-SIDE RECTANGULAR BEAM DATA:
 Outside Distance from Outside of Vessel 50.0000 mm.
 Width of Reinforcing Member 10.0000 mm.

Rectangular Vessel Results, Item number 1, Desc: Rectangular-2**ASME Code, Section VIII, Division 1, 2017 App. 13****Ligament Efficiency Calculations (Section 13-6, Equations (1)-(6)):****Short-side 1 Calculations**

Membrane Ligament Efficiency [Em]:
 = 1.000

Bending Ligament Efficiency [Eb]:
 = 1.000

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:
 = t1 - CA / 2
 = 3.000 - 0.000 / 2
 = 1.500 mm.

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 2

Rectves Analysis : Rectangular-2 Item: 1 5:19p Feb 20,2026

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Short-side 2 Calculations

Membrane Ligament Efficiency [Em]:

$$= 1.000$$

Bending Ligament Efficiency [Eb]:

$$= 1.000$$

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:

$$\begin{aligned}
 &= t_1 - CA / 2 \\
 &= 3.000 - 0.000 / 2 \\
 &= 1.500 \text{ mm.}
 \end{aligned}$$

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Long-side 1 Calculations

Membrane Ligament Efficiency [Em]:

$$= 1.000$$

Bending Ligament Efficiency [Eb]:

$$= 1.000$$

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:

$$\begin{aligned}
 &= t_1 - CA / 2 \\
 &= 3.000 - 0.000 / 2 \\
 &= 1.500 \text{ mm.}
 \end{aligned}$$

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Long-side 2 Calculations

Membrane Ligament Efficiency [Em]:

$$= 1.000$$

Bending Ligament Efficiency [Eb]:

$$= 1.000$$

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:

$$\begin{aligned}
 &= t_1 - CA / 2 \\
 &= 3.000 - 0.000 / 2 \\
 &= 1.500 \text{ mm.}
 \end{aligned}$$

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Ligament Efficiency Calculations (Section 13-6, Equations (1)-(6)):

	Em	Eb	Ci	Co
Short-side 1	1.000	1.000	1.500	-1.500

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 2

Rectves Analysis : Rectangular-2

Item: 1 5:19p Feb 20,2026

	2	1.000	1.000	1.500	-1.500
Long-side	1	1.000	1.000	1.500	-1.500
	2	1.000	1.000	1.500	-1.500

Effective Width of Shell Plate (Section 13-8, Eq. (2))

In Compression [w]:

$$= \text{Min}(\text{Min}(t_1, t_2) * \Delta / \sqrt{S_y}, p)$$

$$= \text{Min}(\text{Min}(3.00, 3.00) * 5840.83 / \sqrt{156.11}, 65.00)$$

$$= 65.00 \text{ mm.}$$

In Tension [w]:

$$= 65.00 \text{ mm.}$$

Moment of Inertia of a Strip of the Vessel Wall:

$$\text{Thickness } t_1, I_1 = 0.0000 \text{ cm}^4$$

$$\text{Thickness } t_2, I_2 = 0.0000 \text{ cm}^4$$

Effective Area of Reinforcement on Shell (t * w):

$$\text{Short-side } A_p = 1.9500 \text{ cm}^2$$

$$\text{Long-side } A_p = 1.9500 \text{ cm}^2$$

Moment of Inertia of Effective Area of Reinforcement (w * t3 / 12):**

$$\text{Short-side } I_s = 0.0146 \text{ cm}^4$$

$$\text{Long-side } I_l = 0.0146 \text{ cm}^4$$

Moment of Inertia of Combined Reinforcement and Effective Width:

$$\text{In Compression } I_{11} = 20.2830 \text{ cm}^4$$

$$I_{21} = 20.2830 \text{ cm}^4$$

$$\text{In Tension } I_{11} = 20.2830 \text{ cm}^4$$

$$I_{21} = 20.2830 \text{ cm}^4$$

Distance from Neutral Axis of Cross Section of Composite Section to the Inside Surface of the Vessel (mm.):

		Ci	Co
Short-side, in Compression		20.5647	-32.4353
	in Tension	20.5647	-32.4353
Long-side, in Compression		20.5647	-32.4353
	in Tension	20.5647	-32.4353

Rectangular Vessel Reinforcement Parameters:

$$\text{Alpha1} = H_1 / h_1 = 0.5652$$

$$k(\text{comp}) = (I_{22}/I_{11}) * \text{Alpha1} = 0.5652$$

$$k(\text{tens}) = (I_{22}/I_{11}) * \text{Alpha1} = 0.5652$$

Membrane Stress Calculations per Section 13-8**Membrane Stresses at Short-side 1**

Membrane Stress at Short-side 1 [Sms]:

$$= p * h * p / (2 * (A_1 + p * t_1))$$

$$= 0.50 * 1370.00 * 65.00 / (2 * (500.000 + 65.00 * 3.00))$$

$$= 3.20 \text{ N./mm}^2$$

Membrane Stresses at Short-side 2

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

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FileName : Rectangular Section - 2 -----

Rectves Analysis : Rectangular-2 Item: 1 5:19p Feb 20,2026

Membrane Stress at Short-side 2 [Sms]:

$$\begin{aligned}
 &= p * h * p / (2 * (A1 + p * t1)) \\
 &= 0.50 * 1370.00 * 65.00 / (2 * (500.000 + 65.00 * 3.00)) \\
 &= 3.20 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stresses at Long-side 1**Membrane Stress at Long-side 1 at A[SmlA]:**

$$\begin{aligned}
 &= p * H * p / (2 * (A2 + p * t2)) \\
 &= 0.50 * 750.00 * 65.00 / (2 * (5.000 + 65.00 * 3.00)) \\
 &= 1.75 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stresses at Long-side 2**Membrane Stress at Long-side 2 at A[SmlA]:**

$$\begin{aligned}
 &= p * H * p / (2 * (A2 + p * t2)) \\
 &= 0.50 * 750.00 * 65.00 / (2 * (5.000 + 65.00 * 3.00)) \\
 &= 1.75 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stresses at Corner sections**Membrane Stress at Short side [Smsc]:**

$$\begin{aligned}
 &= p * h * p / (2 * (A1 + p * t1)) \\
 &= 0.50 * 1370.00 * 65.00 / (2 * (5.000 + 65.00 * 3.00)) \\
 &= 3.20 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stress at Long side [Smlc]:

$$\begin{aligned}
 &= p * H * p / (2 * (A2 + p * t2)) \\
 &= 0.50 * 750.00 * 65.00 / (2 * (5.000 + 65.00 * 3.00)) \\
 &= 1.75 \text{ N./mm}^2
 \end{aligned}$$

MEMBRANE STRESSES: Membrane Stress Calculations per Section 13-8, Equations (3) and (4). (N./mm²) :

STRESS LOCATIONS	Actual	Allowable
Short-side 1	3.20	136.07
Short-side 2	3.20	136.07
Short-side Corner	3.20	136.07
Long-side 1 at A	1.75	136.07
Long-side 2 at A	1.75	136.07
Long-side Corner	1.75	136.07

Bending Stress Calculations per Section 13-8**Bending Stresses at Short-side 1****Bending Stress at Short-side 1 at N Inner[SbsNi]:**

$$\begin{aligned}
 &= P * p * c / (24 * I11) * [-3 * H^2 + 2 * h^2 \\
 &\quad * ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.5 * 65.0 * 20.56 / (24 * 20.3) * [-3 * 750.00^2 + 2 * 1370.00^2 * \\
 &\quad ((1 + 0.57^2 * 0.57) / (1 + 0.57))] \\
 &= 15.71 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 1 at N Outer[SbsNo]:

$$\begin{aligned}
 &= P * p * c / (24 * I11) * [-3 * H^2 + 2 * h^2 \\
 &\quad * ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.5 * 65.0 * -32.44 / (24 * 20.3) * [-3 * 750.00^2 + 2 * 1370.00^2 * \\
 &\quad ((1 + 0.57^2 * 0.57) / (1 + 0.57))] \\
 &= -24.77 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 1 at Q Inner[SbsQi]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I11) * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k))
 \end{aligned}$$

MICROTEC

Rectangular Duct for Flash Dryer - Section 1

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 2

Rectves Analysis : Rectangular-2

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$$= 0.50 * 1370.00^2 * 65.00 * 20.56 / (12 * 20.28) * ((1 + 0.57^2 * 0.57) / (1 + 0.57))$$

$$= 38.88 \text{ N./mm}^2$$

Bending Stress at Short-side 1 at Q Outer[SbsQo]:

$$= P * h^2 * p * c / (12 * I_{11}) * ((1 + \text{Alpha}1^2 * k) / (1 + k))$$

$$= 0.50 * 1370.00^2 * 65.00 * -32.44 / (12 * 20.28) * ((1 + 0.57^2 * 0.57) / (1 + 0.57))$$

$$= -61.32 \text{ N./mm}^2$$

Bending Stresses at Short-side 2

Bending Stress at Short-side 2 at N Inner[SbsNi]:

$$= P * p * c / (24 * I_{11}) * [-3 * H^2 + 2 * h^2 * ((1 + \text{Alpha}1^2 * k) / (1 + k))]$$

$$= 0.5 * 65.0 * 20.56 / (24 * 20.3) * [-3 * 750.00^2 + 2 * 1370.00^2 * ((1 + 0.57^2 * 0.57) / (1 + 0.57))]$$

$$= 15.71 \text{ N./mm}^2$$

Bending Stress at Short-side 2 at N Outer[SbsNo]:

$$= P * p * c / (24 * I_{11}) * [-3 * H^2 + 2 * h^2 * ((1 + \text{Alpha}1^2 * k) / (1 + k))]$$

$$= 0.5 * 65.0 * -32.44 / (24 * 20.3) * [-3 * 750.00^2 + 2 * 1370.00^2 * ((1 + 0.57^2 * 0.57) / (1 + 0.57))]$$

$$= -24.77 \text{ N./mm}^2$$

Bending Stress at Short-side 2 at Q Inner[SbsQi]:

$$= P * h^2 * p * c / (12 * I_{11}) * ((1 + \text{Alpha}1^2 * k) / (1 + k))$$

$$= 0.50 * 1370.00^2 * 65.00 * 20.56 / (12 * 20.28) * ((1 + 0.57^2 * 0.57) / (1 + 0.57))$$

$$= 38.88 \text{ N./mm}^2$$

Bending Stress at Short-side 2 at Q Outer[SbsQo]:

$$= P * h^2 * p * c / (12 * I_{11}) * ((1 + \text{Alpha}1^2 * k) / (1 + k))$$

$$= 0.50 * 1370.00^2 * 65.00 * -32.44 / (12 * 20.28) * ((1 + 0.57^2 * 0.57) / (1 + 0.57))$$

$$= -61.32 \text{ N./mm}^2$$

Bending Stresses at Long-side 1

Bending Stress at Long-side 1 at M Inner[SblMi]:

$$= P * h^2 * p * c / (24 * I_{21}) * [-3 + 2 * ((1 + \text{Alpha}1^2 * k) / (1 + k))]$$

$$= 0.50 * 1370.00^2 * 65.00 * 20.56 / (24 * 20.28) * [-3 + 2 * ((1 + 0.57^2 * 0.57) / (1 + 0.57))]$$

$$= -38.44 \text{ N./mm}^2$$

Bending Stress at Long-side 1 at M Outer[SblMo]:

$$= P * h^2 * p * c / (24 * I_{21}) * [-3 + 2 * ((1 + \text{Alpha}1^2 * k) / (1 + k))]$$

$$= 0.50 * 1370.00^2 * 65.00 * -32.44 / (24 * 20.28) * [-3 + 2 * ((1 + 0.57^2 * 0.57) / (1 + 0.57))]$$

$$= 60.62 \text{ N./mm}^2$$

Bending Stress at Long-side 1 at Q Inner[SblQi]:

$$= P * h^2 * p * c / (12 * I_{21}) * ((1 + \text{Alpha}1^2 * k) / (1 + k))$$

$$= 0.50 * 1370.00^2 * 65.00 * 20.56 / (12 * 20.28) * ((1 + 0.57^2 * 0.57) / (1 + 0.57))$$

$$= 38.88 \text{ N./mm}^2$$

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 2 -----

Rectves Analysis : Rectangular-2 Item: 1 5:19p Feb 20,2026

Bending Stress at Long-side 1 at Q Outer[SblQo]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I21) * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k)) \\
 &= 0.50 * 1370.00^2 * 65.00 * -32.44 / (24 * 20.28) * \\
 &\quad ((1 + 0.57^2 * 0.57) / (1 + 0.57))] \\
 &= -61.32 \text{ N./mm}^2
 \end{aligned}$$

Bending Stresses at Long-side 2**Bending Stress at Long-side 2 at M Inner[SblMi]:**

$$\begin{aligned}
 &= P * h^2 * p * c / (24 * I21) * [-3 + 2 * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.50 * 1370.00^2 * 65.00 * 20.56 / (24 * 20.28) * [-3 + 2 * \\
 &\quad ((1 + 0.57^2 * 0.57) / (1 + 0.57))] \\
 &= -38.44 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Long-side 2 at M Outer[SblMo]:

$$\begin{aligned}
 &= P * h^2 * p * c / (24 * I21) * [-3 + 2 * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.50 * 1370.00^2 * 65.00 * -32.44 / (24 * 20.28) * [-3 + 2 * \\
 &\quad ((1 + 0.57^2 * 0.57) / (1 + 0.57))] \\
 &= 60.62 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Long-side 2 at Q Inner[SblQi]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I21) * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k)) \\
 &= 0.50 * 1370.00^2 * 65.00 * 20.56 / (24 * 20.28) * \\
 &\quad ((1 + 0.57^2 * 0.57) / (1 + 0.57))] \\
 &= 38.88 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Long-side 2 at Q Outer[SblQo]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I21) * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k)) \\
 &= 0.50 * 1370.00^2 * 65.00 * -32.44 / (24 * 20.28) * \\
 &\quad ((1 + 0.57^2 * 0.57) / (1 + 0.57))] \\
 &= -61.32 \text{ N./mm}^2
 \end{aligned}$$

BENDING STRESSES: Bending Stress Calculations per Section 13-8, Equations (5-8). (N./mm²) :

STRESS LOCATIONS	Inner	Outer	Allowable
Short-side 1 at N	15.71	-24.77	104.07
at Q	38.88	-61.32	104.07
Short-side 2 at N	15.71	-24.77	104.07
at Q	38.88	-61.32	104.07
Long-side 1 at M	-38.44	60.62	104.07
at Q	38.88	-61.32	104.07
Long-side 2 at M	-38.44	60.62	104.07
at Q	38.88	-61.32	104.07

Total Stress Calculations per Section 13-8**Total Stresses at Short-side 1****Total Stress at short side 1 at N inner [STS_Ni]:**

$$\begin{aligned}
 &= Sms + SbsNi \\
 &= 3.20 + 15.71 \\
 &= 18.91 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 1 at N outer [STS_No]:

$$\begin{aligned}
 &= Sms + SbsNo \\
 &= 3.20 + -24.77
 \end{aligned}$$

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 2

Rectves Analysis : Rectangular-2

Item: 1 5:19p Feb 20,2026

$$= -21.57 \text{ N./mm}^2$$

Total Stress at short side 1 at Q inner [STS_Qi]:

$$\begin{aligned} &= S_{msc} + S_{bsQi} \\ &= 3.20 + 38.88 \\ &= 42.08 \text{ N./mm}^2 \end{aligned}$$

Total Stress at short side 1 at Q outer [STS_Qo]:

$$\begin{aligned} &= S_{msc} + S_{bsQo} \\ &= 3.20 + -61.32 \\ &= -58.11 \text{ N./mm}^2 \end{aligned}$$

Total Stresses at Short-side 2

Total Stress at short side 2 at N inner [STS_Ni]:

$$\begin{aligned} &= S_{ms} + S_{bsNi} \\ &= 3.20 + 15.71 \\ &= 18.91 \text{ N./mm}^2 \end{aligned}$$

Total Stress at short side 2 at N outer [STS_No]:

$$\begin{aligned} &= S_{ms} + S_{bsNo} \\ &= 3.20 + -24.77 \\ &= -21.57 \text{ N./mm}^2 \end{aligned}$$

Total Stress at short side 2 at Q inner [STS_Qi]:

$$\begin{aligned} &= S_{msc} + S_{bsQi} \\ &= 3.20 + 38.88 \\ &= 42.08 \text{ N./mm}^2 \end{aligned}$$

Total Stress at short side 2 at Q outer [STS_Qo]:

$$\begin{aligned} &= S_{msc} + S_{bsQo} \\ &= 3.20 + -61.32 \\ &= -58.11 \text{ N./mm}^2 \end{aligned}$$

Total Stresses at Long-side 1

Total Stress at long side 1 at M inner [STL_Mi]:

$$\begin{aligned} &= S_{mlA} + S_{blMi} \\ &= 1.75 + -38.44 \\ &= -36.68 \text{ N./mm}^2 \end{aligned}$$

Total Stress at long side 1 at M outer [STL_Mo]:

$$\begin{aligned} &= S_{mlA} + S_{blMo} \\ &= 1.75 + 60.62 \\ &= 62.38 \text{ N./mm}^2 \end{aligned}$$

Total Stress at long side 1 at Q inner [STL_Qi]:

$$\begin{aligned} &= S_{mlc} + S_{blQi} \\ &= 1.75 + 38.88 \\ &= 40.63 \text{ N./mm}^2 \end{aligned}$$

Total Stress at long side 1 at Q outer [STL_Qo]:

$$\begin{aligned} &= S_{mlc} + S_{blQo} \\ &= 1.75 + -61.32 \\ &= -59.56 \text{ N./mm}^2 \end{aligned}$$

Total Stresses at Long-side 2

Total Stress at long side 2 at M inner [STL_Mi]:

$$\begin{aligned} &= S_{mlA} + S_{blMi} \\ &= 1.75 + -38.44 \\ &= -36.68 \text{ N./mm}^2 \end{aligned}$$

MICROTEC**Rectangular Duct for Flash Dryer - Section 1****PV Elite 2019 SP1 Licensee: SL Client****FileName : Rectangular Section - 2****Rectves Analysis : Rectangular-2****Item: 1 5:19p Feb 20,2026****Total Stress at long side 2 at M outer [STL_Mo]:**

$$\begin{aligned}
 &= SmlA + SblMo \\
 &= 1.75 + 60.62 \\
 &= 62.38 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 2 at Q inner [STL_Qi]:

$$\begin{aligned}
 &= Smlc + SblQi \\
 &= 1.75 + 38.88 \\
 &= 40.63 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 2 at Q outer [STL_Qo]:

$$\begin{aligned}
 &= Smlc + SblQo \\
 &= 1.75 + -61.32 \\
 &= -59.56 \text{ N./mm}^2
 \end{aligned}$$

**TOTAL STRESSES: Total Stress Calculations per Section 13-8,
Equations (9-12). (N./mm²) :**

STRESS LOCATIONS	Inner	Outer	Allowable
Short-side 1 at N	18.91	-21.57	104.07
at Q	42.08	-58.11	104.07
Short-side 2 at N	18.91	-21.57	104.07
at Q	42.08	-58.11	104.07
Long-side 1 at M	-36.68	62.38	104.07
at Q	40.63	-59.56	104.07
Long-side 2 at M	-36.68	62.38	104.07
at Q	40.63	-59.56	104.07

SUMMARY OF RESULTS:**MEMBRANE STRESS SUMMARY,**

High Stress (Highest % of Allowable)	3.20	N./mm ²
High Stress Percentage	2.35	%
M.A.W.P. for Membrane Stresses	21.24	bars

BENDING STRESS SUMMARY,

High Stress (Highest % of Allowable)	-61.32	N./mm ²
High Stress Percentage	58.92	%
M.A.W.P. for Bending Stresses	0.85	bars

TOTAL STRESS SUMMARY,

High Stress (Highest % of Allowable)	62.38	N./mm ²
High Stress Percentage	59.94	%
M.A.W.P. for Total Stresses	0.83	bars

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Rectangular Duct for Flash Dryer - Section 1

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 2

Vessel Results Summary

Item: 1 5:19p Feb 20,2026

Rectangular Vessel Results For Item 1 : A4**SUMMARY OF RESULTS:****MEMBRANE STRESS SUMMARY,**

High Stress (Highest % of Allowable)	3.20	N./mm ²
High Stress Percentage	2.35	%
M.A.W.P. for Membrane Stresses	21.24	bars



BENDING STRESS SUMMARY,

High Stress (Highest % of Allowable)	-61.32	N./mm ²
High Stress Percentage	58.92	%
M.A.W.P. for Bending Stresses	0.85	bars



TOTAL STRESS SUMMARY,

High Stress (Highest % of Allowable)	62.38	N./mm ²
High Stress Percentage	59.94	%
M.A.W.P. for Total Stresses	0.83	bars

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	MODIFIED STARCH FLASH DRYER PROJECT						
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.		Rev.
	5204	ME	CL	000	02		00

APPENDIX 4 **CALCULATION** **RECTANGULAR SECTION** **ZONE 3**

	MODIFIED STARCH FLASH DRYER PROJECT						
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.		Rev.
	5204	ME	CL	000	02		00

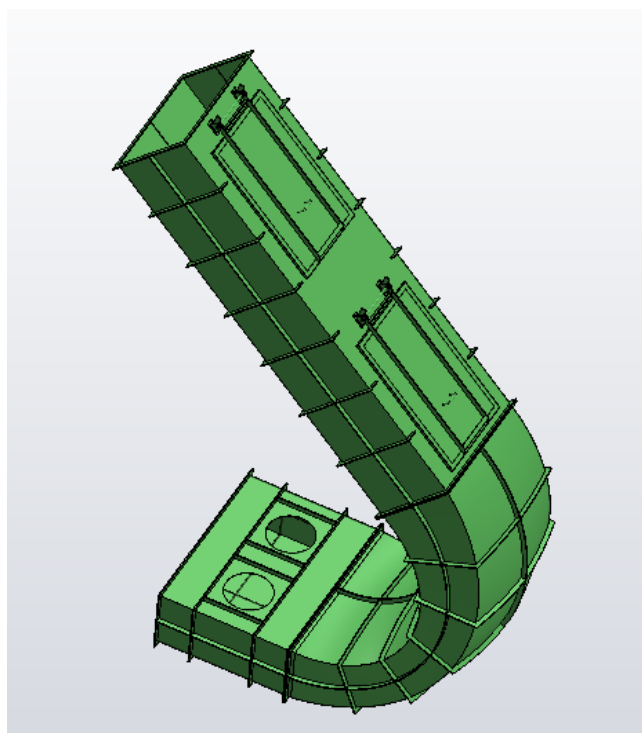
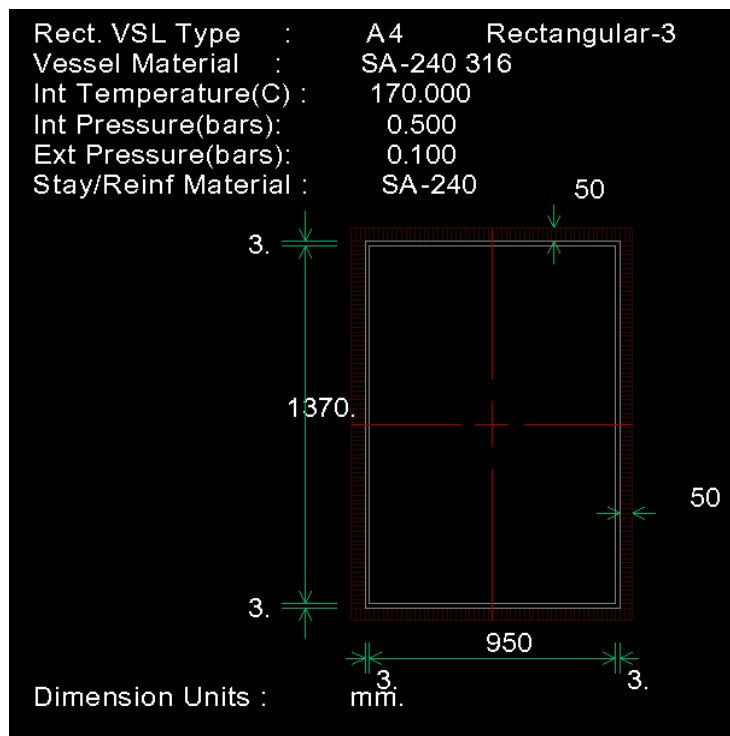


Figure D-Rectangular Duct - Zone 3

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MICROTEC
Rectangular Duct for Flash Dryer - Section 1

DESIGN CALCULATION

ASME Code Version : 2017

Analysis Performed by : SL Client

Job File : D:\2-PAYAM\1-Projects\Microtec Project\13-Microtec -Flash Dryer-5204\5204-Eng

Date of Analysis : Feb 20,2026

PV Elite 2019 , January 2019

Note:

PV Elite performs all calculations internally in U.S. Customary Units to remain compliant with the ASME Code and any built in assumptions in the ASME Code formulae. Finalized results are converted to the user set of selected units using conversion constants with adequate significant digits.

MICROTEC

Rectangular Duct for Flash Dryer - Section 1

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 3

Rectves Analysis : Rectangular-3 Item: 1 5:31p Feb 20,2026

Input Echo, COMPONENT 1, Description: Rectangular-3

Figure Number Analyzed A4

Design Internal Pressure P 0.5000 bars
 Design External Pressure Pext 0.1000 bars
 Design Temperature Temp 170.0000 C

VESSEL MATERIAL DATA:
 Material Specification SA-240 316
 Shell Allowable Stress at Design Temp S 136.0672 N./mm²
 Shell Allowable Stress at Ambient SA 137.9000 N./mm²
 Shell Yield Stress at Design Temperature Sy 156.1065 N./mm²

SHORT-SIDE VESSEL DATA:
 Short-side Length Dimension H 950.0001 mm.
 Minimum Thickness of Short-side Plates t1 3.0000 mm.
 Mid-side Joint Efficiency on Short-side E 1.0000
 Corner Joint Efficiency on Short-side EC 1.0000

LONG-SIDE VESSEL DATA:
 Long-side Length Dimension h 1370.0000 mm.
 Minimum Thickness of Long-side Plates t2 3.0000 mm.
 Mid-side Joint Efficiency on Long-side E 1.0000

REINFORCEMENT MATERIAL DATA:
 Reinforcement Material Specification SA-240
 Reinf Allowable Stress at Design Temp Sr 120.0000 N./mm²
 Reinf Allowable Stress at Ambient SA 153.7585 N./mm²
 Reinf Yield Stress at Design Temp Sy 166.5450 N./mm²

C-Factor for Reinforcement (from UG-47) 2.1000
 DELTA (Reinforcement Material Parameter) 485.0000 N./mm²^{0.5}

SHORT-SIDE RECTANGULAR BEAM DATA:
 Outside Distance from Outside of Vessel 50.0000 mm.
 Width of Reinforcing Member 10.0000 mm.

LONG-SIDE RECTANGULAR BEAM DATA:
 Outside Distance from Outside of Vessel 50.0000 mm.
 Width of Reinforcing Member 10.0000 mm.

Rectangular Vessel Results, Item number 1, Desc: Rectangular-3**ASME Code, Section VIII, Division 1, 2017 App. 13****Ligament Efficiency Calculations (Section 13-6, Equations (1)-(6)):****Short-side 1 Calculations**

Membrane Ligament Efficiency [Em]:
 = 1.000

Bending Ligament Efficiency [Eb]:
 = 1.000

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:
 = t1 - CA / 2
 = 3.000 - 0.000 / 2
 = 1.500 mm.

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

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FileName : Rectangular Section - 3

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Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Short-side 2 Calculations

Membrane Ligament Efficiency [Em]:

$$= 1.000$$

Bending Ligament Efficiency [Eb]:

$$= 1.000$$

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:

$$\begin{aligned}
 &= t_1 - CA / 2 \\
 &= 3.000 - 0.000 / 2 \\
 &= 1.500 \text{ mm.}
 \end{aligned}$$

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Long-side 1 Calculations

Membrane Ligament Efficiency [Em]:

$$= 1.000$$

Bending Ligament Efficiency [Eb]:

$$= 1.000$$

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:

$$\begin{aligned}
 &= t_1 - CA / 2 \\
 &= 3.000 - 0.000 / 2 \\
 &= 1.500 \text{ mm.}
 \end{aligned}$$

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Long-side 2 Calculations

Membrane Ligament Efficiency [Em]:

$$= 1.000$$

Bending Ligament Efficiency [Eb]:

$$= 1.000$$

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:

$$\begin{aligned}
 &= t_1 - CA / 2 \\
 &= 3.000 - 0.000 / 2 \\
 &= 1.500 \text{ mm.}
 \end{aligned}$$

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

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Rectangular Duct for Flash Dryer - Section 1

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FileName : Rectangular Section - 3

Rectves Analysis : Rectangular-3 Item: 1 5:31p Feb 20,2026

Ligament Efficiency Calculations (Section 13-6, Equations (1)-(6)):

		Em	Eb	Ci	Co
Short-side	1	1.000	1.000	1.500	-1.500
	2	1.000	1.000	1.500	-1.500
Long-side	1	1.000	1.000	1.500	-1.500
	2	1.000	1.000	1.500	-1.500

Effective Width of Shell Plate (Section 13-8, Eq. (2))

In Compression [w]:

$$= \text{Min}(\text{Min}(t_1, t_2) * \Delta / \sqrt{S_y}, p)$$

$$= \text{Min}(\text{Min}(3.00, 3.00) * 5840.83 / \sqrt{156.11}, 65.00)$$

$$= 65.00 \text{ mm.}$$

In Tension [w]:

$$= 65.00 \text{ mm.}$$

Moment of Inertia of a Strip of the Vessel Wall:

$$\text{Thickness } t_1, I_1 = 0.0000 \text{ cm}^4$$

$$\text{Thickness } t_2, I_2 = 0.0000 \text{ cm}^4$$

Effective Area of Reinforcement on Shell (t * w):

$$\text{Short-side } A_p = 1.9500 \text{ cm}^2$$

$$\text{Long-side } A_p = 1.9500 \text{ cm}^2$$

Moment of Inertia of Effective Area of Reinforcement (w * t**3 / 12):

$$\text{Short-side } I_s = 0.0146 \text{ cm}^4$$

$$\text{Long-side } I_l = 0.0146 \text{ cm}^4$$

Moment of Inertia of Combined Reinforcement and Effective Width:

$$\text{In Compression } I_{11} = 20.2830 \text{ cm}^4$$

$$I_{21} = 20.2830 \text{ cm}^4$$

$$\text{In Tension } I_{11} = 20.2830 \text{ cm}^4$$

$$I_{21} = 20.2830 \text{ cm}^4$$

Distance from Neutral Axis of Cross Section of Composite Section to the Inside Surface of the Vessel (mm.):

		Ci	Co
Short-side,	in Compression	20.5647	-32.4353
	in Tension	20.5647	-32.4353
Long-side,	in Compression	20.5647	-32.4353
	in Tension	20.5647	-32.4353

Rectangular Vessel Reinforcement Parameters:

$$\text{Alpha1} = H_1 / h_1 = 0.7055$$

$$k(\text{comp}) = (I_{22}/I_{11}) * \text{Alpha1} = 0.7055$$

$$k(\text{tens}) = (I_{22}/I_{11}) * \text{Alpha1} = 0.7055$$

Membrane Stress Calculations per Section 13-8

Membrane Stresses at Short-side 1

Membrane Stress at Short-side 1 [Sms]:

$$= p * h * p / (2 * (A_1 + p * t_1))$$

$$= 0.50 * 1370.00 * 65.00 / (2 * (500.000 + 65.00 * 3.00))$$

$$= 3.20 \text{ N./mm}^2$$

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Rectangular Duct for Flash Dryer - Section 1

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Rectves Analysis : Rectangular-3 Item: 1 5:31p Feb 20,2026

Membrane Stresses at Short-side 2

Membrane Stress at Short-side 2 [Sms]:

$$\begin{aligned}
 &= p * h * p / (2 * (A1 + p * t1)) \\
 &= 0.50 * 1370.00 * 65.00 / (2 * (500.000 + 65.00 * 3.00)) \\
 &= 3.20 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stresses at Long-side 1

Membrane Stress at Long-side 1 at A[SmlA]:

$$\begin{aligned}
 &= p * H * p / (2 * (A2 + p * t2)) \\
 &= 0.50 * 950.00 * 65.00 / (2 * (5.000 + 65.00 * 3.00)) \\
 &= 2.22 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stresses at Long-side 2

Membrane Stress at Long-side 2 at A[SmlA]:

$$\begin{aligned}
 &= p * H * p / (2 * (A2 + p * t2)) \\
 &= 0.50 * 950.00 * 65.00 / (2 * (5.000 + 65.00 * 3.00)) \\
 &= 2.22 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stresses at Corner sections

Membrane Stress at Short side [Smsc]:

$$\begin{aligned}
 &= p * h * p / (2 * (A1 + p * t1)) \\
 &= 0.50 * 1370.00 * 65.00 / (2 * (5.000 + 65.00 * 3.00)) \\
 &= 3.20 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stress at Long side [Smlc]:

$$\begin{aligned}
 &= p * H * p / (2 * (A2 + p * t2)) \\
 &= 0.50 * 950.00 * 65.00 / (2 * (5.000 + 65.00 * 3.00)) \\
 &= 2.22 \text{ N./mm}^2
 \end{aligned}$$

MEMBRANE STRESSES: Membrane Stress Calculations per Section 13-8, Equations (3) and (4). (N./mm²) :

STRESS LOCATIONS	Actual	Allowable
Short-side 1	3.20	120.00
Short-side 2	3.20	120.00
Short-side Corner	3.20	120.00
Long-side 1 at A	2.22	120.00
Long-side 2 at A	2.22	120.00
Long-side Corner	2.22	120.00

Bending Stress Calculations per Section 13-8**Bending Stresses at Short-side 1**

Bending Stress at Short-side 1 at N Inner[SbsNi]:

$$\begin{aligned}
 &= P * p * c / (24 * I11) * [-3 * H^2 + 2 * h^2 \\
 &\quad * ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.5 * 65.0 * 20.56 / (24 * 20.3) * [-3 * 950.00^2 + 2 * 1370.00^2 * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71))] \\
 &= 3.66 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 1 at N Outer[SbsNo]:

$$\begin{aligned}
 &= P * p * c / (24 * I11) * [-3 * H^2 + 2 * h^2 \\
 &\quad * ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.5 * 65.0 * -32.44 / (24 * 20.3) * [-3 * 950.00^2 + 2 * 1370.00^2 * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71))] \\
 &= -5.77 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 1 at Q Inner[SbsQi]:

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$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I_{11}) * \\
 &\quad ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\
 &= 0.50 * 1370.00^2 * 65.00 * 20.56 / (12 * 20.28) * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71)) \\
 &= 40.83 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 1 at Q Outer[SbsQo]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I_{11}) * \\
 &\quad ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\
 &= 0.50 * 1370.00^2 * 65.00 * -32.44 / (12 * 20.28) * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71)) \\
 &= -64.40 \text{ N./mm}^2
 \end{aligned}$$

Bending Stresses at Short-side 2

Bending Stress at Short-side 2 at N Inner[SbsNi]:

$$\begin{aligned}
 &= P * p * c / (24 * I_{11}) * [-3 * H^2 + 2 * h^2 \\
 &\quad * ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\
 &= 0.5 * 65.0 * 20.56 / (24 * 20.3) * [-3 * 950.00^2 + 2 * 1370.00^2 * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71))] \\
 &= 3.66 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 2 at N Outer[SbsNo]:

$$\begin{aligned}
 &= P * p * c / (24 * I_{11}) * [-3 * H^2 + 2 * h^2 \\
 &\quad * ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\
 &= 0.5 * 65.0 * -32.44 / (24 * 20.3) * [-3 * 950.00^2 + 2 * 1370.00^2 * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71))] \\
 &= -5.77 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 2 at Q Inner[SbsQi]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I_{11}) * \\
 &\quad ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\
 &= 0.50 * 1370.00^2 * 65.00 * 20.56 / (12 * 20.28) * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71)) \\
 &= 40.83 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 2 at Q Outer[SbsQo]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I_{11}) * \\
 &\quad ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\
 &= 0.50 * 1370.00^2 * 65.00 * -32.44 / (12 * 20.28) * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71)) \\
 &= -64.40 \text{ N./mm}^2
 \end{aligned}$$

Bending Stresses at Long-side 1

Bending Stress at Long-side 1 at M Inner[SblMi]:

$$\begin{aligned}
 &= P * h^2 * p * c / (24 * I_{21}) * [-3 + 2 * \\
 &\quad ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\
 &= 0.50 * 1370.00^2 * 65.00 * 20.56 / (24 * 20.28) * [-3 + 2 * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71))] \\
 &= -36.48 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Long-side 1 at M Outer[SblMo]:

$$\begin{aligned}
 &= P * h^2 * p * c / (24 * I_{21}) * [-3 + 2 * \\
 &\quad ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\
 &= 0.50 * 1370.00^2 * 65.00 * -32.44 / (24 * 20.28) * [-3 + 2 * \\
 &\quad ((1 + 0.71^2 * 0.71) / (1 + 0.71))] \\
 &= 57.54 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Long-side 1 at Q Inner[SblQi]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I_{21}) * \\
 &\quad ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\
 &= 0.50 * 1370.00^2 * 65.00 * 20.56 / (24 * 20.28) *
 \end{aligned}$$

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$$((1 + 0.71^2 * 0.71) / (1 + 0.71))) \\ = 40.83 \text{ N./mm}^2$$

Bending Stress at Long-side 1 at Q Outer[SblQo]:

$$= P * h^2 * p * c / (12 * I21) * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\ = 0.50 * 1370.00^2 * 65.00 * -32.44 / (24 * 20.28) * \\ ((1 + 0.71^2 * 0.71) / (1 + 0.71))) \\ = -64.40 \text{ N./mm}^2$$

Bending Stresses at Long-side 2

Bending Stress at Long-side 2 at M Inner[SblMi]:

$$= P * h^2 * p * c / (24 * I21) * [-3 + 2 * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\ = 0.50 * 1370.00^2 * 65.00 * 20.56 / (24 * 20.28) * [-3 + 2 * \\ ((1 + 0.71^2 * 0.71) / (1 + 0.71))] \\ = -36.48 \text{ N./mm}^2$$

Bending Stress at Long-side 2 at M Outer[SblMo]:

$$= P * h^2 * p * c / (24 * I21) * [-3 + 2 * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\ = 0.50 * 1370.00^2 * 65.00 * -32.44 / (24 * 20.28) * [-3 + 2 * \\ ((1 + 0.71^2 * 0.71) / (1 + 0.71))] \\ = 57.54 \text{ N./mm}^2$$

Bending Stress at Long-side 2 at Q Inner[SblQi]:

$$= P * h^2 * p * c / (12 * I21) * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\ = 0.50 * 1370.00^2 * 65.00 * 20.56 / (24 * 20.28) * \\ ((1 + 0.71^2 * 0.71) / (1 + 0.71))) \\ = 40.83 \text{ N./mm}^2$$

Bending Stress at Long-side 2 at Q Outer[SblQo]:

$$= P * h^2 * p * c / (12 * I21) * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\ = 0.50 * 1370.00^2 * 65.00 * -32.44 / (24 * 20.28) * \\ ((1 + 0.71^2 * 0.71) / (1 + 0.71))) \\ = -64.40 \text{ N./mm}^2$$

BENDING STRESSES: Bending Stress Calculations per Section 13-8, Equations (5-8). (N./mm²) :

STRESS LOCATIONS	Inner	Outer	Allowable
Short-side 1 at N	3.66	-5.77	104.07
at Q	40.83	-64.40	104.07
Short-side 2 at N	3.66	-5.77	104.07
at Q	40.83	-64.40	104.07
Long-side 1 at M	-36.48	57.54	104.07
at Q	40.83	-64.40	104.07
Long-side 2 at M	-36.48	57.54	104.07
at Q	40.83	-64.40	104.07

Total Stress Calculations per Section 13-8**Total Stresses at Short-side 1**

Total Stress at short side 1 at N inner [STS_Ni]:

$$= S_{ms} + S_{bsNi} \\ = 3.20 + 3.66 \\ = 6.86 \text{ N./mm}^2$$

Total Stress at short side 1 at N outer [STS_No]:

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$$\begin{aligned}
 &= S_{ms} + S_{bsNo} \\
 &= 3.20 + -5.77 \\
 &= -2.56 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 1 at Q inner [STS_Qi]:

$$\begin{aligned}
 &= S_{msc} + S_{bsQi} \\
 &= 3.20 + 40.83 \\
 &= 44.04 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 1 at Q outer [STS_Qo]:

$$\begin{aligned}
 &= S_{msc} + S_{bsQo} \\
 &= 3.20 + -64.40 \\
 &= -61.20 \text{ N./mm}^2
 \end{aligned}$$

Total Stresses at Short-side 2

Total Stress at short side 2 at N inner [STS_Ni]:

$$\begin{aligned}
 &= S_{ms} + S_{bsNi} \\
 &= 3.20 + 3.66 \\
 &= 6.86 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 2 at N outer [STS_No]:

$$\begin{aligned}
 &= S_{ms} + S_{bsNo} \\
 &= 3.20 + -5.77 \\
 &= -2.56 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 2 at Q inner [STS_Qi]:

$$\begin{aligned}
 &= S_{msc} + S_{bsQi} \\
 &= 3.20 + 40.83 \\
 &= 44.04 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 2 at Q outer [STS_Qo]:

$$\begin{aligned}
 &= S_{msc} + S_{bsQo} \\
 &= 3.20 + -64.40 \\
 &= -61.20 \text{ N./mm}^2
 \end{aligned}$$

Total Stresses at Long-side 1

Total Stress at long side 1 at M inner [STL_Mi]:

$$\begin{aligned}
 &= S_{mlA} + S_{blMi} \\
 &= 2.22 + -36.48 \\
 &= -34.26 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 1 at M outer [STL_Mo]:

$$\begin{aligned}
 &= S_{mlA} + S_{blMo} \\
 &= 2.22 + 57.54 \\
 &= 59.76 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 1 at Q inner [STL_Qi]:

$$\begin{aligned}
 &= S_{mlc} + S_{blQi} \\
 &= 2.22 + 40.83 \\
 &= 43.05 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 1 at Q outer [STL_Qo]:

$$\begin{aligned}
 &= S_{mlc} + S_{blQo} \\
 &= 2.22 + -64.40 \\
 &= -62.18 \text{ N./mm}^2
 \end{aligned}$$

Total Stresses at Long-side 2

Total Stress at long side 2 at M inner [STL_Mi]:

$$\begin{aligned}
 &= S_{mlA} + S_{blMi} \\
 &= 2.22 + -36.48
 \end{aligned}$$

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$$= -34.26 \text{ N./mm}^2$$

Total Stress at long side 2 at M outer [STL_Mo]:

$$\begin{aligned} &= SmlA + SblMo \\ &= 2.22 + 57.54 \\ &= 59.76 \text{ N./mm}^2 \end{aligned}$$

Total Stress at long side 2 at Q inner [STL_Qi]:

$$\begin{aligned} &= Smlc + SblQi \\ &= 2.22 + 40.83 \\ &= 43.05 \text{ N./mm}^2 \end{aligned}$$

Total Stress at long side 2 at Q outer [STL_Qo]:

$$\begin{aligned} &= Smlc + SblQo \\ &= 2.22 + -64.40 \\ &= -62.18 \text{ N./mm}^2 \end{aligned}$$

TOTAL STRESSES: Total Stress Calculations per Section 13-8, Equations (9-12). (N./mm²) :

STRESS LOCATIONS	Inner	Outer	Allowable
Short-side 1 at N	6.86	-2.56	104.07
at Q	44.04	-61.20	104.07
Short-side 2 at N	6.86	-2.56	104.07
at Q	44.04	-61.20	104.07
Long-side 1 at M	-34.26	59.76	104.07
at Q	43.05	-62.18	104.07
Long-side 2 at M	-34.26	59.76	104.07
at Q	43.05	-62.18	104.07

Note: The following can be used for outer stress:

Short-side 1 at N, outer allowable	111.03 N./mm ²
Short-side 2 at N, outer allowable	111.03 N./mm ²
Long-side 1 at M, outer allowable	111.03 N./mm ²
Long-side 2 at M, outer allowable	111.03 N./mm ²
At Corner Q, outer allowable	111.03 N./mm ²

SUMMARY OF RESULTS:**MEMBRANE STRESS SUMMARY,**

High Stress (Highest % of Allowable)	3.20	N./mm ²
High Stress Percentage	2.67	%
M.A.W.P. for Membrane Stresses	18.73	bars

BENDING STRESS SUMMARY,

High Stress (Highest % of Allowable)	-64.40	N./mm ²
High Stress Percentage	61.88	%
M.A.W.P. for Bending Stresses	0.81	bars

TOTAL STRESS SUMMARY,

High Stress (Highest % of Allowable)	-62.18	N./mm ²
High Stress Percentage	59.75	%
M.A.W.P. for Total Stresses	0.84	bars

MICROTEC

Rectangular Duct for Flash Dryer - Section 1

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 3

Vessel Results Summary

Item: 1 5:31p Feb 20,2026

Rectangular Vessel Results For Item 1 : A4**SUMMARY OF RESULTS:****MEMBRANE STRESS SUMMARY,**

High Stress (Highest % of Allowable)	3.20	N./mm ²
High Stress Percentage	2.67	%
M.A.W.P. for Membrane Stresses	18.73	bars



BENDING STRESS SUMMARY,

High Stress (Highest % of Allowable)	-64.40	N./mm ²
High Stress Percentage	61.88	%
M.A.W.P. for Bending Stresses	0.81	bars



TOTAL STRESS SUMMARY,

High Stress (Highest % of Allowable)	-62.18	N./mm ²
High Stress Percentage	59.75	%
M.A.W.P. for Total Stresses	0.84	bars

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	MODIFIED STARCH FLASH DRYER PROJECT						
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.		Rev.
	5204	ME	CL	000	02		00

APPENDIX 5 **CALCULATION** **RECTANGULAR SECTION** **ZONE 6**

 Innovation and Equipment	MODIFIED STARCH FLASH DRYER PROJECT						
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.	Rev.	
	5204	ME	CL	000	02	00	

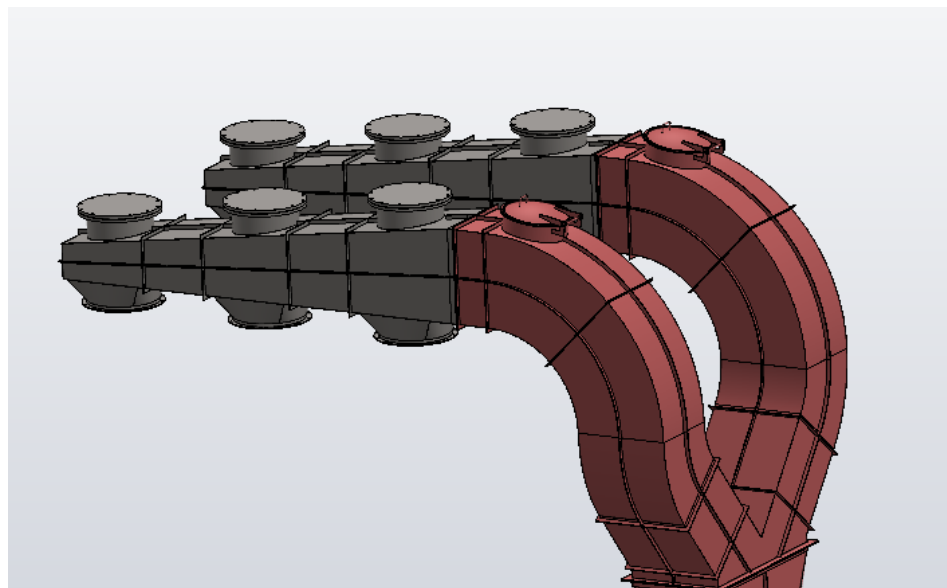
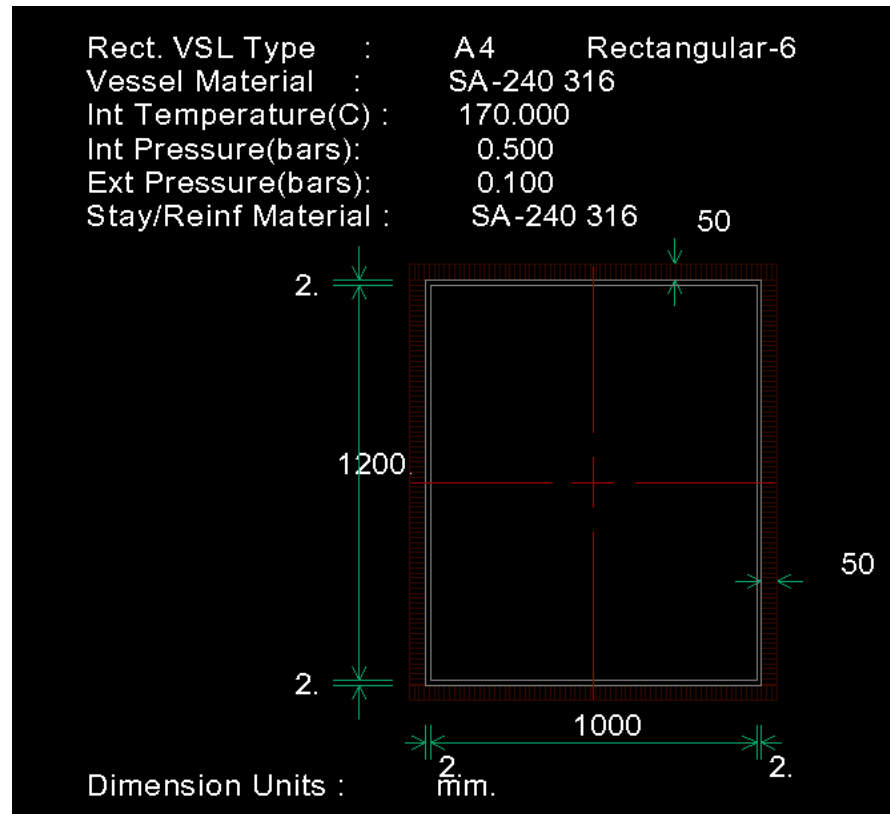


Figure E- Rectangular Duct - Zone 6

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MICROTEC
Rectangular Duct for Flash Dryer - Section 1

DESIGN CALCULATION

ASME Code Version : 2017

Analysis Performed by : SL Client

Job File : D:\2-PAYAM\1-Projects\Microtec Project\13-Microtec -Flash Dryer-5204\5204-Eng

Date of Analysis : Feb 20,2026

PV Elite 2019 , January 2019

Note:

PV Elite performs all calculations internally in U.S. Customary Units to remain compliant with the ASME Code and any built in assumptions in the ASME Code formulae. Finalized results are converted to the user set of selected units using conversion constants with adequate significant digits.

MICROTEC

Rectangular Duct for Flash Dryer - Section 1

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 1

Rectves Analysis : Rectangular-6 Item: 1 5:40p Feb 20,2026

Input Echo, COMPONENT 1, Description: Rectangular-6

Figure Number Analyzed A4

Design Internal Pressure P 0.5000 bars
 Design External Pressure Pext 0.1000 bars
 Design Temperature Temp 170.0000 C

VESSEL MATERIAL DATA:
 Material Specification SA-240 316
 Shell Allowable Stress at Design Temp S 136.0672 N./mm²
 Shell Allowable Stress at Ambient SA 137.9000 N./mm²
 Shell Yield Stress at Design Temperature Sy 156.1065 N./mm²

SHORT-SIDE VESSEL DATA:
 Short-side Length Dimension H 1000.0000 mm.
 Minimum Thickness of Short-side Plates t1 3.0000 mm.
 Mid-side Joint Efficiency on Short-side E 1.0000
 Corner Joint Efficiency on Short-side EC 1.0000

LONG-SIDE VESSEL DATA:
 Long-side Length Dimension h 1200.0000 mm.
 Minimum Thickness of Long-side Plates t2 3.0000 mm.
 Mid-side Joint Efficiency on Long-side E 1.0000

REINFORCEMENT MATERIAL DATA:
 Reinforcement Material Specification SA-240 316
 Reinf Allowable Stress at Design Temp Sr 136.0672 N./mm²
 Reinf Allowable Stress at Ambient SA 137.9000 N./mm²
 Reinf Yield Stress at Design Temp Sy 156.1065 N./mm²

C-Factor for Reinforcement (from UG-47) 2.1000
 DELTA (Reinforcement Material Parameter) 485.0000 N./mm²^{0.5}

SHORT-SIDE RECTANGULAR BEAM DATA:
 Outside Distance from Outside of Vessel 50.0000 mm.
 Width of Reinforcing Member 10.0000 mm.

LONG-SIDE RECTANGULAR BEAM DATA:
 Outside Distance from Outside of Vessel 50.0000 mm.
 Width of Reinforcing Member 10.0000 mm.

Rectangular Vessel Results, Item number 1, Desc: Rectangular-6**ASME Code, Section VIII, Division 1, 2017 App. 13****Ligament Efficiency Calculations (Section 13-6, Equations (1)-(6)):****Short-side 1 Calculations**

Membrane Ligament Efficiency [Em]:
 = 1.000

Bending Ligament Efficiency [Eb]:
 = 1.000

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:
 = $t1 - CA / 2$
 = $3.000 - 0.000 / 2$
 = 1.500 mm.

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

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FileName : Rectangular Section - 1

Rectves Analysis : Rectangular-6 Item: 1 5:40p Feb 20,2026

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Short-side 2 Calculations

Membrane Ligament Efficiency [Em]:

$$= 1.000$$

Bending Ligament Efficiency [Eb]:

$$= 1.000$$

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:

$$\begin{aligned}
 &= t_1 - CA / 2 \\
 &= 3.000 - 0.000 / 2 \\
 &= 1.500 \text{ mm.}
 \end{aligned}$$

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Long-side 1 Calculations

Membrane Ligament Efficiency [Em]:

$$= 1.000$$

Bending Ligament Efficiency [Eb]:

$$= 1.000$$

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:

$$\begin{aligned}
 &= t_1 - CA / 2 \\
 &= 3.000 - 0.000 / 2 \\
 &= 1.500 \text{ mm.}
 \end{aligned}$$

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Long-side 2 Calculations

Membrane Ligament Efficiency [Em]:

$$= 1.000$$

Bending Ligament Efficiency [Eb]:

$$= 1.000$$

Dist from Neutral axis of c/s to inside surface of the vessel [Ci]:

$$\begin{aligned}
 &= t_1 - CA / 2 \\
 &= 3.000 - 0.000 / 2 \\
 &= 1.500 \text{ mm.}
 \end{aligned}$$

Dist from Neutral axis of c/s to extreme outside surface of the section [Co]:

$$\begin{aligned}
 &= -(t_1 - CA) / 2 \\
 &= -(3.000 - 0.000) / 2 \\
 &= -1.500 \text{ mm.}
 \end{aligned}$$

Ligament Efficiency Calculations (Section 13-6, Equations (1)-(6)):

	Em	Eb	Ci	Co
Short-side 1	1.000	1.000	1.500	-1.500

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

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FileName : Rectangular Section - 1

Rectves Analysis : Rectangular-6 Item: 1 5:40p Feb 20,2026

	2	1.000	1.000	1.500	-1.500
Long-side	1	1.000	1.000	1.500	-1.500
	2	1.000	1.000	1.500	-1.500

Effective Width of Shell Plate (Section 13-8, Eq. (2))

In Compression [w]:

$$= \text{Min}(\text{Min}(t_1, t_2) * \Delta / \sqrt{S_y}, p)$$

$$= \text{Min}(\text{Min}(3.00, 3.00) * 5840.83 / \sqrt{156.11}, 80.00)$$

$$= 80.00 \text{ mm.}$$

In Tension [w]:

$$= 80.00 \text{ mm.}$$

Moment of Inertia of a Strip of the Vessel Wall:

$$\text{Thickness } t_1, I_1 = 0.0000 \text{ cm}^4$$

$$\text{Thickness } t_2, I_2 = 0.0000 \text{ cm}^4$$

Effective Area of Reinforcement on Shell (t * w):

$$\text{Short-side } A_p = 2.4000 \text{ cm}^2$$

$$\text{Long-side } A_p = 2.4000 \text{ cm}^2$$

Moment of Inertia of Effective Area of Reinforcement (w * t3 / 12):**

$$\text{Short-side } I_s = 0.0180 \text{ cm}^4$$

$$\text{Long-side } I_l = 0.0180 \text{ cm}^4$$

Moment of Inertia of Combined Reinforcement and Effective Width:

$$\text{In Compression } I_{l1} = 21.8225 \text{ cm}^4$$

$$I_{t1} = 21.8225 \text{ cm}^4$$

$$\text{In Tension } I_{l1} = 21.8225 \text{ cm}^4$$

$$I_{t1} = 21.8225 \text{ cm}^4$$

Distance from Neutral Axis of Cross Section of Composite Section to the Inside Surface of the Vessel (mm.):

		Ci	Co
Short-side,	in Compression	19.4054	-33.5946
	in Tension	19.4054	-33.5946
Long-side,	in Compression	19.4054	-33.5946
	in Tension	19.4054	-33.5946

Rectangular Vessel Reinforcement Parameters:

$$\text{Alpha1} = H_1 / h_1 = 0.8408$$

$$k(\text{comp}) = (I_{22}/I_{11}) * \text{Alpha1} = 0.8408$$

$$k(\text{tens}) = (I_{22}/I_{11}) * \text{Alpha1} = 0.8408$$

Membrane Stress Calculations per Section 13-8**Membrane Stresses at Short-side 1**

Membrane Stress at Short-side 1 [Sms]:

$$= p * h * p / (2 * (A_1 + p * t_1))$$

$$= 0.50 * 1200.00 * 80.00 / (2 * (500.000 + 80.00 * 3.00))$$

$$= 3.24 \text{ N./mm}^2$$

Membrane Stresses at Short-side 2

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

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FileName : Rectangular Section - 1 -----

Rectves Analysis : Rectangular-6 Item: 1 5:40p Feb 20,2026

Membrane Stress at Short-side 2 [Sms]:

$$\begin{aligned}
 &= p * h * p / (2 * (A1 + p * t1)) \\
 &= 0.50 * 1200.00 * 80.00 / (2 * (500.000 + 80.00 * 3.00)) \\
 &= 3.24 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stresses at Long-side 1**Membrane Stress at Long-side 1 at A[SmlA]:**

$$\begin{aligned}
 &= p * H * p / (2 * (A2 + p * t2)) \\
 &= 0.50 * 1000.00 * 80.00 / (2 * (5.000 + 80.00 * 3.00)) \\
 &= 2.70 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stresses at Long-side 2**Membrane Stress at Long-side 2 at A[SmlA]:**

$$\begin{aligned}
 &= p * H * p / (2 * (A2 + p * t2)) \\
 &= 0.50 * 1000.00 * 80.00 / (2 * (5.000 + 80.00 * 3.00)) \\
 &= 2.70 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stresses at Corner sections**Membrane Stress at Short side [Smsc]:**

$$\begin{aligned}
 &= p * h * p / (2 * (A1 + p * t1)) \\
 &= 0.50 * 1200.00 * 80.00 / (2 * (5.000 + 80.00 * 3.00)) \\
 &= 3.24 \text{ N./mm}^2
 \end{aligned}$$

Membrane Stress at Long side [Smlc]:

$$\begin{aligned}
 &= p * H * p / (2 * (A2 + p * t2)) \\
 &= 0.50 * 1000.00 * 80.00 / (2 * (5.000 + 80.00 * 3.00)) \\
 &= 2.70 \text{ N./mm}^2
 \end{aligned}$$

MEMBRANE STRESSES: Membrane Stress Calculations per Section 13-8, Equations (3) and (4). (N./mm²) :

STRESS LOCATIONS	Actual	Allowable
Short-side 1	3.24	136.07
Short-side 2	3.24	136.07
Short-side Corner	3.24	136.07
Long-side 1 at A	2.70	136.07
Long-side 2 at A	2.70	136.07
Long-side Corner	2.70	136.07

Bending Stress Calculations per Section 13-8**Bending Stresses at Short-side 1****Bending Stress at Short-side 1 at N Inner[SbsNi]:**

$$\begin{aligned}
 &= P * p * c / (24 * I11) * [-3 * H^2 + 2 * h^2 \\
 &\quad * ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.5 * 80.0 * 19.41 / (24 * 21.8) * [-3 * 1000.00^2 + 2 * 1200.00^2 * \\
 &\quad ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\
 &= -7.49 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 1 at N Outer[SbsNo]:

$$\begin{aligned}
 &= P * p * c / (24 * I11) * [-3 * H^2 + 2 * h^2 \\
 &\quad * ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.5 * 80.0 * -33.59 / (24 * 21.8) * [-3 * 1000.00^2 + 2 * 1200.00^2 * \\
 &\quad ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\
 &= 12.97 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Short-side 1 at Q Inner[SbsQi]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I11) * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k)) \\
 &= 0.50 * 1200.00^2 * 80.00 * 19.41 / (12 * 21.82) *
 \end{aligned}$$

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Rectangular Duct for Flash Dryer - Section 1

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FileName : Rectangular Section - 1

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$$((1 + 0.84^2 * 0.84) / (1 + 0.84)) \\ = 36.97 \text{ N./mm}^2$$

Bending Stress at Short-side 1 at Q Outer[SbsQo]:

$$= P * h^2 * p * c / (12 * I_{11}) * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\ = 0.50 * 1200.00^2 * 80.00 * -33.59 / (12 * 21.82) * \\ ((1 + 0.84^2 * 0.84) / (1 + 0.84)) \\ = -64.00 \text{ N./mm}^2$$

Bending Stresses at Short-side 2

Bending Stress at Short-side 2 at N Inner[SbsNi]:

$$= P * p * c / (24 * I_{11}) * [-3 * H^2 + 2 * h^2 \\ * ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\ = 0.5 * 80.0 * 19.41 / (24 * 21.8) * [-3 * 1000.00^2 + 2 * 1200.00^2 * \\ ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\ = -7.49 \text{ N./mm}^2$$

Bending Stress at Short-side 2 at N Outer[SbsNo]:

$$= P * p * c / (24 * I_{11}) * [-3 * H^2 + 2 * h^2 \\ * ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\ = 0.5 * 80.0 * -33.59 / (24 * 21.8) * [-3 * 1000.00^2 + 2 * 1200.00^2 * \\ ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\ = 12.97 \text{ N./mm}^2$$

Bending Stress at Short-side 2 at Q Inner[SbsQi]:

$$= P * h^2 * p * c / (12 * I_{11}) * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\ = 0.50 * 1200.00^2 * 80.00 * 19.41 / (12 * 21.82) * \\ ((1 + 0.84^2 * 0.84) / (1 + 0.84)) \\ = 36.97 \text{ N./mm}^2$$

Bending Stress at Short-side 2 at Q Outer[SbsQo]:

$$= P * h^2 * p * c / (12 * I_{11}) * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\ = 0.50 * 1200.00^2 * 80.00 * -33.59 / (12 * 21.82) * \\ ((1 + 0.84^2 * 0.84) / (1 + 0.84)) \\ = -64.00 \text{ N./mm}^2$$

Bending Stresses at Long-side 1

Bending Stress at Long-side 1 at M Inner[SblMi]:

$$= P * h^2 * p * c / (24 * I_{21}) * [-3 + 2 * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\ = 0.50 * 1200.00^2 * 80.00 * 19.41 / (24 * 21.82) * [-3 + 2 * \\ ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\ = -27.06 \text{ N./mm}^2$$

Bending Stress at Long-side 1 at M Outer[SblMo]:

$$= P * h^2 * p * c / (24 * I_{21}) * [-3 + 2 * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k))] \\ = 0.50 * 1200.00^2 * 80.00 * -33.59 / (24 * 21.82) * [-3 + 2 * \\ ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\ = 46.84 \text{ N./mm}^2$$

Bending Stress at Long-side 1 at Q Inner[SblQi]:

$$= P * h^2 * p * c / (12 * I_{21}) * \\ ((1 + \text{Alpha}1^2 * k) / (1 + k)) \\ = 0.50 * 1200.00^2 * 80.00 * 19.41 / (24 * 21.82) * \\ ((1 + 0.84^2 * 0.84) / (1 + 0.84)) \\ = 36.97 \text{ N./mm}^2$$

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Rectangular Duct for Flash Dryer - Section 1

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FileName : Rectangular Section - 1

Rectves Analysis : Rectangular-6

Item: 1 5:40p Feb 20,2026

Bending Stress at Long-side 1 at Q Outer[SblQo]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I21) * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k)) \\
 &= 0.50 * 1200.00^2 * 80.00 * -33.59 / (24 * 21.82) * \\
 &\quad ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\
 &= -64.00 \text{ N./mm}^2
 \end{aligned}$$

Bending Stresses at Long-side 2

Bending Stress at Long-side 2 at M Inner[SblMi]:

$$\begin{aligned}
 &= P * h^2 * p * c / (24 * I21) * [-3 + 2 * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.50 * 1200.00^2 * 80.00 * 19.41 / (24 * 21.82) * [-3 + 2 * \\
 &\quad ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\
 &= -27.06 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Long-side 2 at M Outer[SblMo]:

$$\begin{aligned}
 &= P * h^2 * p * c / (24 * I21) * [-3 + 2 * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k))] \\
 &= 0.50 * 1200.00^2 * 80.00 * -33.59 / (24 * 21.82) * [-3 + 2 * \\
 &\quad ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\
 &= 46.84 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Long-side 2 at Q Inner[SblQi]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I21) * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k)) \\
 &= 0.50 * 1200.00^2 * 80.00 * 19.41 / (24 * 21.82) * \\
 &\quad ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\
 &= 36.97 \text{ N./mm}^2
 \end{aligned}$$

Bending Stress at Long-side 2 at Q Outer[SblQo]:

$$\begin{aligned}
 &= P * h^2 * p * c / (12 * I21) * \\
 &\quad ((1 + Alpha1^2 * k) / (1 + k)) \\
 &= 0.50 * 1200.00^2 * 80.00 * -33.59 / (24 * 21.82) * \\
 &\quad ((1 + 0.84^2 * 0.84) / (1 + 0.84))] \\
 &= -64.00 \text{ N./mm}^2
 \end{aligned}$$

BENDING STRESSES: Bending Stress Calculations per Section 13-8, Equations (5-8). (N./mm²) :

STRESS LOCATIONS	Inner	Outer	Allowable
Short-side 1 at N	-7.49	12.97	104.07
at Q	36.97	-64.00	104.07
Short-side 2 at N	-7.49	12.97	104.07
at Q	36.97	-64.00	104.07
Long-side 1 at M	-27.06	46.84	104.07
at Q	36.97	-64.00	104.07
Long-side 2 at M	-27.06	46.84	104.07
at Q	36.97	-64.00	104.07

Total Stress Calculations per Section 13-8**Total Stresses at Short-side 1**

Total Stress at short side 1 at N inner [STS_Ni]:

$$\begin{aligned}
 &= Sms + SbsNi \\
 &= 3.24 + -7.49 \\
 &= -4.25 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 1 at N outer [STS_No]:

$$\begin{aligned}
 &= Sms + SbsNo \\
 &= 3.24 + 12.97 \\
 &= 16.22 \text{ N./mm}^2
 \end{aligned}$$

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

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FileName : Rectangular Section - 1

Rectves Analysis : Rectangular-6

Item: 1 5:40p Feb 20,2026

Total Stress at short side 1 at Q inner [STS_Qi]:

$$\begin{aligned}
 &= S_{msc} + S_{bsQi} \\
 &= 3.24 + 36.97 \\
 &= 40.21 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 1 at Q outer [STS_Qo]:

$$\begin{aligned}
 &= S_{msc} + S_{bsQo} \\
 &= 3.24 + -64.00 \\
 &= -60.76 \text{ N./mm}^2
 \end{aligned}$$

Total Stresses at Short-side 2

Total Stress at short side 2 at N inner [STS_Ni]:

$$\begin{aligned}
 &= S_{ms} + S_{bsNi} \\
 &= 3.24 + -7.49 \\
 &= -4.25 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 2 at N outer [STS_No]:

$$\begin{aligned}
 &= S_{ms} + S_{bsNo} \\
 &= 3.24 + 12.97 \\
 &= 16.22 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 2 at Q inner [STS_Qi]:

$$\begin{aligned}
 &= S_{msc} + S_{bsQi} \\
 &= 3.24 + 36.97 \\
 &= 40.21 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at short side 2 at Q outer [STS_Qo]:

$$\begin{aligned}
 &= S_{msc} + S_{bsQo} \\
 &= 3.24 + -64.00 \\
 &= -60.76 \text{ N./mm}^2
 \end{aligned}$$

Total Stresses at Long-side 1

Total Stress at long side 1 at M inner [STL_Mi]:

$$\begin{aligned}
 &= S_{mlA} + S_{blMi} \\
 &= 2.70 + -27.06 \\
 &= -24.35 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 1 at M outer [STL_Mo]:

$$\begin{aligned}
 &= S_{mlA} + S_{blMo} \\
 &= 2.70 + 46.84 \\
 &= 49.55 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 1 at Q inner [STL_Qi]:

$$\begin{aligned}
 &= S_{mlc} + S_{blQi} \\
 &= 2.70 + 36.97 \\
 &= 39.67 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 1 at Q outer [STL_Qo]:

$$\begin{aligned}
 &= S_{mlc} + S_{blQo} \\
 &= 2.70 + -64.00 \\
 &= -61.30 \text{ N./mm}^2
 \end{aligned}$$

Total Stresses at Long-side 2

Total Stress at long side 2 at M inner [STL_Mi]:

$$\begin{aligned}
 &= S_{mlA} + S_{blMi} \\
 &= 2.70 + -27.06 \\
 &= -24.35 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 2 at M outer [STL_Mo]:

MICROTEC**Rectangular Duct for Flash Dryer - Section 1**

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 1

Rectves Analysis : Rectangular-6 Item: 1 5:40p Feb 20,2026

$$\begin{aligned}
 &= SmlA + SblMo \\
 &= 2.70 + 46.84 \\
 &= 49.55 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 2 at Q inner [STL_Qi]:

$$\begin{aligned}
 &= Smlc + SblQi \\
 &= 2.70 + 36.97 \\
 &= 39.67 \text{ N./mm}^2
 \end{aligned}$$

Total Stress at long side 2 at Q outer [STL_Qo]:

$$\begin{aligned}
 &= Smlc + SblQo \\
 &= 2.70 + -64.00 \\
 &= -61.30 \text{ N./mm}^2
 \end{aligned}$$

**TOTAL STRESSES: Total Stress Calculations per Section 13-8,
Equations (9-12). (N./mm²) :**

STRESS LOCATIONS	Inner	Outer	Allowable
Short-side 1 at N	-4.25	16.22	104.07
at Q	40.21	-60.76	104.07
Short-side 2 at N	-4.25	16.22	104.07
at Q	40.21	-60.76	104.07
Long-side 1 at M	-24.35	49.55	104.07
at Q	39.67	-61.30	104.07
Long-side 2 at M	-24.35	49.55	104.07
at Q	39.67	-61.30	104.07

SUMMARY OF RESULTS:**MEMBRANE STRESS SUMMARY,**

High Stress (Highest % of Allowable)	3.24	N./mm ²
High Stress Percentage	2.38	%
M.A.W.P. for Membrane Stresses	20.98	bars

BENDING STRESS SUMMARY,

High Stress (Highest % of Allowable)	-64.00	N./mm ²
High Stress Percentage	61.50	%
M.A.W.P. for Bending Stresses	0.81	bars

TOTAL STRESS SUMMARY,

High Stress (Highest % of Allowable)	-61.30	N./mm ²
High Stress Percentage	58.90	%
M.A.W.P. for Total Stresses	0.85	bars

MICROTEC

Rectangular Duct for Flash Dryer - Section 1

PV Elite 2019 SP1 Licensee: SL Client

FileName : Rectangular Section - 1

Vessel Results Summary

Item: 1 5:40p Feb 20,2026

Rectangular Vessel Results For Item 1 : A4**SUMMARY OF RESULTS:****MEMBRANE STRESS SUMMARY,**

High Stress (Highest % of Allowable)	3.24	N./mm ²
High Stress Percentage	2.38	%
M.A.W.P. for Membrane Stresses	20.98	bars



BENDING STRESS SUMMARY,

High Stress (Highest % of Allowable)	-64.00	N./mm ²
High Stress Percentage	61.50	%
M.A.W.P. for Bending Stresses	0.81	bars



TOTAL STRESS SUMMARY,

High Stress (Highest % of Allowable)	-61.30	N./mm ²
High Stress Percentage	58.90	%
M.A.W.P. for Total Stresses	0.85	bars

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	MODIFIED STARCH FLASH DRYER PROJECT						
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.		Rev.
	5204	ME	CL	000	02		00

APPENDIX 6 **CALCULATION HOPPER** **ZONE 7**

 Innovation and Equipment	MODIFIED STARCH FLASH DRYER PROJECT						 A.G.F.D. Tândărei
	FLASH DRYER THICKNESS CALCULATION						
	Project No.	Discipline	Document Type	Plant / Equipment No	Sequence No.	Rev.	
	5204	ME	CL	000	02	00	

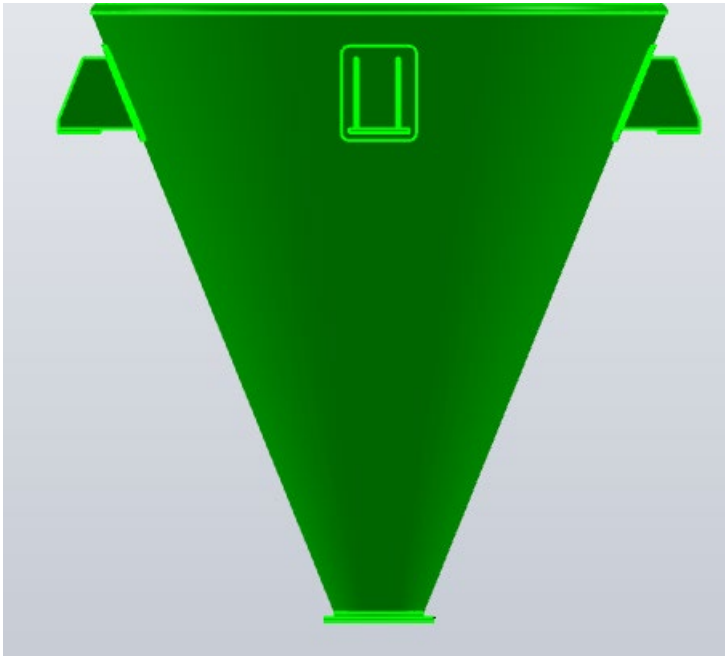
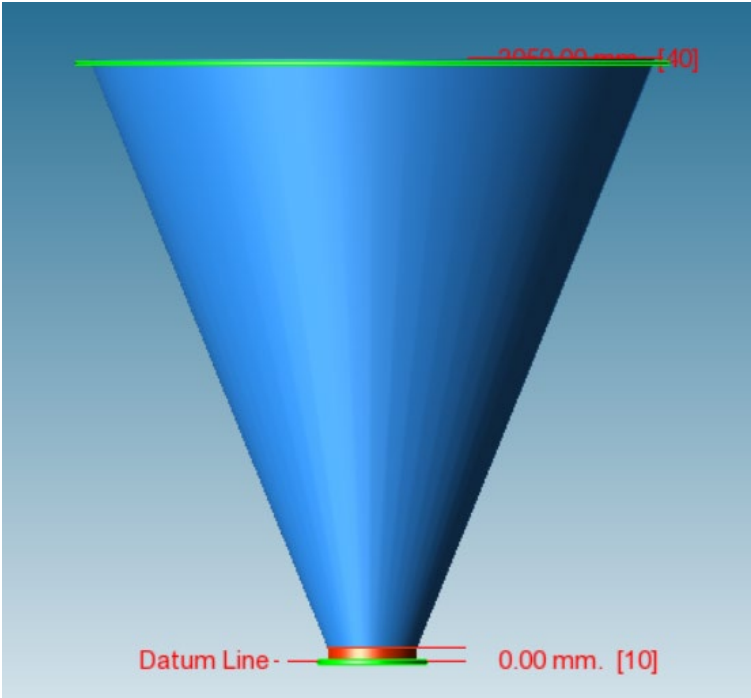


Figure F-Hopper - Zone7

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DESIGN CALCULATION

In Accordance with ASME Section VIII Division 1

ASME Code Version : 2017

Analysis Performed by : SPLM Licensed User

Job File : D:\2-PAYAM\1-PROJECTS\MICROTEC PROJECT\13-MICROT

Date of Analysis : Feb 20,2026 5:10pm

PV Elite 2019 SP1, March 2019

Note:

PV Elite performs all calculations internally in Imperial Units to remain compliant with the ASME Code and any built in assumptions in the ASME Code formulas. The finalized results are reflected to show the user's set of selected units.

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FileName : Hopper -01 - 02-20-2026 -----

Warnings and Errors: Step: 0 5:10pm Feb 20,2026

Class From To : Basic Element Checks.

=====

Class From To: Check of Additional Element Data

=====

There were no geometry errors or warnings.

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FileName : Hopper -01 - 02-20-2026 -----

Input Echo: Step: 1 5:10pm Feb 20,2026

PV Elite Vessel Analysis Program: Input Data

Design Internal Pressure (for Hydrotest)	0.5	bars
Design Internal Temperature	170.0	°C
Type of Hydrotest	not Specified	
Hydrotest Position	Horizontal	
Projection of Nozzle from Vessel Top	0	mm.
Projection of Nozzle from Vessel Bottom	0	mm.
Minimum Design Metal Temperature	-28.9	°C
Type of Construction	Welded	
Special Service	None	
Degree of Radiography	RT-1	
Use Higher Longitudinal Stresses (Flag)	Y	
Select t for Internal Pressure (Flag)	N	
Select t for External Pressure (Flag)	N	
Select t for Axial Stress (Flag)	N	
Select Location for Stiff. Rings (Flag)	N	
Consider Vortex Shedding	N	
Perform a Corroded Hydrotest	N	

Load Case 1	NP+EW+WI+FW+BW
Load Case 2	NP+EW+EE+FS+BS
Load Case 3	NP+OW+WI+FW+BW
Load Case 4	NP+OW+EQ+FS+BS
Load Case 5	NP+HW+HI
Load Case 6	NP+HW+HE
Load Case 7	IP+OW+WI+FW+BW
Load Case 8	IP+OW+EQ+FS+BS
Load Case 9	EP+OW+WI+FW+BW
Load Case 10	EP+OW+EQ+FS+BS
Load Case 11	HP+HW+HI
Load Case 12	HP+HW+HE
Load Case 13	IP+WE+EW
Load Case 14	IP+WF+CW
Load Case 15	IP+VO+OW
Load Case 16	IP+VE+EW
Load Case 17	NP+VO+OW
Load Case 18	FS+BS+IP+OW
Load Case 19	FS+BS+EP+OW

Wind Design Code	ASCE-7 93
Basic Wind Speed [V]	112.65 Km/hr
Surface Roughness Category	C: Open Terrain
Importance Factor	1.0
Type of Surface	Moderately Smooth
Base Elevation	0 mm.
Percent Wind for Hydrotest	33.0
Using User defined Wind Press. Vs Elev.	N
Damping Factor (Beta) for Wind (Ope)	0.0100
Damping Factor (Beta) for Wind (Empty)	0.0000
Damping Factor (Beta) for Wind (Filled)	0.0000

Seismic Design Code	UBC 94
UBC Seismic Zone (1=1,2=2a,3=2b,4=3,5=4)	0.000
UBC Importance Factor	1.000
UBC Soil Type	S1
UBC Horizontal Force Factor	3.000
UBC Percent Seismic for Hydrotest	0.000

Design Pressure + Static Head	Y
Consider MAP New and Cold in Noz. Design	N
Consider External Loads for Nozzle Des.	Y
Use ASME VIII-1 Appendix 1-9	N

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FileName : Hopper -01 - 02-20-2026 -----

Input Echo: Step: 1 5:10pm Feb 20,2026

Material Database Year Current w/Addenda or Code Year

Configuration Directives:

Do not use Nozzle MDMT Interpretation VIII-1 01-37	No
Use Table G instead of exact equation for "A"	Yes
Shell Head Joints are Tapered	Yes
Compute "K" in corroded condition	Yes
Use Code Case 2286	No
Use the MAWP to compute the MDMT	Yes
For thickness ratios ≤ 0.35 , MDMT will be -155F (-104C)	Yes
For PWHT & P1 Materials the MDMT can be $< -55F (-48C)$	No
Using Metric Material Databases, ASME II D	No
Calculate B31.3 type stress for Nozzles with Loads	Yes
Reduce the MDMT due to lower membrane stress	Yes
Consider Longitudinal Stress in MDMT calcs. (Div. 1)	No

Complete Listing of Vessel Elements and Details:

Element From Node	10	
Element To Node	20	
Element Type	Cylinder	
Description		
Distance "FROM" to "TO"	50	mm.
Inside Diameter	300	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Allowable Stress, Ambient	137.9	N./mm ²
Allowable Stress, Operating	136.07	N./mm ²
Allowable Stress, Hydrotest	179.27	N./mm ²
Material Density	0.008027	kg./cm ³
P Number Thickness	0	mm.
Yield Stress, Operating	156.11	N./mm ²
External Pressure Chart Name	HA-2	
UNS Number	S31600	
Product Form	Plate	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Weld is pre-Heated	No	

Element From Node	10	
Detail Type	Ring	
Detail ID	Ring:[1 of 1]	
Dist. from "FROM" Node / Offset dist	0	mm.
Inside Diameter of Ring	306	mm.
Thickness of Ring	14	mm.
Outside Diameter of Ring	370	mm.
Material Name	SA-240 304	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	20
Element To Node	40
Element Type	Conical
Description	

FileName : Hopper -01 - 02-20-2026 -----

Input Echo: Step: 1 5:10pm Feb 20,2026

Distance "FROM" to "TO"	2000	mm.
Inside Diameter	300	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Cone Diameter at "To" End	1920	mm.
Design Length of Cone	2000	mm.
Half Apex Angle of Cone	22.047947	degrees
Toriconical (Y/N)	N	
Weld is pre-Heated	No	
Element From Node	20	
Detail Type	Ring	
Detail ID	Flange -1	
Dist. from "FROM" Node / Offset dist	1983	mm.
Inside Diameter of Ring	1912.2	mm.
Thickness of Ring	12	mm.
Outside Diameter of Ring	2020.9	mm.
Material Name	SA-240 304	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

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FileName : Hopper -01 - 02-20-2026 -----
XY Coordinate Calculations: Step: 2 5:10pm Feb 20,2026

XY Coordinate Calculations:

From	To	X (Horiz.) mm.	Y (Vert.) mm.	DX (Horiz.) mm.	DY (Vert.) mm.
10	20	...	50	...	50
20	40	...	2050	...	2000

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FileName : Hopper -01 - 02-20-2026 -----

Internal Pressure Calculations: Step: 3 5:10pm Feb 20,2026

Element Thickness, Pressure, Diameter and Allowable Stress :

From	To	Int. Press + Liq. Hd bars	Nominal Thickness mm.	Total Corr Allowance mm.	Element Diameter mm.	Allowable Stress (SE) N./mm ²
10	20	0.5	3	...	300	136.07
20	40	0.5	3	...	1920	136.07

Element Required Thickness and MAWP :

From	To	Design Pressure bars	M.A.W.P. Corroded bars	M.A.P. New & Cold bars	Minimum Thickness mm.	Required Thickness mm.
10	20	0.5	26.8892	27.2514	3	1.5
20	40	0.5	3.93408	3.98707	3	1.5
Minimum			3.934	3.987		

MAWP: 3.934 bars, limited by: Cone.

Internal Pressure Calculation Results :

ASME Code, Section VIII Division 1, 2017

Cylindrical Shell From 10 To 20 SA-240 316 at 170 °C

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 \cdot 150.0) / (136.07 \cdot 1.0 - 0.6 \cdot 0.5) \\
 &= 0.0551 + 0.0000 = 0.0551 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (136.07 \cdot 1.0 \cdot 3.0) / (150.0 + 0.6 \cdot 3.0) \\
 &= 26.889 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (150.0 + 0.6 \cdot 3.0) \\
 &= 27.251 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t) \\
 &= (0.5 \cdot (150.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0) \\
 &= 2.530 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.990 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Conical Section From 20 To 40 SA-240 316 at 170 °C

Material UNS Number: S31600

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FileName : Hopper -01 - 02-20-2026 -----

Internal Pressure Calculations: Step: 3 5:10pm Feb 20, 2026

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot D) / (2 \cdot \cos(a) \cdot (S \cdot E - 0.6 \cdot P)) \text{ per Appendix 1-4 (e)} \\
 &= (0.5 \cdot 1920.0) / (2 \cdot 0.9269 \cdot (136.07 \cdot 1.0 - 0.6 \cdot 0.5)) \\
 &= 0.3807 + 0.0000 = 0.3807 \text{ mm.}
 \end{aligned}$$

Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 136.07 \cdot 1.0 \cdot 3.0 \cdot 0.927) / (1920.0 + 1.2 \cdot 3.0 \cdot 0.927) \\
 &= 3.934 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 137.9 \cdot 1.0 \cdot 3.0 \cdot 0.9269) / (1920.0 + 1.2 \cdot 3.0 \cdot 0.9269) \\
 &= 3.987 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (D + 1.2 \cdot t \cdot \cos(a))) / (2 \cdot E \cdot t \cdot \cos(a)) \\
 &= (0.5 \cdot (1920.0 + 1.2 \cdot 3.0 \cdot 0.9269)) / (2 \cdot 1.0 \cdot 3.0 \cdot 0.9269) \\
 &= 17.293 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1 $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$ 0.989 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

Note: The Pressure at the Large Diameter is used in the TR calculation.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Elements Suitable for Internal Pressure.

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FileName : Hopper -01 - 02-20-2026 -----

External Pressure Calculations: Step: 4 5:10pm Feb 20,2026

External Pressure Calculation Results :

External Pressure Calculations:

From	To	Section Length mm.	Outside Diameter mm.	Corroded Thickness mm.	Factor A	Factor B N./mm ²
10	Ring	No Calc	...	3	No Calc	No Calc
Ring	20	50	306	3	0.01054	99.0623
20	Ring	1150.16	1912.7	3	0.00012505	12.0049
Ring	40	16.9392	1926.47	3	0.0021617	81.3491

External Pressure Calculations:

From	To	External Actual T. mm.	External Required T. mm.	External Design Pressure bars	External M.A.W.P. bars
10	Ring	3	No Calc	0.1	No Calc
Ring	20	3	0.18486	0.1	12.9486
20	Ring	3	2.15271	0.1	0.23268
Ring	40	3	0.75209	0.1	1.56547
Minimum					0.233

External Pressure Calculations:

From	To	Actual Length mm.	Allowable Length mm.	Ring Inertia Required cm**4	Ring Inertia Available cm**4
10	Ring	No Calc	No Calc	No Calc	No Calc
Ring	20	50	5133.63	0.00051345	6.33378
20	Ring	1150.16	1150.16	No Calc	No Calc
Ring	40	16.9392	16.9392	2.92476	30.9459

Elements Suitable for External Pressure.

ASME Code, Section VIII Division 1, 2017

Cone From 20 to Flange -1 Ext. Chart: HA-2 at 50 °C

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.781	1912.70	1150.16	687.87	0.6013	0.0001251	12.00

EMAP = (4*B)/(3*(D/t)) = (4*12.0049)/(3*687.8719) = 0.2327 bars

Note: The cone thickness used in the calculation has been modified per UG-33(f), $t_e = t * \cos(\alpha)$.

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.995	1912.70	1150.16	958.61	0.6013	0.0000749	7.19

EMAP = (4*B)/(3*(D/t)) = (4*7.1904)/(3*958.6123) = 0.1 bars

Note: The cone thickness used in the calculation has been modified per UG-33(f), $t_e = t * \cos(\alpha)$.

Cone From Flange -1 to the end: Ext. Chart: HA-2 at 50 °C

FileName : Hopper -01 - 02-20-2026 -----

External Pressure Calculations: Step: 4 5:10pm Feb 20,2026

Stiffening Ring Calcs for : Ring:[1 of 1] , SA-240 304 , Bar Ring: 32 x 14 mm.

Effective Length of Shell: 33 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	1.000	1.5000	1.500
Ring :	4.480	19.0000	85.120
Total:	5.480		86.620

Centroid of Ring plus Shell: 16 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.007	14.3070	204.659
Ring :	3.823	-3.1930	45.676
Total:	3.830		250.335

Available Moment of Inertia, Ring plus Shell: 6 cm**4

Required Stress in Ring plus Shell Breq 0.11 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000011

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (306.0^2 * 25.0(3.0 + 4.48/25.0)0.000001) / 10.9$$

$$= 0 \text{ cm**4}$$

Stiffening Ring Calcs for : Flange -1 , SA-240 304 , Bar Ring: 54 x 12 mm.

Effective Length of Shell: 83 mm.

	Area (cm ²)	Distance (mm.)	Area*Dist
Shell:	2.499	1.5000	3.749
Ring :	6.523	30.1800	196.870
Total:	9.023		200.619

Centroid of Ring plus Shell: 22 mm.

	Inertia	Distance	A*Dist ²
Shell:	0.019	20.7351	1074.620
Ring :	16.063	-7.9449	411.755
Total:	16.082		1486.376

Available Moment of Inertia, Ring plus Shell: 31 cm**4

Required Stress in Ring plus Shell Breq 3.48 N./mm²
 Required Strain in Ring plus Shell Areq 0.0000363

Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1912.2301^2 * 583.5508(3.0 + 6.5232/583.5508)0.000036) / 10.9$$

$$= 3 \text{ cm**4}$$

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FileName : Hopper -01 - 02-20-2026 -----

Element and Detail Weights: Step: 5 5:10pm Feb 20,2026

Element and Detail Weights:

From	To	Element		Corroded		Corroded		Extra due
		Metal	Wgt. ID	Volume	Metal	Wgt. ID	Volume	Misc %
		kg.		Cm.	kg.		Cm.	kg.
10	20	1.14618		3534.92	1.14618		3534.92	...
20	40	181.738		2279319	181.738		2279319	...
Total		182		2282854.50	182		2282854.50	0

Weight of Details:

From	Type	Weight of Detail	X Offset, Dtl. Cent.	Y Offset, Dtl. Cent.	Description
		kg.	mm.	mm.	
10	Ring	3.81869	Ring:[1 of 1]
20	Ring	32.3515	...	1983	Flange -1

Total Weight of Each Detail Type:

Stiffeners	36.2
Sum of the Detail Weights	36.2 kg.

Weight Summation Results: (kg.)

	Fabricated	Shop Test	Shipping	Erected	Empty	Operating
Main Elements	182.9	182.9	182.9	182.9	182.9	182.9
Stif. Rings	36.2	36.2	36.2	36.2	36.2	36.2
Test Liquid	...	2281.5
Totals	219.1	2500.5	219.1	219.1	219.1	219.1

Weight Summary:

Fabricated Wt.	- Bare Weight without Removable Internals	219.1 kg.
Shop Test Wt.	- Fabricated Weight + Water (Full)	2500.5 kg.
Shipping Wt.	- Fab. Weight + removable Intls.+ Shipping App.	219.1 kg.
Erected Wt.	- Fab. Wt + or - loose items (trays,platforms etc.)	219.1 kg.
Ope. Wt. no Liq	- Fab. Weight + Internals. + Details + Weights	219.1 kg.
Operating Wt.	- Empty Weight + Operating Liq. Uncorroded	219.1 kg.
Field Test Wt.	- Empty Weight + Water (Full)	2500.5 kg.
Mass of the Upper 1/3 of the Vertical Vessel		94.4 kg.

Outside Surface Areas of Elements:

From	To	Surface Area
		cm ²
10	20	480.664
20	40	75684.9
Total		76165.602 cm ²

Element and Detail Weights:

From	To	Total Ele. Empty Wgt.	Total. Ele. Oper. Wgt.	Total. Ele. Hydro. Wgt.	Total Dtl. Offset Mom.	Oper. Wgt. No Liquid
		kg.	kg.	kg.	Kg-m.	kg.

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FileName : Hopper -01 - 02-20-2026 -----

Element and Detail Weights: Step: 5 5:10pm Feb 20,2026

10	20	4.96487	4.96487	4.96487	...	4.96487
20	40	214.09	214.09	214.09	...	214.09

Cumulative Vessel Weight

From	To	Cumulative Ope Wgt. No Liquid kg.	Cumulative Oper. Wgt. kg.	Cumulative Hydro. Wgt. kg.
10	20	219.055	219.055	219.055
20	40	214.09	214.09	214.09

Note: The cumulative operating weights no liquid in the column above
are the cumulative operating weights minus the operating liquid
weight minus any weights absent in the empty condition.

Cumulative Vessel Moment

From	To	Cumulative Empty Mom. Kg-m.	Cumulative Oper. Mom. Kg-m.	Cumulative Hydro. Mom. Kg-m.
10	20
20	40

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FileName : Hopper -01 - 02-20-2026 -----

Center of Gravity Calculation: Step: 6 5:10pm Feb 20,2026

Shop/Field Installation Options :

Note : The CG is computed from the first Element From Node

Center of Gravity of Stiffening Rings	1818.365 mm.
Center of Gravity of Bare Shell New and Cold	1285.295 mm.
Center of Gravity of Bare Shell Corroded	1285.295 mm.
Vessel CG in the Operating Condition	1373.315 mm.
Vessel CG in the Fabricated (Shop/Empty) Condition	1373.315 mm.

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MDMT Summary: Step: 7 5:10pm Feb 20,2026

Minimum Design Metal Temperature Results Summary :

Description	Curve	Basic MDMT °C	Reduced MDMT °C	UG-20 (f) MDMT °C	Thickness ratio	Gov Thk mm.	E*	PWHT reqd
Notes								
[16]		-196						
[16]		-196						

Warmest MDMT:		-196						

Required Minimum Design Metal Temperature					-28.9	°C		
Warmest Computed Minimum Design Metal Temperature					-196.0	°C		

Notes:

- [!] - This was an impact tested material.
- [1] - Governing Nozzle Weld.
- [4] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-c).
- [5] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-b).
- [6] - MDMT Calculations at the Shell/Head Joint.
- [7] - MDMT Calculations for the Straight Flange.
- [8] - Cylinder/Cone/Flange Junction MDMT.
- [9] - Calculations in the Spherical Portion of the Head.
- [10] - Calculations in the Knuckle Portion of the Head.
- [11] - Calculated (Body Flange) Flange MDMT.
- [12] - Calculated Flat Head MDMT per UCS-66.3
- [13] - Tubesheet MDMT, shell side, if applicable
- [14] - Tubesheet MDMT, tube side, if applicable
- [15] - Nozzle Material
- [16] - Shell or Head Material
- [17] - Impact Testing required
- [18] - Impact Testing not required, see UCS-66(b)(3)
- [19] - Select a valid hydrotest type to get the UG-20(f) exemption
- [20] - Cylinder/Cone Junction MDMT based on Longitudinal Stress considerations
- [21] - Bolting Material

UG-84(b)(2) was not considered.

UCS-66(g) was not considered.

UCS-66(i) was not considered.

Notes:

Impact test temps were not entered in and not considered in the analysis.
 UCS-66(i) applies to impact tested materials not by specification and
 UCS-66(g) applies to materials impact tested per UG-84.1 General Note (c).
 The Basic MDMT includes the (30F) PWHT credit if applicable.

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Vessel Design Summary:

Step: 8 5:10pm Feb 20,2026

ASME Code, Section VIII Division 1, 2017

Diameter Spec : 300.000 mm. ID
 Vessel Design Length, Tangent to Tangent 2050.00 mm.
 Distance of Bottom Tangent above Grade 50.00 mm.
 Distance of Base above Grade 0.00 mm.
 Specified Datum Line Distance 0.00 mm.
 Internal Design Temperature 170 °C
 Internal Design Pressure 0.500 bars
 External Design Temperature 50 °C
 External Design Pressure 0.100 bars
 Maximum Allowable Working Pressure 3.934 bars
 External Max. Allowable Working Pressure 0.233 bars
 Hydrostatic Test Pressure 0.000 bars
 Required Minimum Design Metal Temperature -28.9 °C
 Warmest Computed Minimum Design Metal Temperature -196.0 °C
 Wind Design Code ASCE-93
 Earthquake Design Code UBC-94

Materials of Construction:

Component Type	Material	Class	Thickness	UNS #	Normal ized	Impact Tested
Shell	SA-240 316	S31600	No	No
Cone	SA-240 316	S31600	No	No
Rings	SA-240 304	S30400	No	No

Normalized is determined based on the UCS-66 material curve selection and Figure UCS-66.

Impact Tested is based on material selection and material data properties.

Element Pressures and MAWP (bars & mm.):

Element Description or Type	Design Pressure + Stat. head	Ext. Press.	Element M.A.W.P	Corrosion Allowance	Str. Flg. Gov.	In Creep Range
Cylinder	0.500	0.10	26.889	0.0000	N/A	No
Conical	0.500	0.10	3.934	0.0000	N/A	No

Stiffener Ring Specifications:

Elevation mm.	Selected Type	User Description
0.00	Bar 32.0 x 14.	Ring:[1 of 1]
2033.00	Bar 54.4 x 12.	Flange -1

Element Types and Properties:

Element Type	"To" Elev mm.	Element Length mm.	Nominal Thickness mm.	Finished Thickness mm.	Reqd Thk Internal mm.	Reqd Thk External mm.	Long Eff	Circ Eff
Cylinder	50.0	50.0	3.0	3.0	1.5	...	1.00	1.00
Conical	2050.0	2000.0	3.0	3.0	1.5	...	1.00	1.00

External Pressure Calculations:

From	To	External Actual T.	External Required T.	External Design Pressure	External M.A.W.P.
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Vessel Design Summary: Step: 8 5:10pm Feb 20,2026

		mm.	mm.	bars	bars
10	Ring	3	No Calc	0.1	No Calc
Ring	20	3	0.18486	0.1	12.9486
20	Ring	3	2.15271	0.1	0.23268
Ring	40	3	0.75209	0.1	1.56547

External Pressure Calculations:

From	To	Actual Length Bet. Stiffeners mm.	Allowable Length Bet. Stiffeners mm.	Ring Inertia Required cm**4	Ring Inertia Available cm**4
10	Ring	No Calc	No Calc	No Calc	No Calc
Ring	20	50	5133.63	0.00051345	6.33378
20	Ring	1150.16	1150.16	No Calc	No Calc
Ring	40	16.9392	16.9392	2.92476	30.9459

Factored Loads:

Un-Factored Loads:

Weights:

Fabricated - Bare W/O Removable Internals	219.1	kg.
Shop Test - Fabricated + Water (Full)	2500.5	kg.
Shipping - Fab. + Rem. Intls.+ Shipping App.	219.1	kg.
Erected - Fab. + Rem. Intls.+ Insul. (etc)	219.1	kg.
Empty - Fab. + Intls. + Details + Wghts.	219.1	kg.
Operating - Empty + Operating Liquid (No CA)	219.1	kg.
Field Test - Empty Weight + Water (Full)	2500.5	kg.

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